

**EASA**

**TYPE-CERTIFICATE  
DATA SHEET**

**MDM-1 “Fox”**

Type Certificate Holder:

ZAKŁADY LOTNICZE  
Margański & Mysłowski  
Sp. z o.o.  
ul. Strażacka 60  
43-300 Bielsko-Biała  
POLAND

EASA TCDS No. A.039

For variants:      MDM-1 “Fox”  
                         MDM-1P “Fox-P”

Issue 02, 14 February 2007

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**0.III. Change Record**

Issue	Date	Changes
01	23 June 2005	Transfer from Polish Type Certificate No. BG-197 to the EASA Type Design.
02	14 February 2007	Including the MDM-1P "Fox P" version which was included to the Polish TC No. BG-197 with Certificate of Approval No Z-BG-01/99 on January 12 <sup>th</sup> , 1999. Editorial changes.

## **Section A: MDM-1 "Fox"**

### **A.I. General**

1. Data Sheet No.: EASA.039
2. a) Type: MDM-1 "Fox"  
b) Variant: MDM-1 "Fox"
3. Airworthiness Category: Sailplane, Aerobatic
4. Type Certificate Holder: ZAKŁADY LOTNICZE  
Margański & Mysłowski  
Sp. z o.o.  
ul. Strażacka 60  
43-300 Bielsko-Biała  
POLAND
5. Manufacturer: for S/N P-11, P-13÷P-16, 201÷235 :  
Zakład Remontów i Produkcji Sprzętu Lotniczego  
Edward Margański  
ul. Cieszyńska 321  
43-300 Bielsko-Biała  
POLAND  
  
from S/N 236 :  
ZAKŁADY LOTNICZE  
Margański & Mysłowski  
Sp. z o.o.  
ul. Strażacka 60  
43-300 Bielsko-Biała  
POLAND
6. Polish CAA Certification Date 27 July 1994
7. This EASA Type Certificate Data Sheet replaces Polish TCDS issue 4 (January 1999) to Type Certificate No. BG-197.

### **A.II. Certification Basis**

1. Certification Basis: Defined 13 February 1985
2. Airworthiness Requirements: JAR-22 Change 4, issued on 7 May 1987
3. Requirements elected to comply: None
4. Special Conditions: None
5. Exemptions:
  1. Stall speed with two person crew exceeds 80 km/h (JAR 22.49)
  2. Airbrakes closing force exceeds 20 daN (JAR 22.143)
6. Equivalent Safety Findings: None

### **A.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition: List of drawings, January 19<sup>th</sup> 1998
2. Description: Two-seat (tandem configuration) aerobatic and trainer glider.  
Cantilever mid-wing monoplane with conventional tail unit (fixed stabilizer with elevator, fin and rudder).  
All composite glass-epoxy structure. Bipartite tapered wing with Schempp-Hirth airbrakes protruding only from upper surface.  
Fixed landing gear - main wheel with disk brake and tail wheel.
3. Equipment: Minimum equipment:
  - airspeed indicator,
  - altimeter,
  - accelerometer,
  - 5-points pilot harness,
  - parachute or back cushion (8 cm thick) for each crew member.Standard equipment (besides minimum equipment):
  - variometer with compensator,
  - pressure compensation vessel,
  - bank indicator,
  - compass,
  - seat cushion.
4. Dimensions:

Span	14,00 m
Wing area	12,34 m <sup>2</sup>
Aspect Ratio	15,88
Length	7,38 m
Height	2,25 m
5. Launching Hook: Nose towing hook TOST E 85 or SZD-III A56 P  
CG towing hook (optional) TOST Europa G 88
6. Weak links: Ultimate Strength: 677 daN ( $\pm 10\%$ )
7. Air Speeds:

Manoeuvring Speed	$V_A$	214 km/h
Never Exceed Speed	$V_{NE}$	282 km/h
Maximum permitted speeds		
- in rough air	$V_{RA}$	225 km/h
- in aero-tow	$V_T$	150 km/h
- in winch launching	$V_W$	150 km/h
8. Operational Capability: VFR Day
9. Masses:

Max. Mass	530 kg
Max. Empty Mass (without balance mass)	350 kg
10. Centre of Gravity Range: Empty glider with standard equipment:

Forward Limit	620 mm aft of datum point
Rearward Limit	645 mm aft of datum point

Centre of Gravity operational limits:

Forward Limit	213 mm aft of datum point (22,0% MAC)
Rearward Limit	379 mm aft of datum point (39,0% MAC)

MAC is 971 mm; 0% MAC is 2,1 mm aft of datum.  
Datum: Leading edge and wing-fuselage division plane intersection.  
Levelling means: Leading and trailing points of root chord (1308 mm) at the same level.

