MASTER MINIMUM EQUIPMENT LIST

DASSAULT AVIATION

FAN JET FALCON (FALCON 20)

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MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

REVISION ONE

This Master Minimum Equipment List (MMEL) is issued by the Civil Aviation Authority at the above revision and is approved as the basis for the preparation and approval of individual operator's Minimum Equipment Lists (MELs) for aircraft of this Type.

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Aircraft Projects MMEL Section

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

REVISION RECORD

REVISION No.	ISSUE DATE	INCORPORATED BY	DATE
Original	15 January 1993		
Revision 1	1 September 1994		

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

TEMPORARY REVISION RECORD

TR No.	Date	Page Affected	Incorporated By	Date Incorporation	Superseded By
1	12/09/2001	31-1 23-2 34-2		moorporaaton	
G4	29/10/2001	TR Record Page Preamble Definitions 23-1 25-1 31-1 34-1			
G6	20/03/2002	34-1			

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

TABLE OF CONTENTS

LIST OF EFFECTIVE PAGES PREAMBLE

NOTES AND DEFINITIONS

- 21 AIR CONDITIONING
- 22 AUTO FLIGHT
- 23 COMMUNICATIONS
- 24 ELECTRICAL POWER
- 25 EQUIPMENT/FURNISHINGS
- 26 FIRE PROTECTION
- 27 FLIGHT CONTROLS
- 28 FUEL
- 29 HYDRAULIC POWER
- 30 ICE AND RAIN PROTECTION
- 31 INDICATING/RECORDING SYSTEMS
- 32 LANDING GEAR
- 33 LIGHTS
- 34 NAVIGATION
- 35 OXYGEN
- 36 PNEUMATIC
- 49 AIRBORNE AUXILIARY POWER
- 52 DOORS
- 73 ENGINE FUEL AND CONTROL
- 74 IGNITION
- 77 ENGINE INDICATING
- 78 EXHAUST
- 79 ENGINE OIL
- 80 STARTING

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

LIST OF EFFECTIVE PAGES

Revision

Date

Page

(i)	Approval Sheet	Revision 1	1 September 1994
(iii)	Revision Record	Revision 1	1 September 1994
(v)	Temporary Revision Record	Revision 1	1 September 1994
(vii)	Table of Contents	Revision 1	1 September 1994
(ix)	List of Effective Pages	Revision 1	1 September 1994
(x)	List of Effective Pages (Cont)	Revision 1	1 September 1994
(xi)	Preamble	Revision 1	1 September 1994
(xii)	Preamble (Cont)	Revision 1	1 September 1994
(xiii)	Definitions	Revision 1	1 September 1994
(xiv)	Definitions (Cont)	Revision 1	1 September 1994
(xv)	Definitions (Cont)	Revision 1	1 September 1994
(xvii)	Highlights to Revision 1	Revision 1	1 September 1994
(xviii)	Highlights to Revision 1 (Cont)	Revision 1	1 September 1994
(xix)	Highlights to Revision 1 (Cont)	Revision 1	1 September 1994
(xx)	Highlights to Revision 1 (Cont)	Revision 1	1 September 1994
	21-1	Revision 1	1 September 1994
	21-2	Revision 1	1 September 1994
	21-3	Revision 1	1 September 1994
	21-4	Revision 1	1 September 1994
	22-1	Revision 1	1 September 1994
	23-1	Revision 1	1 September 1994
	23-2	Revision 1	1 September 1994
	23-3	Revision 1	1 September 1994
	24-1	Revision 1	1 September 1994
	25-1	Revision 1	1 September 1994
	25-2	Revision 1	1 September 1994
	25-3	Revision 1	1 September 1994
	26-1	Revision 1	1 September 1994
	27-1	Revision 1	1 September 1994
	27-2	Revision 1	1 September 1994
	27-3	Revision 1	1 September 1994
	28-1	Revision 1	1 September 1994
	28-2	Revision 1	1 September 1994
	29-1	Revision 1	1 September 1994
	29-2	Revision 1	1 September 1994
	30-1	Revision 1	1 September 1994
	30-2	Revision 1	1 September 1994
	31-1	Revision 1	1 September 1994
	32-1	Revision 1	1 September 1994
			-

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

LIST OF EFFECTIVE PAGES (Cont...)

Page	Revision	Date
33-1	Revision 1	1 September 1994
33-2	Revision 1	1 September 1994
33-3	Revision 1	1 September 1994
33-4	Revision 1	1 September 1994
34-1	Revision 1	1 September 1994
34-2	Revision 1	1 September 1994
34-3	Revision 1	1 September 1994
34-4	Revision 1	1 September 1994
34-5	Revision 1	1 September 1994
34-6	Revision 1	1 September 1994
34-7	Revision 1	1 September 1994
35-1	Revision 1	1 September 1994
35-2	Revision 1	1 September 1994
36-1	Revision 1	1 September 1994
49-1	Revision 1	1 September 1994
52-1	Revision 1	1 September 1994
73-1	Revision 1	1 September 1994
74-1	Revision 1	1 September 1994
77-1	Revision 1	1 September 1994
77-2	Revision 1	1 September 1994
78-1	Revision 1	1 September 1994
79-1	Revision 1	1 September 1994
80-1	Revision 1	1 September 1994

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

PREAMBLE

- 1. The CAA approved Master Minimum Equipment List (MMEL) provides owners/operators of United Kingdom registered aircraft, of the relevant type, with the basis for the preparation of their individual Minimum Equipment List (MELs). In the case of holders of Air Operators Certificates the MEL will be included in that Company's Operations Manual.
- 2. The approved MMEL represents a list of items of equipment which, under particular circumstances, can, to the satisfaction of the CAA, be unserviceable when the aircraft is despatched, while still retaining the required level of safety.
- 3. The CAA recognises that in some respects the standard and scale of equipment provided in the aircraft may exceed the minimum required to satisfy airworthiness or Air Navigation Legislation requirements. Where necessary to achieve a satisfactory level of safety with an inoperative item, appropriate limitations are imposed or the function transferred to another component.
- 4. The MMEL does not include items such as wings, engines and landing gear that are always required, nor is reference made to equipment such as passenger convenience and entertainment items which when inoperative obviously do not affect airworthiness. It is important to note therefore that ANY ITEM WHICH IS RELATED TO THE AIRWORTHINESS OF THE AIRCRAFT AND WHICH IS <u>NOT</u> INCLUDED IN THE MMEL IS ALWAYS REQUIRED TO BE OPERATIVE BEFORE A FLIGHT IS DESPATCHED. Likewise items required by Air Navigation Legislation. Additional Certification Requirements as appropriate, which are not listed must be operative.
- 5. The MMEL may not waive a limitation or an emergency procedure which is given in the Flight Manual (FM) or override an Airworthiness Directive (AD) /Mandatory Inspection unless the FM/AD provides otherwise. Similarly any Additional Certification Requirements, or other special provisions, as appropriate which have been determined as necessary by the CAA shall not be waived unless otherwise agreed or varied by the CAA.
- 6. An Owner/Operators MEL must receive CAA approval which thereby conveys the permission, required by the UK Air Navigation Order, to the Commander, for operation of the aircraft with specified items of equipment unserviceable.
- 7. The MEL may not be less restrictive than the MMEL therefore the number of items required for despatch shall not be less than the corresponding number in column 3 of the MMEL and any associated conditions shall be at least as severe as those specified in column 4.
- 8. The MMEL does not anticipate the effects of combinations of apparently unrelated unserviceabilities or allow for situations where systems are made inoperative for special purposes such as demonstration, test or crew training. Other provisions may apply to positioning or ferrying flights but these may not necessarily be included in the MMEL.
- 9. The MEL should indicate that a decision to operate the aircraft with multiple unserviceabilities should only be made after due consideration of possible interrelated or additive effects and, if necessary, following consultation with appropriate engineering specialists.

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

PREAMBLE (Cont...)

- 10. It is not the purpose of the MMEL to allow defects of other than optional items to remain unrectified indefinitely. The operational flexibility provided under the MMEL policy is justified only within a framework of controlled and sound programmes of repairs, replacement and servicing. Defects should be rectified expeditiously thus retaining the intended overall level of safety and reducing the possibility of a subsequent failure necessitating the removal of the aircraft from service. Some particular items in the MMEL may be subject to a limitation of flight hours, number of flights or consecutive calendar days, and these must be transferred into the MEL. A limit of three calendar days for completion of repairs or replacements has been applied to some items. Other time limits for rectification, such as those specified by the ANO, may also be applied as appropriate. Operators with established routes shall specify in the MEL at which stations, in addition to the main maintenance base, repair facilities exist.
- 11. This MMEL is based upon UK legislation and some of the alleviations it provides may not therefore necessarily comply with foreign legislation.

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

DEFINITIONS

- 1. In this list, the items of equipment are classified in systems according to the ATA 100 specification. Individual items within a given ATA classification are numbered sequentially.
- 2. <u>"Item"</u> (Column 1): The equipment, system, components or function as listed in Column 1.

NOTE: Items annotated in UPPER CASE letters indicates the precise flight deck legend used.

3. <u>"Number Installed"</u> (Column 2): The number of the specified items normally installed in the aircraft. This number identifies the aircraft configuration considered in developing the MMEL.

NOTE: The operator's MEL should list the number installed in a particular aircraft.

- 4. <u>"Number Required for Despatch"</u> (Column 3): The minimum number of the specified items required for operation provided the conditions defined in Column 4 are met.
- 5. <u>"Remarks or Exceptions"</u> (Column 4): This column includes a statement prohibiting operation or permitting operation with a specific number of items inoperative, provisos (conditions and limitations) for such operation and appropriate notes.
- 6. <u>Dash (-)</u>: This symbol indicates a variable quantity when used in Columns 2 or 3.

NOTE: The operator's MEL should list the numbers appropriate to his particular aircraft in Columns 2 and 3.

