

Swiss Confederation

Federal Office of Civil Aviation FOCA Safety Division - Aircraft

Lufttüchtigkeitsanweisung (LTA) Consigne de Navigabilité (CN) Direttive sulla Navigabilità (DN) Airworthiness Directive (AD)			FOCA AD HB-2025-001-E	
Inkraftsetzung Mise en vigueur Entrata in vigore Effective Date	01 May 2025		Pilatus – PC-21 FOCA TC/TCDS No: F 56-35	
Issue Date:		29 April 2025		
ATA Chapter:		ATA 95 – Crew Escape and Safety		
Subject:		Crew Escape and Safety – Ejection Seats – Barostatic Time Release Unit Sear – Inspection		
Foreign AD:		Not applicable		
Supersedure / Re AD(s):	evised	None		
Type Certificate Holder's Name:		Pilatus Aircraft Ltd.		
Manufacturer(s):		Pilatus Aircraft Ltd.		
Applicability:		Model PC-21 aeroplanes, all Manufacturers Serial Numbers (MSN) with Martin-Baker Mk 16 ejection seats installed.		
Definitions		For the purpose of this AD, the following definitions apply:		
		The Service Bulletin (SB): Pilatus Aircraft PC-21 SB 95-010.		
		The Special Information Leaflet: Martin-Baker Aircraft Company Limited Special Information Leaflet No. 948		
			lease Unit (BTRU) assemblies installed in the front and with an incorrectly cocked sear.	
		Serviceable part: BTRU assemblies installed in the front and rear ejection seats with a correctly cocked sear.		
		Affected aeroplan Any aeroplane hav	e: ing an affected part installed.	
Reason:			reported where after a maintenance event a BTRU ked sear on an ejection seat was detected.	
			ot corrected, could lead to a malfunction of the ejection conditions possibly resulting in severe injuries.	

Lufttüchtigkeitsanweisung (Consigne de Navigabilité (Direttive sulla Navigabilità (Airworthiness Directive (AD	ČN) (DN)	FOCA AD HB-2025-001-E	
	To address this potential unsafe condition, Pilatus issued the SB and Martin-Baker issued the Special Information Leaflet to provide inspection instructions.		
	For the reason described above, this AD requires the inspection of the BTRU in the front and rear ejection seats, as defined in this AD, and prohibits (re-)installation of affected parts.		
Required Action(s)	Required as indicated below, unless already accomplished:		
and Compliance Time(s):	(1) Befo BTR acco	Dection: The next flight after the effective date of this AD, inspect the RU assemblies installed in the front and rear ejection seats in Dordance with the instructions of section 3.B. of the Pilatus SB and Martin-Baker Special Information Leaflet.	
	(2) If, du disci Spec appl	rective action(s): uring the inspection as required by paragraph (1) of this AD, repancies are detected, as identified in the applicable SB and the cial Information Leaflet, before next flight, accomplish the licable corrective action(s) in accordance with the instructions of Special Information Leaflet.	
	(3) Fron asse	ts Installation: n the effective date of this AD, it is allowed to install a BTRU embly in the front and rear ejection seats, provided it is a riceable part, as defined in this AD.	
Ref. Publication(s):	Pilatus Aircraft PC-21 SB 95-010 original issue dated 23 April 2025.		
	Martin-Baker Aircraft Company Special Information Leaflet No. 948 original issue dated 10 April 2025.		
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
For further information contact:	The applicable manufacturer's documentation may be obtained directly from:		
	PILATUS AIRCRAFT LTD. Customer Technical Support (MCC) Pilatusstrasse 1 CH-6371 Stans, Switzerland		
		(0)41 619 30 76 <u>echsupport@pilatus-aircraft.com</u>	
	Website: <u>www.pilatus-aircraft.com</u>		
	For further information contact:		
	FEDERAL OFFICE OF CIVIL AVIATION (FOCA) Airworthiness Engineering (STIL) CH-3003 Bern, Switzerland		

E-mail: airdir@bazl.admin.ch