11. Seating Capacity: 2
12. Lifetime limitations: Refer to Maintenance Manual
13. Other limitations: Solo flight is permissible only on front seat.  
Cross-country flight (gliding or towed) only with one person crew.  
No baggage permitted.  
The following kinds of operation are forbidden:  
- auto-towing,  
- flights in icing conditions,  
- night flying,  
- cloud flying without turn-and-bank indicator,  
- aerobatic in rough air.  
Manoeuvring load factor limits:  
two person crew +7,0/-5,0  
one person (no more than 100 kg incl. parachute) +9,0/-6,0
14. Deflection of control surfaces:
- |           |         |     |             |
|-----------|---------|-----|-------------|
| Aileron:  | - up    | 22° | ± 1°        |
|           | - down  | 17° | ± 1°        |
| Elevator: | - up    | 25° | ± 1°        |
|           | - down  | 25° | ± 1°        |
| Rudder:   | - left  | 30° | + 1° / - 2° |
|           | - right | 30° | + 1° / - 2° |

#### **A.IV. Operating and Service Instructions**

1. Flight Manuals:

- Polish: Instrukcja Użytkowania w Locie  
Szybowca MDM-1 „Fox”,  
wydanie III, Październik 1996
- Polish: Instrukcja Użytkowania w Locie  
Szybowców MDM-1 „Fox” oraz MDM-1P „Fox P”,  
wydanie IV, Grudzień 1998
- English: Flight Manual  
MDM-1 „Fox”  
issue III, October 1996
- English: Flight Manual  
MDM-1 „Fox”, MDM-1P „Fox P”,  
issue IV, December 1998
- German: Flughandbuch  
MDM-1 „Fox”  
Ausc. III, Oktober 1996, LBA-anerkannt

2. Maintenance Manual:

- Polish: Opis Techniczny, Instrukcja Obsługi Technicznej,  
Prace Okresowe, Szybowca MDM-1 „Fox”,  
wydanie III, Grudzień 1996
- English: Technical Service Manual, Periodic Works,  
MDM-1 „Fox”  
issue III, December 1996
- German: Wartungshandbuch  
MDM-1 „Fox”  
Ausc. III, Dezember 1996

#### **A.V. Notes**

1. Serial Numbers:  
P-11, P-13, P-14, P-15, P-16,  
201 and following
2. All glider outside surfaces must be white painted.  
No registration number or any colour marks on the wings and stabilizer upper surfaces are allowed.

## **Section B: MDM-1P "FOX-P"**

### **B.I. General**

1. Data Sheet No.: EASA.A.039
2. a) Type: MDM-1 "Fox"  
b) Variant: MDM-1P "Fox-P"
3. Airworthiness Category: Sailplane, Aerobatic or Utility
4. Type Certificate Holder: ZAKŁADY LOTNICZE  
Margański & Mysłowski  
Sp. z o.o.  
ul. Strażacka 60  
43-300 Bielsko-Biała  
POLAND
5. Manufacturer: for S/N P-13, P-14, P-15, P-16, 201÷235 :  
Zakład Remontów i Produkcji Sprzętu Lotniczego  
Edward Margański  
ul. Cieszyńska 321  
43-300 Bielsko-Biała  
POLAND  
  
from S/N 236 :  
ZAKŁADY LOTNICZE  
Margański & Mysłowski  
Sp. z o.o.  
ul. Strażacka 60  
43-300 Bielsko-Biała  
POLAND
6. Polish CAA Certification Date: 12 January 1999
7. This EASA Type Certificate Data Sheet replaces Polish TCDS issue 4 (January 1999) to Type Certificate No. BG-197.

### **B.II. Certification Basis**

1. Certification Basis:
2. Airworthiness Requirements: JAR-22 Change 4, issued on May 7<sup>th</sup> 1987
3. Requirements elected to comply: None
4. Special Conditions: None
5. Exemptions:
  1. Stall speed in aerobatic version and with two person crew exceeds 80 km/h (JAR 22.49)
  2. Airbrakes closing force exceeds 20 daN (JAR 22.143)
6. Equivalent Safety Findings: None

### **B.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition:
 

List of drawings:		
MDM-1 Fox	issue	January 19 <sup>th</sup> 1998
MDM-1P Fox-P	issue	March 1998
  
2. Description:
 

Two-seat (tandem configuration) aerobatic and trainer glider.  
 Cantilever mid-wing monoplane with conventional tail unit (fixed stabilizer with elevator, fin and rudder).  
 All composite glass-epoxy structure.  
 Bipartite tapered wing with Schempp-Hirth airbrakes protruding only from upper surface. Exchangeable wingtips - short for Aerobatic and long for Utility version.  
 Fixed landing gear - main wheel with disk brake and tail wheel.
  