- 7. Each inoperative item must be placarded to inform and remind the crew members and maintenance personnel of the equipment condition. To the extent practicable, placards should be located adjacent to the control or indicator for the item affected such that it is clear to the operating crew that it or its associated system in inoperative.
- 8. <u>"Inoperative"</u>: A system or item of equipment is deemed inoperative if it malfunctions such that it does not accomplish its intended purpose and/or is not consistently functioning within it's designed operating limit(s) or tolerance(s).
- 9. <u>"(0)"</u>: The use of this symbol in Column 4 indicates that an appropriate operating procedure (or change to an existing procedure) must be established, published and utilised to maintain the required level of safety while operating under the terms of the (M)MEL.

Normally, these procedures are accomplished by the flight crew. However, other personnel may be qualified and authorised to perform certain functions.

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

DEFINITIONS (Cont...)

10. <u>"(M)"</u>: The use of this symbol in Column 4 indicates that an appropriate maintenance procedure must be established, published and utilised prior to the first flight undertaken following discovery of the defect and, if necessary, repeated at specified intervals during operation under the terms of the (M)MEL to maintain the required level of safety.

Normally, these procedures are accomplished by maintenance personnel. However, other personnel may be qualified and authorised to perform certain functions.

- NOTE: Where an item is annotated (0)/(M), the "/" is defined as "and/or", which shows that there may be different options available in respect of the MEL procedures.
- 11. <u>"As required by Air Navigation Legislation"</u>: The associated item must comply with legal provisions such as the Air Navigation Order or any other legislation in force during the flight.
- 12. <u>"VMC" and "IMC"</u>: The definitions of these terms are those used in Section 2 of the Air Navigation Order - Rules of the air.
- 13. <u>"Icing Conditions"</u>: An atmospheric condition that may cause ice to form on the aircraft or in the engines.
- 14. <u>"Visible Moisture"</u>: An atmospheric environment containing water in any form that can be seen in natural or artificial light, i.e. clouds, fog, rain, sleet, hail, snow.
- 15. <u>"Flight Hour"</u>: The time from the moment an aircraft leaves the surface of the earth until it touches it at the next point of landing.
 - NOTE: The definition differs from that given in the Air Navigation Order.
- 16. <u>"ETOPS"</u>: Refers to "extended range" operations which may be defined as "operation of a two-engined aeroplane over a route that contains a point farther than one hour flying time at the normal one-engined inoperative cruise speed (in still air) from an adequate airport".

In the MEL, for an operator who has received approval to extend maximum diversion time from 120 minutes to 138 minutes, unless otherwise stated, "120 minutes" may be interpreted as "138 minutes".

- 17. <u>"Flight day"</u>: A 24 hour period (from midnight to midnight) during which at least one flight is scheduled for the affected aircraft.
- 18. <u>"Authority"</u>: The competent regulatory authority according to the country of registry; for aircraft registered in the U.K. this is the Civil Aviation Authority.
- 19. <u>"Deleted"</u>: When applied to an item number, indicates that the item was previously listed but is now required to be operative.

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

DEFINITIONS (Cont...)

20. <u>"Combustible Material"</u>: is defined as material which is capable of catching fire and burning.

When an MMEL item specifies the condition that only non-combustible materials are to be carried, it is the operator's responsibility to determine that all materials (including containers, packing material and pallets etc) in the associated compartments is of a non-combustible nature.

If it cannot be determined whether any proposed cargo is non-combustible, it must not be loaded in compartments where combustible materials are prohibited.

- 21. <u>"System"</u>: System means the group of directly related components which together performs a specified function, for example 'RPM indication system' would include the RPM indicator, tachometer generator, circuit breaker and associated circuitry.
- 22. <u>"Extended Overwater Flight"</u>: Refers to an operation overwater at a horizontal distance of more than 50 nautical miles from the nearest shoreline.
- 23. <u>Repair Intervals</u>
 - Calendar Day

A period of 24 hours elapsed time, commencing at midnight on the day of discovery and recording of a malfunction in the aircraft's maintenance record/logbook and ending at midnight on the next day. For example, if it were recorded at 10 am on January 26th that a malfunction had occurred, and the MMEL allowed three calendar days for completion of repairs and replacements, the three day interval would commence at midnight on 26th January and end at midnight on 29th January.

24. "Despatch"

The point at which an aircraft first moves under its own power for the purpose of commencing a flight.

- <u>NOTE:</u> The definition above is in accordance with that given in Article 106(2)(a) of the ANO and it is at the point of despatch that the provisions of the MMEL cease to apply. They come into effect again when the aircraft next comes to rest at the end of its flight. In the case of a helicopter which comes to rest without stopping rotors, it is deemed to have ended its flight and the provisions of the MMEL then apply until it is next despatched.
- 25. <u>Not Used</u>: An item which appeared in the base document (e.g. FAA MMEL) but which has been removed from the CAA MMEL. The base document item number is retained to maintain continuity.
- 26. This MMEL is only applicable to the Dassault Aviation Fan Jet Falcon, Models Basic, D, E and F fitted with General Electric CF 700 Engines.
- 27. Base documents used for the preparation of this MMEL are:
 - (a) FAA Falcon MMEL Revision 9 dated 12 October 1993.
 - (b) CAA Policy Statements, as effective at end August 1994.

MASTER MINIMUM EQUIPMENT LIST

REVISION 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

MASTER MINIMUM EQUIPMENT LIST

Revision 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

HIGHLIGHTS OF REVISION 1

General

- 1) In accordance with latest FAA policy the * has been removed from the MMEL see DEFINITIONS item 7.
- 2) A limit of three calendar days has been introduced for completion of repairs or replacements see PREAMBLE Item 10.
- 3) A new DEFINITION "NOT USED" has been introduced see DEFINITION item 25.

21 AIR CONDITIONING

21-8	Cabin Pressurisation Control System		-	Additional alleviation for manual system.
21-14	Electrical Rack Blowers		-	3 day repair policy applied.
21-17	Pilot Foot Warmer		-	New item.
22	AUTO FLIGHT			
22-1	Autopilot Systems		-	Latest CAA policy applied.
23	COMMUNICATIONS			
23-2	Communications Systems	-		UHF added.
23-5	CVR		-	Latest CAA policy applied.
23-10	Radio Master Switch		-	Proviso amended.
24	ELECTRICAL POWER			
24-1	Inverters	-	3 day r	epair policy applied. Proviso (a)

revised and (0) added. 24-3 Batteries - New proviso (b) inserted.

MASTER MINIMUM EQUIPMENT LIST

Revision 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

HIGHLIGHTS OF REVISION 1

25 <u>EQUIPMENT/FURNISHINGS</u>

25-1	Flight Crew Shoulder Harness	-	3 day repair policy applied.
25-4	Pilots Seat Adjustment	-	3 day repair policy applied.
27	FLIGHT CONTROLS		
27-1	Arthur Q Units	-	3 day repair policy applied.
27-2	Arthur Q Unit Warning Light System	-	3 day repair policy applied.
27-3	Air Brake System Warning Lights -		Provisos revised.
27-4	Trailing Edge Flap Indicator Light -		New item.
27-13	Rudder Pedal Adjustment -		New item.

28 FUEL

28-1	Booster Pumps		-	3 day repair policy applied.
28-6	Wing Tank Fuel Quantity Indicators	-		Proviso (a) revised.
28-9	Fuel Temperature Indicator		-	New item.
28-10	Fuel Used System		-	New item.

29 <u>HYDRAULIC POWER</u>

29-6 Hydraulic Reservoir Pressurisation - 3 day repair policy applied, OR added. Warning Light

MASTER MINIMUM EQUIPMENT LIST

Revision 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

HIGHLIGHTS OF REVISION 1

30 **ICE AND RAIN PROTECTION** 30-3 Pitot Heater Systems 3 day repair policy applied. _ 30-4 Pitot Heater Light System 3 day repair policy applied. _ <u>31</u> **INDICATING/RECORDING SYSTEMS** 31-1 Clocks Revised, aircraft clock provides input to FMS. 31-2 FDR Latest CAA policy applied. 32 LANDING GEAR 32-2 Landing Gear Selector Handle Provisos expanded, (0) added. _ Warning Light <u>33</u> LIGHTS 33-1 Flight Compartment and Instrument Latest CAA policy applied. _ Lighting Systems 33-2 Cabin Interior Lighting System Latest CAA policy applied. _ 33-3 Passenger Notice System Proviso (c) added. 33-15 Floor Proximity Escape Path Latest CAA policy applied. Marking Systems 34 NAVIGATION 34-3 Turn and Bank Indicators Title and provisos revised. 34-5 Standby Compass 3 day repair policy applied. _ 34-10 GPWS NOTE expanded. _ 34-16 Mach/Airspeed Warning 3 day repair policy applied. _

MASTER MINIMUM EQUIPMENT LIST

Revision 1 1 SEPTEMBER 1994

FAN JET FALCON (FALCON 20)

HIGHLIGHTS OF REVISION 1

34	Navigation (Cont)		
34-26	GPS/LORAN System	-	GPS added to title.
34-35	Storm Scope		Storm scope not required by ANL.
34-45	TCAS	-	6 sectors/48 hours becomes 10 calendar days.
52	DOORS		
52	DOOKS		
52-1	Door Warning Light System	-	New wording for proviso (b) and 3 day repair policy applied.
73	ENGINE FUEL AND CONTROL		
73-1	Fuel Flowmeters	-	3 day repair policy applied.
77	ENGINE INDICATING		
77-1	Fan RPM indicators	-	3 day repair policy applied.
77-7	Engine Instruments		New item.

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FAI	LCON 20)		DATE: 1 September 1994	21-1
(1) Sys	stem & Sequence Numbers	(2) Numb	er Installe	ed	·
	Item		(3) Nur	mber required for despatch	
				(4) Remarks or Exceptions	
<u>21</u>	AIR CONDITIONING				
1.	Out-Flow/Safety Valves	2	0	(M) One or both may be inoperative provid	ed:
				(a) Flight is conducted in an configuration,	unpressurised
				(b) Flight remains at or below 10,000	feet MSL, and
				(c) Extended overwater flights are pro	hibited.
2.	Cabin Altitude/Overpressure (CAB) Warning Light	1	0	May be inoperative provided:	
				(a) Cabin altimeter operates normally,	
				(b) Cabin altitude warning horn ope and	rates normally,
				(c) Cabin pressure indicator operates r	ormally.
3.	Cabin Altitude/Overpressure Warning Horn	1	0	(M) May be inoperative provided:	
				(a) Flight remains at or below 10,000	feet MSL.
				OR	
				(b) Flight is conducted in an configuration.	unpressurised
4.	Cabin Altitude Indication	1	0	(M) May be inoperative provided:	
				(a) Cabin differential pressure ind normally, and	icator operates
				(b) A chart is provided to convert capressure to cabin altitude.	ibin differential
				OR	
				(c) Flight is conducted in an configuration.	unpressurised
				I	

AIF	RCRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FAI	LCON 20)		DATE: 1 September 1994	21-2
(1) Sys	stem & Sequence Numbers	(2) Number	r Installe	d	
	Item		(3) Nun	nber required for despatch	
			[(4) Remarks or Exceptions	
<u>21</u>	AIR CONDITIONING				
	<u>(Cont)</u>				
5.	Cabin Differential Pressure	1	0	(M) May be inoperative provided:	
	Indicator			(a) Cabin altimeter operates normally,	and
				(b) A chart is provided to convert ca cabin differential pressure.	bin altitude to
				OR	
				(c) Flight is conducted in an	unpressurised
				configuration.	1
6.	Cabin Rate of Climb Indicator	1	0	(M) May be inoperative provided:	
				(a) Automatic pressurisation control s normally, and	ystem operates
				(b) Cabin differential pressure indica altimeter operate normally.	tor and cabin
				OR	
				(c) Flight is conducted in an	unpressurised
				configuration.	1
7.	Electronic Temperature Control	2	1	(M) Automatic control may be inoperative p	rovided
	Systems		0	manual electric temperature control operates	-
		2	0	(M) Both automatic and manual temperatur be inoperative provided manual tempe (located in aft cabin) operates normally an departure in accordance with an approved pr	rature control d is set before
8.	Cabin Pressurisation Control				
0.	System				
	(1) Automatic	1	0	(M) (0) May be inoperative provided:	
				(a) Manual cabin pressure control sy normally, and	vstem operates
				(Cont)	

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	21-3
(1) Sys	tem & Sequence Numbers	(2) Numbe	er Installe	ed	
	Item] [(3) Nur	nber required for despatch	
				(4) Remarks or Exceptions	
<u>21</u>	AIR CONDITIONING (Cont)				
	<u>(••••••••)</u>				
8.	Cabin Pressurisation Control System (Cont)				
	(1) Automatic (Cont)				
				(b) Cabin altimeter and cabin differentia indicator operate normally.	l pressure
				OR	
				(c) Flight is conducted in an unp	pressurised
				configuration.	
	(2) Manual	1	0	(M) (0) May be inoperative provided:	
				(a) Automatic cabin pressure control system normally, and	m operates
				(b) Cabin altimeter and cabin differentia indicator operate normally.	l pressure
				OR	
				(c) Flight is conducted in an unp	pressurised
				configuration.	
0	Cabin Temperature Remote	1	0	May be inoperative.	
9.	Control		0	May be moperative.	
10.	Temperature Control Valve Position Indicator (If Installed)	1	0	May be inoperative.	
	Position indicator (11 instance)				
11.	Cabin Temperature Indicator	1	0	May be inoperative.	
11.			0	May be moperative.	
12.	Cabin Air Conditioning Valve	1	0	(M) (0) May be inoperative in the CLOSEI	D position
	J			provided:	
				(a) Flight is conducted is an unp configuration, and	pressurised
				(Cont)	
		1 I		•	

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FAL	CON 20)		DATE: 1 September 1994	21-4
(1) Sys	stem & Sequence Numbers	(2) Number	r Install	ed	
	Item		(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>21</u>	AIR CONDITIONING				
	<u>(Cont)</u>				
12.	Cabin Air Conditioning Valve (Cont)				
				(b) Cabin ram air scoop is OPEN.	
		1	0	(0) May be inoperative in the full OPEN posit	ion provided:
				(a) Both bleed valves operate normally,	
				(b) Both bleed valves are selected OFF f	or take off,
				(c) Cabin Altitude/Overpressure Wa operates normally, and	rning Light
				(d) Cabin Altitude/Overpressure Wa operates normally.	rning Horn
13.	Cockpit Gasper Outlet	2	1	One may be inoperative.	
14.	Electrical Rack Blowers	2	1	(0) One may be inoperative provided:	
				(a) Flight Manual specifically provid operation,	es for such
				(b) Flight Manual limitations are observed	ed, and
				(c) Repairs or replacements are carrie three calendar days.	d out within
15.	Flood Duct System	1	0	May be inoperative.	
	·				
16.	Ram Air Scoop	1	0	(0) May be inoperative in the OPEN position.	
	F		-		
17.	Pilot Foot Warmer	1	0	May be inoperative CLOSED (Bleed A (Electric).	ir) or OFF
				I	

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	22-1
(1) Sys	tem & Sequence Numbers	(2) Numb	per Installe		
	Item		(3) Nur	nber required for despatch	
		1	(0) 1401		
				(4) Remarks or Exceptions	
22	AUTO FLIGHT				
	AUTOTEIOIII				
1.	Autopilot Systems	-	-	May be wholly or partially inoperative for Pul Transport Operations provided the composition Flight Crew is in accordance with the appropri- requirements of Air Navigation Legislation or arrangements approved by the Authority for a this type.	on of the riate
		-	0	May be inoperative for aircraft flying for p than public transport.	urposes other
				NOTE: Any mode which functions normally	may be used.
2.	AUTO PILOT DISENGAGE Lights	-	1	Any in excess of the minimum of one required flights that require the autopilot may be inope	1 for rative.
3.	Control Wheel Autopilot Disengage Button	2	1	One may be inoperative provided autopilot is below 1500 feet AGL.	not used
4.	Yaw Damper	1	0	May be inoperative provided Flight Manual I not require its use.	Procedures do

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATION					REVISION NO: REVISION 1 PAGE:			
		FAN JET FALCON (FAL	CON 20)		DATE: 1 September 1994 23-1			
(1) Sys	stem & Seo	quence Numbers	(2) Number	r Installe	-			
		Item		(3) Nur	mber required for despatch			
				. ,	(4) Remarks or Exceptions			
<u>23</u>	COM	IMUNICATIONS						
1.	Passen	nger Address (PA) System	1	0	(0) May be inoperative provided:			
	(If Inst	taned)			(a) Cabin Interphone System is operative, and			
					(b) Chime System is operative, and			
					(c) Alternate normal and emergency procedures are established and utilised, and			
					(d) Aircraft may continue the flight or series of flights but shall not depart an airport where repairs can be made and shall not exceed 25 flight hours prior to completion of repairs.			
2.	Comm	unication Systems						
	(1)	HF	-	-	As required by Air Navigation Legislation.			
	(2)	VHF	-	-	As required by Air Navigation Legislation.			
	(3)	UHF	-	-	May be inoperative.			
3.	Cockp	it Speakers						
	(1)	Communications	2	0	One or both may be inoperative for communication purposes provided each crew member on flight deck duty has a serviceable headset.			
	(2)	Aural Warning Alerts	2	0	One or both may be inoperative provided all appropriate aural alert functions are operating normally and the associated audible warnings are available to the crew, by means other than loudspeakers.			
4.	Servic	e Interphone System Flight	1	0	May be inoperative provided:			
	Deck to Cabin/Cabin to Flight Deck/Flight Deck to Ground			(a) Alternate normal and emergency procedures are established and used, and				
					(b) PA System operates normally.			

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FAI	LCON 20)		DATE: 1 September 1994	23-2
(1) Sys	tem & Sequence Numbers	(2) Numb	per Install	ed	
	Item	r	(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>23</u>	COMMUNICATIONS (Cont)				
5.	Cockpit Voice Recorder (CVR) System (If Installed)	1	0	As required by Air Navigation Legislation. If may be inoperative provided:	required,
				(a) It is not reasonably practical to repa before commencement of the flight.	ir or replace
				(b) The aircraft shall not exceed six (6) flights with the CVR unserviceable with the first flight after the CV operating throughout the flight.	e beginning
				(c) Not more than 48 hours have elaps CVR became unserviceable.	ed since the
				(d) The aircraft must not depart from its base with the CVR unserviceable, and	maintenance 1
				(e) The Flight Data Recorder (if requ Navigation Legislation) must be normally.	ired by Air e operating
6.	Flight Phone (If Installed)	-	-	May be inoperative.	
7.	Selective Call System (Selcal) (If Installed)	-	-	May be inoperative provided:	
				(a) Procedures do not require its use. OR	
				(b) Flight Crew monitor appropriate radio	o frequency.
8.	Pre-Recorded Passenger Announcement System (If Installed) (Automatic Cabin Briefer)	-	-	May be inoperative.	
9.	Hand Held Microphones	-	-	Any or all may be inoperative.	

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)				REVISION NO: REVISION 1	PAGE:				
				DATE: 1 September 1994	23-3				
(1) Sy	stem & Sequence Numbers	(2) Numl	ber Instal	Installed					
	Item	-	(3) Nu	Number required for despatch					
				(4) Remarks or Exceptions					
<u>23</u>	COMMUNICATIONS (Cont)								
10.	Radio Master Switch	1	0	(0) May be inoperative ON.					
11.	Headsets	-	-	One Headset (including boom microph operative for each crew member on flight de	oone) must be eck duty.				
12.	Audio Selector Panels	-	-	One required for each crew member on flig	ht deck duty.				
13.	Crew Inter-Communication System	-	1	As required by Air Navigation Legislation.					
14.	Passenger Entertainment Systems (If Installed)	-	0	May be inoperative.					
15.	Cabin Chime System	1	0	(0) May be inoperative provided passenger is operating normally and procedures to inf of any requirement for the need for "No S to "Fasten Seat Belts" are established and us	form passengers moking" and/or				
16.	ACARS (If Installed)	-	0	May be inoperative.					