3. Equipment:
 

Minimum equipment:

  - airspeed indicator,
  - altimeter,
  - accelerometer,
  - 5-points pilot harness,
  - parachute or back cushion (8 cm thick) for each crew member.

Standard equipment (besides minimum equipment):

  - variometer with compensator,
  - pressure compensation vessel,
  - bank indicator,
  - compass,
  - seat cushion.
  
4. Dimensions:
 

	Aerobatic	Utility
Span	14,00 m	16,15 m
Wing area	12,34 m <sup>2</sup>	13,09 m <sup>2</sup>
Aspect Ratio	15,88	19,92
Length	7,38 m	7,38 m
Height	2,25 m	2,25 m
  
5. Launching Hook:
 

Nose towing hook	TOST E 85 or SZD-III A56 P
CG towing hook (optional)	TOST Europa G 88
  
6. Weak links:
 

Ultimate Strength:	677 daN (± 10%)
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7. Air Speeds:
 

Manoeuvring Speed	$V_A$	214 km/h
Never Exceed Speed	$V_{NE}$	282 km/h
Maximum permitted speeds		
- in rough air	$V_{RA}$	225 km/h
- in aero-tow	$V_T$	150 km/h
- in winch launching	$V_W$	150 km/h
		<i>winch launching permissible only with short wingtips</i>
  
8. Operational Capability:
 

VFR Day
  
9. Masses:
 

	Aerobatic	Utility
Max. Mass	530 kg	535 kg
Max. Empty Mass (without balance mass)	350 kg	355 kg



10. Centre of Gravity Range: Empty glider with standard equipment:  
 Forward Limit 620 mm aft of datum point  
 Rearward Limit 645 mm aft of datum point  
 Centre of Gravity operational limits:  
 Forward Limit 213 mm aft of datum point (22,0% MAC)  
 Rearward Limit 379 mm aft of datum point (39,0% MAC)  
 MAC is 971 mm for wings with short wingtips  
 MAC is 938 mm for wings with long wingtips  
 0% MAC is 2,1 mm aft of datum.  
 Datum: Leading edge and wing-fuselage division  
 plane intersection.  
 Levelling means: Leading and trailing points of root chord  
 (1308 mm) at the same level.
11. Seating Capacity: 2
12. Lifetime limitations: Refer to Maintenance Manual
13. Other limitations: Solo flight is permissible only on front seat.  
 Cross-country flight (gliding or towed) with short wingtips only with  
 one person crew.  
 No baggage permitted.  
 The following kinds of operation are forbidden:  
 - auto-towing,  
 - winch launching with long wingtips,  
 - flights in icing conditions,  
 - night flying,  
 - cloud flying without turn-and-bank indicator,  
 - aerobatic in rough air  
 - full aerobatic with long wingtips.  
 Manoeuvring load factor limits:  
 Aerobatic version with short wingtips:  
 - two person crew +7,0/-5,0  
 - one person (no more than 100 kg incl. parachute) +9,0/-6,0  
 Utility version with long wingtips +5,3/-2,65
14. Deflection of control surfaces:
- |           |         |     |             |
|-----------|---------|-----|-------------|
| Aileron:  | - up    | 22° | ± 1°        |
|           | - down  | 17° | ± 1°        |
| Elevator: | - up    | 25° | ± 1°        |
|           | - down  | 25° | ± 1°        |
| Rudder:   | - left  | 30° | + 1° / - 2° |
|           | - right | 30° | + 1° / - 2° |

#### **B.IV. Operating and Service Instructions**

1. Flight Manual:

- Polish: Instrukcja Użytkowania w Locie  
Szybowców MDM-1 „Fox” oraz MDM-1P „Fox P”,  
wydanie IV, Grudzień 1998
- English: Flight Manual  
MDM-1 „Fox”, MDM-1P „Fox P”,  
issue IV, December 1998

2. Maintenance Manual:

- Polish: Opis Techniczny, Instrukcja Obsługi Technicznej,  
Prace Okresowe, Szybowca MDM-1P „Fox P”,  
wydanie I, Grudzień 1998
- English: Technical Service Manual, Periodic Works,  
MDM-1P „Fox P”  
issue I, December 1998

#### **B.V. Notes**

1. Serial Numbers:

P-11, P-13, P-14, P-15, P-16,  
201 and following

Previously built MDM-1 "Fox" gliders can be modified into MDM-1P "Fox-P" variant.

2. All glider outside surfaces exposed to sunlight must be white painted apart from registration number and anti-collision marks.