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT: DASSAULT AVIATION		REVISION NO: REVISION 1	AGE:
	FAN JET FALCON (FALC	ON 20)	DATE: 1 September 1994	24-1
(1) Sys	tem & Sequence Numbers	(2) Number	r Installed	
	Item		(3) Number required for despatch	
		1 `		
			(4) Remarks or Exceptions	
24	ELECTRICAL POWER			
1	In and and			
1.	Inverters3	2	(0) One may be inoperative provided:	
			(a) Remaining two inverters are verifie operating normally before take-off, and	d to be
			(b) Repairs or replacements are carried ou three calendar days.	ıt within
2.	DC Voltmeter	1	0 May be inoperative provided all other generator components operate normally.	r system
3.	Batteries2	1	(M) One may be inoperative for day VMC operation provided:	ons only
			(a) It is disconnected from the electrical syste	em,
			(b) Both generators operate normally, and	
			(c) The aircraft may continue the flight or	series of
			flights not exceeding one Flight day and depart an airport where repairs or repla can be made.	shall not
			<u>NOTE</u> 1: With one battery disconnected it possible to start the engines other the external electrical power.	
			<u>NOTE</u> 2: Starter assisted engine re-light will possible.	not be
4.	Battery Temperature Indicator System	-	0 May be inoperative provided battery temperature warning light system is installed and operates norm	nally.
5.	Generator Warning Lights	2	1 (0) One may be inoperative provided the D.C. voor operates normally, and is regularly monitored during	

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATION				REVISION NO: REVISION 1 PAGE:			
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	25-1		
(1) Sys	tem & Sequence Numbers	(2) Numb	er Install	*			
	Item	Г	(3) Nu	mber required for despatch			
		1		(4) Remarks or Exceptions			
25	EQUIPMENT/ FURNISHINGS						
1.	Flight Crew Shoulder Harness	-	-	As required by Air Navigation Legislation.			
	(1) Inertia Reels	-	-	May be inoperative provided:			
				(a) The affected harness is adjusted and approved means to suit the requir- individual Flight Crew Member, and	ements of the		
				(b) Repairs or replacements are carried three calendar days.	ed out within		
2.	Emergency Locator Transmitter	-	-	As required by Air Navigation Legislation.			
3.	Passenger Seats (Including Seat Backs)	-	-	(M) May be inoperative secured in upright po	osition.		
	Seat Dacks)	-	0	(M) One or more may be inoperative provide	d:		
				(a) Affected seat does not block an eme	rgency exit,		
				(b) Does not restrict any passenger from main aircraft aisle, and	access to the		
				(c) Affected seat(s) is blocked and p NOT OCCUPY".	lacarded "DO		
				NOTE 1: A seat with an inoperative considered to be inoperative.	seat belt is		
				NOTE 2: A seat with an inoperative reclin is considered to be inoperative in cannot be secured in the upright	f the seat back		
				NOTE 3: Inoperative seats do not affect to Cabin Crew required by Air Legislation.			
4.	Pilots Seat Adjustment	1	1	Fore and Aft adjustment must operate normal	ly.		
		1	1	(M) Vertical and/or recline adjustments may provided:	be inoperative		
				(Cont)			

AIRCRAFT: DASSAULT AVIATION				REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	25-2
(1) Sys	tem & Sequence Numbers	(2) Numb	er Instal		I
	Item	Г	(3) Nu	imber required for despatch	
			()	(4) Remarks or Exceptions	
25	EQUIPMENT/ FURNISHINGS (Cont)				
4.	Pilots Seat Adjustment (Cont)				
	,			(a) The seat is secured and locked in a p the individual pilots requirements, an	
				(b) Repairs or replacements are carrie three calendar days.	d out within
5.	Drag Chute	1	0	(M) May be inoperative or removed provided:	
				(a) Weight and balance are verified in with an approved procedure,	n accordance
				(b) Flight Manual Limitations are observ	ed, and
				(c) Drag chute container and cap must be	e installed.
6.	Passenger Convenience Items			See Preamble Paragraph 4.	
7.	Cabin Attendant/Observers Seat and Harness (If Installed)	-	0	(M) (0) As required by Air Navigation Legisla May be inoperative provided the seat is not can be correctly stowed.	ation. required and
				<u>NOTE</u> 1: A folding seat that will not stow or remain stowed is consid inoperative and shall be sec retracted position or removed.	lered to be
				<u>NOTE</u> 2: A seat with a defective harness to be inoperative and shall be prohibit occupancy.	
8.	Flight Crew Smoke Protection Equipment (Basic and Portable)	-	-	As required by Air Navigation Legislation. In specified items may be missing or inoperative accordance with arrangements approved by th	in

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)				REVISION NO: REVISION 1 DATE: 1 September 1994	PAGE: 25-3
(1) Sys	tem & Sequence Numbers	(2) Numb	ber Instal	*	23 3
Item		г	(2) NI		
		1	(3) NI	umber required for despatch	
				(4) Remarks or Exceptions	
25	EQUIPMENT/ FURNISHINGS (Cont)				
9.	Crew Flash-Light	-	-	As required by Air Navigation Legislation.	
10.	Passenger Seat Ashtrays	-	-	All may be inoperative or missing provided:	
				(a) Affected ashtray opening is covered,	and
				(b) Associated seat is placarded " <u>No</u> smoking is prohibited on the entire ai	<u>Smoking</u> " or rcraft.
11.	First Aid Kits	-	-	As required by Air Navigation Legislation.	
12.	Life Jackets	-	-	As required by Air Navigation Legislation.	
13.	Life Rafts	-	-	As required by Air Navigation Legislation.	
14.	Survival Kits	-	-	As required by Air Navigation Legislation.	
MASTER MINIMUM EQUIPMENT LIST

AIR	AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)			REVISIC		EVISION 1	PAGE: 26-1
(1) Svs	tem & Sequence Numbers	(2) Numb	er Installe		1 Septemb	er 1994	20-1
	Item	(_)					
		-	(3) Nur	mber require	d for despatc	h	
				(4) Remar	ks or Excepti	ons	
<u>26</u>	FIRE PROTECTION						
1.	Fire Extinguisher Thermal Discharge Discs	2	0	associat indicato	One or both ted fire extinors are chech dequate cha	n may be missing providenguisher bottle pressure red before each departure rge.	ed the e to
2.	APU Fire Warning System	1	0	May be (a) (b)		provided: t used, and s do not require its use.	
3.	APU Fire Extinguisher System	1	0	May be (a) (b)		e provided: t used, and s do not require its use.	
4.	Hand Held Fire Extinguishers	-	-	As re Extingu inopera	ishers in ex	Airworthiness Not access of the minimum re	ice No. 60. equired may be

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT: DASSAULT AVIATION	(ON 20)		REVISIO	DN NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	UN 20)		DATE:	1 September 1994	27-1
(1) Syst	em & Sequence Numbers	(2) Numb	ber Install	led		
	Item		(3) Nu	mber require	ed for despatch	
				(4) Rema	rks or Exceptions	
<u>27</u>	FLIGHT CONTROLS					
1.	Arthur Q Units (Aileron and Elevator)	2	0	(M) (0)	One or both may be inoperative pr	ovided:
				(a)	Arthur Q units are verified to be position in accordance with maintenance procedure,	in the low speed an approved
				(b)	Airspeed remains at or below Mach,	250 KIAS8
				(c)	The aircraft is operated with disengaged,	h the autopilot
				(d)	Flight Manual Abnormal observed, and	Procedures are
				(e)	Repairs or replacements are ca three calendar days.	rried out within
2.	Arthur Q Unit Warning Light	2	0	(M) (0)	One or both may be inoperative pr	ovided:
	System			(a)	Associated Arthur Q unit(s) are the low speed position in accorapproved maintenance procedure	ordance with an
				(b)	Airspeed remains at or below Mach,	250 KIAS - 8.
				(c)	The aircraft is operated with disengaged,	h the autopilot
				(d)	Flight Manual Abnormal observed, and	Procedures are
				(e)	Repairs or replacements are ca three calendar days.	rried out within
3.	Air Brake System Warning Lights	2	1	(0) One	e may be inoperative provided:	
				(a)	Airbrakes operate normally,	
				(b)	Verify airbrakes are properly set and	before departure,
				(c)	T/O configuration warning operation	tes normally.

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	UN 20)		DATE: 1 September 1994	27-2
(1) Sys	stem & Sequence Numbers	(2) Numb	per Install	ed	
	Item		(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>27</u>	FLIGHT CONTROLS				
	<u>(Cont)</u>				
4.	Trailing Edge Flap Indicator Light	1	0	May be inoperative provided flap position indi- operates normally.	cator
5.	Leading Edge Device Indicator Light	1	0	(0) May be inoperative provided:	
				(a) Position of leading edge devices checked before each departure and commanded movement of the le devices in flight,	l after each
				(b) All other flap position indicators operate normally, and	and lights
				(c) For night flights, wing ice detection hormally.	ights operate
6.	Stall Warning Horn Test System	1	0	(M) (0) May be inoperative provided the s system is verified to operate normally departure.	tall warning before each
7.	Flap Bypass System	1	0	(0) May be inoperative provided:	
				(a) AMD SB 537 is installed,	
				(b) Manual flap system operates normally	, and
				(c) Standby pump operates normally.	
8.	Horizontal Stabiliser Trim Indicator	1	0	(0) May be inoperative provided:	
				(a) Takeoff trim system is visually check reference marks on the vertical stab each departure, and	
				(b) Audio trim in motion warning system and operates normally.	ı is installed

AIF	RCRAFT: DASSAULT AVIATION FAN JET FALCON (FALC	CON 20)		REVISIO	ON NO: REVISION 1	PAGE:
				DATE:	1 September 1994	27-3
(1) Sy	stem & Sequence Numbers	(2) Numb	per Install	led		
	ltem		(3) Nu	mber require	ed for despatch	
				(4) Rema	rks or Exceptions	
27	FLIGHT CONTROLS					
	<u>(Cont)</u>					
9.	Horizontal Stabiliser Trim Takeoff Warning System	1	0	(0) Lig	ht may be inoperative provided:	
				(a)	Stabiliser trim is checked to be in before each departure, and	proper positio
				(b)	Horizontal stabiliser trim pos operates normally.	sition indicato
10.	Horizontal Stabiliser Trim	1	0	(0) Ma	y be inoperative provided:	
Operating Aural Alert (Clacker)	Operating Aural Alert (Clacker)			(a)	Horizontal stabiliser trim pos operates normally,	sition indicato
				(b)	Trim is monitored during takeoff,	and
				(c)	Autopilot is not used.	
11.	Aileron Trim Indicator	1	0	(0) Ma	y be inoperative provided:	
				(a)	Trim is verified to operate norm full range of operation,	ally through it
				(b)	Trim System operates normally, an	nd
				(c)	Trim is positioned to neutral departure and neutral position is v inspection.	pior to eac erified by visua
12.	Rudder Trim Indicator	1	0	(0) Ma	y be inoperative provided:-	
				(a)	Trim is verified to operate norm full range of operation,	ally through it
				(b)	Trim System operates normally an	d
				(c)	Trim is positioned to neutral departure and neutral position is v inspection.	
13.	Rudder Pedal Adjustment	-	-	(M) Al	l may be inoperative provided:	
	Mechanism			(a)	The flight crew individual re satisfied and rudder and brake normally, and	
				(b)	Repairs or replacements are car three calendar days.	ried out withi

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT: DASSAULT AVIATION FAN JET FALCON (FALC	'ON 20)		REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	28-1
(1) Sys	tem & Sequence Numbers	(2) Numb	per Install	ed	
	Item		(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>28</u>	FUEL				
1.	Booster Pumps	2	1	(M) (0) One may be inoperative provided:	
				(a) Both transfer pumps operate normal	ly,
				(b) Feeder tank pressurisation is no checked before each departure,	ormal and is
				(c) Crossfeed valve operates normally,	
				(d) Flight Manual provisions for eng without assistance of any boost	gine operation for pump are
				observed, and	
				(e) Repairs or replacements are carri three calendar days.	ed out within
2.	Transfer Pumps	2	1	(M) One may be inoperative provided:	
	-			(a) Both booster pumps operate normal	ly,
				(b) Wing interconnect operates normal to prevent fuel imbalance, and	ly and is used
				(c) Crossfeed valve operates normally.	
3.	Low Fuel Pressure Warning Lights	2	0	(0) One or both may be inoperative provided fuel pressure indicators operate normally.	the associated
4.	Fuel Pressure Indicators	2	0	(0) One or both may be inoperative provided fuel low pressure warning light(s) operate no	the associated rmally.
5.	Wing Interconnect System	1	0	(0) May be inoperative provided all othe components operate normally.	r fuel system
6.	Wing Tank Fuel Quantity Indicators	1	0	 (M) One tank indication may be inoperative p (a) Alternate procedures are established to ensure fuel is adequate for op conducted, (Cont) 	ed and utilised

FAN JET FALCON (FALCON 20) DATE: 1 September 1994 28-2 (1) System & Sequence Numbers Item (2) Number Installed (4) Remarks or Exceptions 28 FUEL (Cont) (4) Remarks or Exceptions 6. Wing Tank Fuel Quantity Indicators (Cont) (b) Both Fuel Flow/Fuel Used indicators operate normally, and (c) The aircraft may continue the flight or series of flights but shall not exceed 6 sectors prior to the completion of replacements or repairs. 7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided: they are verified in the looked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally, and (b) All cockpit fuel quantity indicators operate crossfeed operatic normally, and (c) The ullsed System 1 0 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems operate normally.	AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
(1) System & Sequence Numbers (2) Number Installed (1) System & Sequence Numbers (2) Number Installed (2) Number Installed (3) Number required for despatch (4) Remarks or Exceptions (4) Remarks or Exceptions (5) Wing Tank Fuel Quantity Indicators (b) Both Fuel Flow/Fuel Used indicators operate normally, and (c) The aircraft may continue the flight or series of flights but shall not exceed 6 sectors prior to the completion of replacements or repairs. 7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 9. Fuel Temperature Indicator (If installed) 1 0 (M) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems		FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	28-2
(3) Number required for despatch (3) Number required for despatch (4) Remarks or Exceptions (5) Wing Tank Fuel Quantity Indicators (Cont) (6) Wing Tank Fuel Quantity Indicators (Cont) (7) Wing Fuel Tank Dip Sticks (8) Pressure Fuelling (1) 0 (M) May be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally. 9. Fuel Temperature Indicator (If installed) 1 0 10. Fuel Used System 1 0	(1) Sys	tem & Sequence Numbers	(2) Number	r Install	*	<u> </u>
28 FUEL (Cont) 6. Wing Tank Fuel Quantity Indicators (Cont) (b) Both Fuel Flow/Fuel Used indicators operate normally, and (c) The aircraft may continue the flight or series of flights but shall not exceed 6 sectors prior to the completion of replacements or repairs. 7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally. 9. Fuel Temperature Indicator (If installed) 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems		Item		(3) Nu	mber required for despatch	
28 FUEL (Cont) 6. Wing Tank Fuel Quantity Indicators (Cont) (b) Both Fuel Flow/Fuel Used indicators operate normally, and (c) The aircraft may continue the flight or series of flights but shall not exceed 6 sectors prior to the completion of replacements or repairs. 7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally. 9. Fuel Temperature Indicator (If installed) 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems				(0) 144		
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7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are completion of replacements or repairs. 7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: 8. Pressure Fuelling 1 0 (M) May be inoperative provided: 9. Fuel Temperature Indicator (If installed) 1 0 (0) May be inoperative provided: 9. Fuel Temperature Indicator 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems	6.					
7. Wing Fuel Tank Dip Sticks 2 0 (M) One or both may be inoperative provided they are verified in the locked CLOSED position. 8. Pressure Fuelling 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally. 9. Fuel Temperature Indicator 1 0 (0) May be inoperative provided:						ators operate
 8. Pressure Fuelling 9. Fuel Temperature Indicator (If installed) 9. Fuel Verified in the locked CLOSED position. 1 0 (M) May be inoperative provided: (a) The system is secured and deactivated in accordance with an approved procedure, and (b) All cockpit fuel quantity indicators operate normally. 9. Fuel Temperature Indicator 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems 					flights but shall not exceed 6 sector	s prior to the
 9. Fuel Temperature Indicator (If installed) 9. Fuel Temperature Indicator (If installed) 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems 	7.	Wing Fuel Tank Dip Sticks	2	0	(M) One or both may be inoperative proviverified in the locked CLOSED position.	ded they are
 9. Fuel Temperature Indicator (If installed) 9. Fuel Verder System 1 1 0 0<td>8.</td><td>Pressure Fuelling</td><td>1</td><td>0</td><td>(M) May be inoperative provided:</td><td></td>	8.	Pressure Fuelling	1	0	(M) May be inoperative provided:	
 9. Fuel Temperature Indicator (If installed) 9. Fuel Temperature Indicator (If installed) 1 0 (0) May be inoperative provided: (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems 					(a) The system is secured and de accordance with an approved procedu	activated in ure, and
 (If installed) (a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 					(b) All cockpit fuel quantity indica normally.	tors operate
(a) Boost pumps, transfer systems and engine crossfeed operate normally, and (b) Operations are not conducted at an RAT below the fuel freeze point. 10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems	9.	Fuel Temperature Indicator	1	0	(0) May be inoperative provided:	
10. Fuel Used System 1 0 May be inoperative provided all fuel quantity systems		(If installed)				and engine
					(b) Operations are not conducted at an the fuel freeze point.	RAT below
	10.	Fuel Used System	1	0	May be inoperative provided all fuel quan operate normally.	ntity systems

AIF	RCRAFT				REVISION NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	LON 20)		DATE: 1 September 1994	29-1
(1) Sy	stem & Se	quence Numbers	(2) Num	ber Install	led	_
		Item		(3) Nu	mber required for despatch	
					· · ·	
					(4) Remarks or Exceptions	
<u>29</u>	HYD	RAULIC POWER				
1.	Hydra (Dual	ulic Quantity Indicators Indicating)	2	0	(M) (0) One or both quantity indications may inoperative provided:	be
					(a) Hydraulic quantity in the associated filled to the manufacturers recommender using an approved procedure departure, and	reservoir(s) is nded capacity before each
					(b) All other hydraulic system indicator lights operate normally.	s and warning
2.	Main Lights	System Pressure Warning s (HYDR1 and HYDR2)	2	0	One or both may be inoperative provided all on hydraulic system pressure indicators and we operate normally.	other varning lights
3.	Stand (Arrov	by Pump Warning Light ws)	2	0	One or both may be inoperative provided all on hydraulic system pressure indications and operate normally.	other waring lights
4.	Syster (Fligh 1 and	n Pressure Warning Lights t Control Circuit Red Light 2)	2	0	One or both may be inoperative provided all hydraulic system pressure indications and wa operate normally.	other rning lights
5.		ulic Pressure Indicator e Indicating)				
	(1)	Main Pressure Systems	2	0	One or both main system pressure functions r inoperative provided the associated main sy light(s) operate normally.	nay be stem warning
	(2)	Standby Pressure System	1	0	May be inoperative provided all other main system warning lights and pressure gaug operate normally.	

AIRCRAFT: DASSAULT A FAN JET FALC	VIATION CON (FALCON 20)		REVISION NO: REVISION 1	PAGE:
(1) System & Sequence Numbers	(2) Numbe	er Installe	DATE: 1 September 1994 ed	29-2
Item		(3) Nun	mber required for despatch	
			(4) Remarks or Exceptions	
29 HYDRAULIC POWI (Cont)	<u>ER</u>			
6. Hydraulic Reservoir Press Warning Light	urisation 1	0	(M) (0) May be inoperative provided:	
			(a) Repairs or replacements are of three calendar days, and	arried out within
			(b) Reservoir pressurisation is detenormally using an approved each departure,	procedure before
			OR	
			(c) The flight is operated at or b MSL.	below 12,000 feet

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	30-1
(1) Sys	tem & Sequence Numbers	(2) Number	r Installe	*	501
	Item		(2) Nices		
		-	(3) Nur	nber required for despatch	
				(4) Remarks or Exceptions	
20					
<u>30</u>	ICE AND RAIN PROTECTION				
1.	Airframe Anti-Icing System		0	May be inoperative CLOSED provided the a	aircraft is not
1.			Ũ	operated in known or forecast icing conditions	3.
2.	Engine Inlet Anti-Icing Systems	2	1	One may be inoperative CLOSED provided	the aircraft is
				not operated in known or forecast icing condit	ions.
3.	Pitot Heater Systems	2	1	One may be inoperative provided:	
				(a) The aircraft is not operated into vis	ible moisture
				or known or forecast icing conditions	
				(b) Repairs or replacements are carrie three calendar days.	d out within
4.	Pitot Heater Light System	2	1	(0) One may be inoperative provided:	
ч.	Thot freder Eight System		1	(a) The associated heater element operat	as normally
				(b) The aircraft is not operated in know icing conditions, and	'n or forecast
				(c) Repairs or replacements are carrie	d out within
				three calendar days.	
5.	Static Heater Systems	2	1	One may be inoperative provided:	
				(a) The aircraft is not operated in know icing conditions, and	'n or forecast
				(b) Taxiway or runway is not covered	with standing
				water or slush.	
6.	Windshield Heating Systems	2	1	(0) One may be inoperative provided:	
				(a) The aircraft is not operated in know	n or forecast
				icing conditions,	
				(Cont)	
		1 I		1	

AIR	CRAFT: DASSAULT AVIATION	$(\mathbf{ON}, 20)$		REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	30-2
(1) Sys	tem & Sequence Numbers	(2) Num	ber Installe	ed	L
	Item		(3) Nur	mber required for despatch	
			(*)		
				(4) Remarks or Exceptions	
<u>30</u>	ICE AND RAIN				
	ICE AND RAIN PROTECTION (Cont)				
6.	Windshield Heating Systems				
	(Cont)				
				(b) Windshield de-fog system operates no	ormally, and
				(c) Flight Manual Procedures are observe	ed.
7.	Side Window Heating System	1	0	May be inoperative.	
8.	Windshield Wipers	2	0	(0) One or both may be inoperative provided	the aircraft is
0.	windsment wipers	2		not flown in precipitation within arrival a	nd departure
				areas.	
9.	Rain Repellent Systems (If Installed)	2	0	One or both may be inoperative.	
	(II IIIbuileu)				
10					
10.	Defog System	1	0	May be inoperative provided windshield heat operate normally.	iting systems
11.	Angle of Attack Heating System	1	0	May be inoperative provided the aircraft is not	r
11.	(for Speed Index) (If Installed)			operated in known or forecast icing conditions	•
12.	Stall Warning Sensor Heating	2	1	(0) One may be inoperative provided the aircra	aft is not
	System			operated in known or forecast icing conditions	•

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	UN 20)		DATE: 1 September 1994	31-1
(1) Sys	stem & Sequence Numbers	(2) Numb	ber Installe	ed	·
	ltem		(3) Nur	mber required for despatch	
				(4) Remarks or Exceptions	
<u>31</u>	INDICATING/RECORDING SYSTEMS				
	<u>••••</u>				
1.	Clocks	-	1	One may be inoperative at either the pilo station.	s or co-pilots
2.	Flight Data Recorder (FDR) System	1	0	As required by Air Navigation Legislation. I may be inoperative provided:	f required,
				(a) It is not reasonably practical to republic before commencement of flight.	bair or replace
				(b) The aircraft shall not exceed six (flights with the FDR unservicea with the first flight after the F operating throughout the flight.	ble beginning
				(c) Not more than 48 hours have elay FDR became unserviceable.	osed since the
				(d) The aircraft may not depart from it base with the FDR unserviceable.	s maintenance
				(e) The Cockpit Voice Recorder (if re Navigation Legislation) must normally.	quired by Air be operating
3.	TAS Indicator	1	0	May be inoperative.	
4.	SAT/TAT/TAS Indicator	1	0	(0) May be inoperative provided RAT indinormally.	cator operates
5.	RAT Indicator	1	0	(0) May be inoperative provided air data associated SAT/TAT indicator system are operate normally.	

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT: DASSAULT AVIATION	CON 20)		REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	LON 20)		DATE: 1 September 1994	32-1
(1) Sys	tem & Sequence Numbers	(2) Numb	ber Installe		
	Item		(3) Nur	nber required for despatch	
				(4) Remarks or Exceptions	
<u>32</u>	LANDING GEAR				
1.	Anti-Skid System	1	0	(0) May be inoperative provided appropriate performance decrements are applied.	Flight Manual
2.	Landing Gear Selector Handle Warning Light	1	0	May be inoperative provided:	
				(a) The landing gear position indic normally, and	ators operate
				(b) Audible warning is verified to normally.	be operating
3.	Parking Brake Annunciator Light (P Brake)	1	0	(M) (0) May be inoperative provided:	
	(r blake)			(a) Emergency Brake accumulator press normal before each departure, and	ure is verified
				(b) The No. 2 braking system operates n	ormally.

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT:	DASSAULT AVIATION			REVISIC	ON NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	ON 20)		DATE:	1 September 1994	33-1
(1) Sys	tem & Sequ	ence Numbers	(2) Numb	ber Install	ed	*	I
		Item		(3) Nu	mber require	d for despatch	
						ks or Exceptions	
33	LIGHT	S					
1.	Flight D Lighting	eck and Instrument Systems	-	0	As requiinopera	ired by Air Navigation Legislati tive for daylight operations only	on. May be
					OR		
			-	-		uired by Air Navigation Legis nay be inoperative provided:	slation. Individual
					(a)	Sufficient lighting is operat required instrument, control, a which it is provided easily read	and other device for
					(b)	Sufficient flight deck emer operative.	gency lighting is:
					(c)	Lighting configuration at dispatche flight crew.	tch is acceptable to
2.	Cabin In	terior Lighting System	-	-		uired by Air Navigation Leg tive provided: Lighting is adequate for the ca their required duties, and Cabin emergency lighting is op	bin crew to perform
						OR	
					(c)	Passengers are not carried.	
					NOTE:	Cabin emergency lighting doe proximity lights.	es not include floor
3.	SMOKĪN RETURI <u>NOTE</u> :	er Notice System ("NO NG/FASTEN SEAT BELT/ N TO CABIN") Signs If the installation of a Passenger Address System is not a requirement, proviso (a) may be deleted.	-	-	Passeng occupie Belt/Re seat or I NOT O OR (0) No may be	As required by Air Navigation I ger seat, cabin attendant seat or le ed from which a "No Smoking/Fa turn to Cabin" sign is not readily lavatory must be blocked and pla CCUPY". Smoking/Fasten Seat Belt/Ret inoperative and the affected pas nt seat(s) or lavatories may be or	avatory may be asten Seat y legible or that acarded - "DO urn to Cabin signs ssenger seat(s) cabin
					(Cont))	

AIRCRAFT: DASSAULT AVIATION				REVISION NO: REVISION 1 PAGE:				
	FAN JET FALCON (FALC	ON 20)		DATE:	1 September 1994	33-2		
(1) Sys	tem & Sequence Numbers	(2) Numb	per Installe	ed	· · ·	L.		
	Item		(3) Nur	mber require	d for despatch			
				(4) Reman	ks or Exceptions			
33	LIGHTS (Cont)							
3.	Passenger Notice System ("NO SMOKING/FASTEN SEAT BELT/ RETURN TO CABIN") Signs (Cont)							
				(a)	The PA System operates norm clearly heard throughout the ca and	nally and can be bin during flight,		
				(b)	An acceptable procedure is passengers when seat belts n smoking is prohibited and (if passengers should return to a compartments.	nust be fastened, applicable) when		
					OR			
				(c)	Passengers are not carried.			
4.	Landing Lights (Wing Root or Retractable)	2	1		y be inoperative for night operation light is operating normally.	ons provided		
		2	0	Both ma	ay be inoperative for daylight ope	rations.		
5.	Landing Light Retraction Systems	2	1	(0) One night op	e may be inoperative in the retro perations provided the taxi light of	acted position for perates normally.		
		2	0	One or provide	both may be inoperative in the d Flight Manual Speed Restriction	extended position are observed.		
6.	Wing Root Recognition Lights	2	0	One or l	both may be inoperative.			
7.	Taxi Light	1	0	May be	inoperative.			
8.	Navigation Lights (Position Lights)	3	0	Any or a	all may be inoperative for dayligh	t operations only.		
		I		1				

	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	33-3
(1) Sys	stem & Sequence Numbers	(2) Numb	er Installe	ed	1
	Item	l r	(3) Nur	mber required for despatch	
			()	(4) Remarks or Exceptions	
<u>33</u>	LIGHTS (Cont)				
9.	Anti-Collision Lights (Red Beacons)				
	(1) Daylight Operations	-	0	As required by Air Navigation Legislation. be inoperative provided the light(s) is r earliest practicable opportunity.	Any or all may epaired at the
	(2) Night Operations	-	1	As required by Air Navigation Legislation. operative, and a high intensity strobe light s installed and operative.	One must be system must be
				<u>NOTE</u> : Operations with unserviceable anti are limited to flights within the UK	collision lights FIR only.
10.	Wing Ice Inspection Lights	2	0	One or both may be inoperative for daylight	operations.
		2	1	One may be inoperative for night operations	
		2	0	(0) Both may be inoperative for night opera an alternate means is available and utilised illuminate ice accretion on another outside from the flight deck.	l to adequately
11.	Wing/Tail Anti-Collision Light System (White Strobes) (If Installed)	1	0	May be inoperative.	
12.	Interior/Exterior Emergency Lighting Systems	-	-	As required by Air Navigation Legislation.	
13.	Logo Lights (If Installed)	-	-	May be inoperative.	
14.	Baggage Compartment Lights	-	0	May be inoperative.	

AIR	CRAFT:	DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	UN 20)		DATE: 1 September 1994	33-4
(1) Sys	tem & Seque	nce Numbers	(2) Num	ber Installe	ed	I
		Item		(3) Nur	nber required for despatch	
]		(4) Remarks or Exceptions	
<u>33</u>	LIGHTS	<u>S (Cont)</u>				
15.	Floor Pro Marking S	ximity Escape Path System	1	1	As required by Air Navigation Legislatic lights may be inoperative for a p configuration.	on, specific particular lighting
					If the equipment becomes unserviceabl continue to fly in accordance with arran by the Authority.	e the aircraft may agements approved
16.	Rear Equi (If Installe	ipment Bay Lighting ed)	-	0	May be inoperative.	

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	34-1
(1) Sys	tem & Sequence Numbers	(2) Numb	per Install	-	
	Item		(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>34</u>	NAVIGATION				
1.	Rate of Climb Indicator	2	1	One may be inoperative.	
2.	Angle of Attack Systems (If Installed)	-	0	(0) May be inoperative provided stall warning operate normally.	g systems
3.	Turn and Bank Indicators				
	(1) Rate of Turn Indicators	2	1	One may be inoperative.	
	(1) Rate of Full Indicators	2	0	Both may be inoperative provided Stan	dhu Uorizon
		2	0	Indicator operates normally.	doy monzon
4.	Stabilised Heading Indication	2	1	One may be inoperative provided:-	
	System			(a) At least one compass heading is ava pilots panel, and	ilable on each
				(b) Repairs or replacements are carried three calendar days.	ed out within
5.	Non-Stabilised Magnetic Compass (Standby)	1	0	May be inoperative provided:	
	(Sundoy)			(a) At least two independent stabilise gyro systems are installed and opera	
				(b) Repairs or replacements are carried three calendar days.	ed out within
6.	Flight Director Systems	2	0	One or both may be inoperative provided we or operating procedures do not depend on its	
7.	Distance Measuring Equipment (DME)	-	-	As required by Air Navigation Legislation.	
		1	l	I	

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	34-2
(1) Sys	tem & Sequence Numbers	(2) Number	Installe	*	1
	Item		3) Nun	nber required for despatch	
			,	(4) Remarks or Exceptions	
<u>34</u>	NAVIGATION (Cont)				
8.	Weather Radar/Lightning Detection System	1	0	 (0) As required by Air Navigation Legislation Required when flying for the purposes of purexcept that a flight may commence if trunserviceable such that: (a) The weather radar display is provide pilot, so long as the aircraft is flyin place at which it first become practicable for the system to be repaired the commander of the aircraft cumulonimbus clouds or other hazardous weather conditions, we detected by the system when in work unlikely to be encountered on the i or any planned diversion there commander has satisfied himself the weather conditions will be encountered on the interval of the second planned diversion there commander has satisfied himself the weather conditions will be encountered on the interval of the second planned diversion there commander has satisfied himself the weather conditions will be encountered on the interval of the second planned diversion there commander has satisfied himself the weather conditions will be encountered on the interval of the second planned diversion there is not planned diversion there commander has satisfied himself the weather conditions will be encountered on the interval of the second daylight and can be seen avoided, and is in either case operated throughour accordance with any relevant instruct the Operations Manual. 	blic transport, he system is ed to only one g only to the s reasonably red; or es available to indicate that potentially hich can be sing order, are ntended route from or the hat any such countered in d the aircraft t the flight in
9.	VHF Navigation Systems	-	-	As required by Air Navigation Legislation.	
10.	Ground Proximity Warning System	1	0	As required by Air Navigation Legislatic inoperative. The aircraft may continue the f of flights but shall not depart an airport reasonably practicable for repairs or replace made. <u>NOTE</u> : Particular circumstances may requi additional alternate procedures. The alternate procedure would operator to consider the routes ove flying and ensure that the pilot ad- path which would give the prot would otherwise be afforded.	light or series where it is ements to be re the use of require the r which he is opted a flight

AIR	CRAFT: DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	34-3
(1) Sys	stem & Sequence Numbers	(2) Numb	per Install	1	<u> </u>
	Item	[(3) Nu	mber required for despatch	
		1	(3) Nul		
				(4) Remarks or Exceptions	
24					
<u>34</u>	NAVIGATION (Cont)				
11.	Radio Altimeter System	-	-	(0) May be inoperative provided:	
				(a) Despatch deviation for GPWS is obs	erved, and
				(b) Weather minima or operating proce dependent upon its use.	dures are not
				NOTE: Radio altitude data may be required	by Autopilot,
				Flight Director System and GPWS.	
12	Instrument Componetor		1	Must be energine	
12.	Instrument Comparator		1	Must be operative.	
13.	Marker Beacon	-	-	As required by Air Navigation Legislation.	
14.	Radio Compass (ADF) System	-	-	As required by Air Navigation Legislation.	
15.	Glide Slope Receiver	-	-	As required by Air Navigation Legislation.	
16.	Mach/Airspeed Warning	1	0	(0) May be inoperative provided:	
				(a) Airspeed remains at or below Vmo	- 300 KIAS,
				Mmo 0.82 M,	
				(b) Both mach/airpseed indicators oper and	ate normally,
				(c) Repairs or replacements are carrie	d out within
				three calendar days.	
17.	ATC Transponder/Automatic	-	-	As required by Air Navigation Legislation.	
	Altitude Reporting				
18.	Altitude Alerting System	-	0	As required by Air Navigation Legislatio inoperative. The aircraft may continue the fl	
				of flights but shall not depart an airport	where it is
				reasonably practicable for repairs or replace made.	ements to be
		1		I	

AIF	CRAFT: DASSAULT AVIATION FAN JET FALCON (FALC	ON 20)		REVISION NO: REVISION 1	PAGE:
		-		DATE: 1 September 1994	34-4
1) Sys	stem & Sequence Numbers Item	(2) Num	ber Installe	ed	
	item				
				(4) Remarks or Exceptions	
5 A					
34	NAVIGATION (Cont)				
9.	Standby Attitude Indicator	1		Must be operative.	
γ.	Standoy Manude Indicator	1		Must be operative.	
20.	Radio Magnetic Indicator	2		One may be inoperative provided the HSI of	on the
	(RMI)			associated pilots instrument panel operates	normally.
21.	Inertial Navigation System (INS)	-	0	As required by Air Navigation Legislation.	
22.	Flight Management System (FMS)			May be inoperative provided required	novigation
22.	r light Management System (1103)	-	-	communication systems are not affected.	navigation a
23.	Vertical Navigation System	-	0	May be inoperative.	
24.	VLF/OMEGA System	-	-	As required by Air Navigation Legislation.	
25.	Microwave Landing Systems (MLS)			As required by Air Navigation Legislation.	
23.	where wave Landing Systems (wLS)	-	-	As required by All Navigation Legislation.	
26.	GPS/LORAN System	_	_	As required by Air Navigation Legislation.	
27.	RNAV System	-	-	As required by Air Navigation Legislation.	
28.	EFIS Display Source Select System			NOT USED.	
29.	EFIS Symbol Generator Units (SGU, DPU, and/or MPU)			NOT USED.	
	(555, 515, and/or 1110)				
30.	Multifunction Display (MFD)			NOT USED.	
- *	······································				

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)				REVISION NO: REVISION 1 DATE: 1 September 1994	PAGE: 34-5
(1) Sys	stem & Sequence Numbers	(2) Numb	er Instal	1	
	Item	Г			
		-	(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
34	NAVIGATION (Cont)				
<u> </u>					
31.	EADI Annunciators Displays			NOT USED.	
32.	EHSI Annunciators Displays			NOT USED.	
33.	VOR Angular/Linear Deviation Selector	-	0	May be inoperative provided at least one VOF is operating normally in the angular mode.	t system
34.	Navigation Data Bank	-	0	May be inoperative.	
35.	Storm Scope	-	0	May be inoperative.	
36.	NAV/COM Preselect Tuning Functions	-	0	May be inoperative provided direct tuning mo installed and operates normally for each affect	de is ted unit.
37.	Voice Advisory/Flight Profile Advisory System	1	0	May be inoperative.	
38.	NAV/COM/ADF/TDR Memory Channels	-	0	May be inoperative provided manual tuning on normally.	perates
39.	NAV/COMM/ADF/TDR Digital Frequency Selector/LCD/LED Display Units	-	1	 One pilot side only may be inoperative provid (a) Manual remote tune or dual FM capability operates normally, and (b) All digital Frequency Selector/LCD/ units on opposite pilot side operate normality 	S/CDU tune
40.	Compass Transfer	2	0	One or both may be inoperative provid information remains in the (onside) selection.	led compass

AIRCRAFT: DASSAULT AVIATION				REVISION NO: REVISION 1 PAGE:				
		FAN JET FALCON (FALC	CON 20)		DATE: 1 September 1994	34-6		
(1) Sys	tem & Sec	quence Numbers	(2) Numb	per Install	1			
		Item		(3) Nu	mber required for despatch			
				(0) Nu				
					(4) Remarks or Exceptions			
34	ΝΔνι	GATION (Cont)						
<u>0</u>								
		A A						
41.	ADI Ti	ransfer System	2	0	One or both may be inoperative provided:			
					(a) Pilot and copilot attitude disp connected to independent sources, an	lays remain Id		
					(b) The associated sources operate norma	ally.		
42.	Radar .	Auto Inhibit	1	0	May be inoperative provided the primary ra operates normally.	dar indicator		
43.	Radarn	av/Datanav System	1	0	May be inoperative.			
		·						
44.	Airbor	ne Flight Information	1	0	May be inoperative.			
		8			· · · · · · · · · · · · · · · · · · ·			
45.	Traffic	Alert Collision Avoidance						
		(TCAS) (If Installed)						
	(1)	TCAS System	1	0	May be inoperative provided the system is de secured.	activated and		
					If the aircraft is intended to be flown in airsp TCAS operation is required, it may fly for n	bace in which		
					10 calendar days with the equipment unserviceable, but shall not depart from a	completely		
					where it is reasonably practical for the equ	ipment to be		
					repaired or replaced.			
	(2)	Combined TA and RA Dual Displays	2	1	(0) May be inoperative on the non-flying pilot provided:	side		
					(a) TA and RA elements and audio a operative on flying pilot side, and	functions are		
					(b) TA and RA display indications are	visible to the		
		5 1 1 1 1 1			non-flying pilot.			
	(3)	Resolution Advisory (RA) Display System(s)	2	1	(0) One may be inoperative on non flying pilo	t side.		
					(Cont)			

AIR	CRAFT				REVISIO	N NO:	REVISION 1	PAGE:
		FAN JET FALCON (FALC	ON 20)		DATE:	1 Septer	mber 1994	34-7
(1) Sys	tem & Sec	quence Numbers	(2) Numb	er Install				1 /
		Item	Г	(3) Nu	nber require	d for desp	atch	
						-		
					(4) Remarl	KS OF EXCE	eptions	
34	NAV	GATION (Cont)						
45.	Traffic System (Cont	Alert Collision Avoidance n (TCAS) (If Installed) .)						
	(3)	Resolution Advisory (RA) Display System(s) (Cont)						
			-	0	(0) May	be opera	ative provided:	
					(a)	All traf	fic Alert (TA) display elemend audio functions are operations	nts and voice ive, and
					(b)	TA only	y mode is selected by the crew	v.
	(4)	TA Display System(s)	-	0	(0) May and aud	/ be inop io functio	perative provided all installe	d RA display
					I			

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATIO		REVISION NO: REVISION 1 PAGE:
FAN JET FALCON (FA	LCON 20)	DATE: 1 September 1994 35-1
(1) System & Sequence Numbers	(2) Number	Installed
Item		3) Number required for despatch
		(4) Remarks or Exceptions
<u>35 OXYGEN</u>		
1. Passenger Oxygen System	1	0 (M) or (0) As required by Air Navigation Legislation. The automatic presentation system may be inoperative provided:
		(a) The manual deployment system operates normally, and
		(b) The flight is limited to FL 300 or below.
	-	- (M) or (0) One or more passenger service units (PSUs may be inoperative without flight altitude restriction provided:
		(a) Affected seats are blocked and placarded to prevent occupancy, and
		(b) Units operate normally for all usable passenge seats, toilet compartments and flight attendan locations,
		OR
	1	0 (0) May be inoperative provided:
		(a) Flight is not conducted where the minimum en route altitude is above 12,000 feet MSL,
		(b) All other components of the pressurisation system operate normally,
		(c) Maximum flight altitude does not exceed FL 250,
		(d) Portable oxygen units containing sufficien oxygen for 30 minutes endurance are provided fo 10% of the passengers,
		(e) Passengers are appropriately briefed, and
		(f) Repairs or replacements are carried out within three calendar days.
		NOTE:
		The ANO oxygen requirements are given in Schedule 4 Scales L1 and L2. The effectively depends upon date o first issue of a certificate of airworthiness. Therefore, a given type of aircraft may have examples subject to either of the two scales of requirements.
		(Cont)

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALC				REVISION NO: REVISION 1	PAGE:	
	TAILULT TALEON (TALE	01(20)		DATE: 1 September 1994	35-2	
1) Sy	stem & Sequence Numbers	(2) Num	ber Install	ed		
	Item		(3) Number required for despatch			
		(4) Remarks or Exceptions				
5	OXYGEN (Cont)					
<u>J</u>						
	Passenger Oxygen System (Cont)					
				NOTE: (Cont)		
				The amount of oxygen required varie	es considera	
				between L1 and L2, particularly for opera 250/300. Provided the operator supplie amount of oxygen, despatch is considered a	tions above es the requi	
				Since there are a large number of per	mutations, it	
				proposed to refer to Air Navigation Legi the operator to adapt the MEL as nece constraints applicable. The main constraint	essary with	
				(a) The date of first issue of a Airworthiness for individual aircra		
				(b) The aircraft altitude and cabin al flown, and	titude on rou	
				(c) The numbers of passengers and cre	ew carried.	
•	Therapeutic Oxygen	-	-	As required by Air Navigation Legislation of those required may be inoperative.	. Any in exc	
				Note: The portable oxygen supplies require and L2 are totally separate from the require R2.	ed by Scales ements of Sc	
	Portable Oxygen Sets	-	-	As required by Air Navigation Legislation of those required may be inoperative.	. Any in exc	
	Crew Oxygen System and Equipment	-	-	As required by Air Navigation Legislation.		

AIF	RCRAFT: DASSAULT AVIAT FAN JET FALCON (REVISION NO: REVISION 1 PAGE:
(1) System & Sequence Numbers Item		(2) Number	DATE: 1 September 1994 36-1 er Installed (3) Number required for despatch
			(4) Remarks or Exceptions
<u>36</u>	PNEUMATIC		
1.	Engine Bleed Valves	2	 (M) One may be inoperative provided: (a) Associated valve is secured CLOSED, and (b) Flight is conducted at or below FL 250.
		2	 (M) Both may be inoperative provided: (a) Valves are secured CLOSED, and (b) Flight is conducted in an unpressurise configuration with ram air scoop OPEN.
2.	NOT USED.		

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)				REVISION NO: REVISION 1	PAGE:			
	FAN JET FALCON (FAL	UN 20)		DATE: 1 September 1994	49-1			
(1) Sy	stem & Sequence Numbers	(2) Num	ber Install		I			
Item			(3) Nu	(3) Number required for despatch				
				(4) Remarks or Exceptions				
49	AIRBORNE AUXILIARY POWER							
	POWER							
1.	Auxiliary Power Unit (If Installed)	1	0	May be inoperative provided procedure	es do not require its			
				use.				

MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT:	DASSAULT AVIATION			REVISIO	ON NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	ON 20)		DATE:	1 September 1994	52-1
(1) Sys	tem & Seque	ence Numbers	(2) Num	ber Installe		1.	I
		Item		(3) Nur	mber require	ed for despatch	
						-	
					(4) Rema	rks or Exceptions	
52		c					
<u>52</u>	DOOR	<u>5</u>					
1.	Door Wa (DOOR)	rning Light System	1	0	May be	e inoperative provided:	
					(a)	All doors and hatches are a inspection to be closed and prior to each departure,	confirmed by visual locked immediately
					(b)	Cabin altitude aural warning and	g operates normally,
					(c)	Repairs or replacements are three calendar days.	e carried out within
2.	GPU Doo (If Install	or Light System led)	1	0	inspect	y be inoperative provided it is v ion that the door is CLOSED eparture.	erified by visual and locked before
			I	I	I		
MASTER MINIMUM EQUIPMENT LIST

AIR	CRAFT:	DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	73-1
(1) Sys	stem & Seque	ence Numbers	(2) Num	ber Installe		1
		Item		(3) Nur	nber required for despatch	
			1			
					(4) Remarks or Exceptions	
<u>73</u>	ENGIN	IE FUEL AND				
	CONT	IE FUEL AND ROL				
1.	Fuel Flo	wmeters	2	1	(0) One may be inoperative provided:	
					(a) All other engine instruments for the engine operate normally, and	ne associated
					(b) All fuel quantity indicators operate no	ormally.
					(c) Repairs or replacements are carrie three calendar days.	d out within
2.	Fuel Cou	inters	2	0	One or both may be inoperative.	
3.	Fan Syno	chronisation System	1	0	(M) May be inoperative provided it is deactiv approved procedure.	ated using an
4.	NOT US	ED.				
5.	NOT US	ED.				
			1	I	1	

MASTER MINIMUM EQUIPMENT LIST

AIF	RCRAFT:	DASSAULT AVIATION			REVISION NO: REVISION 1	PAGE:
		FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	74-1
(1) Sys	stem & Sequ	ence Numbers	(2) Num	ber Installe	-	
		Item		(3) Nur	nber required for despatch	
			1			
					(4) Remarks or Exceptions	
<u>74</u>	IGNITI	ION				
<u>14</u>						
1.	Ignitor I	ndicator Lights	2	1	(M) One may be inoperative provided al associated ignition system operate normally	l modes of the
			I	I	l	

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATION					REVISIO	N NO:	REVISION 1	PAGE:
FAN JET FALCON (FALCON 20)					DATE:	1 Septer	mber 1994	77-1
(1) System & Sequence Numbers (2) Number			(2) Numbe	er Installe				
		Item	Г	(3) Nur	mber required	t for desp	atch	
			1	(0) 110		-		
					(4) Remark	(s or Exce	ptions	
77	ENG	INE INDICATING						
1.	Fan RF	PM Indicators (N2)						
	(1)	Operations above 12,000 feet MSL	2	2	Both Mu	ust be op	erative.	
	(2)	Operations below 12,000 feet MSL	2	1	(M) (0)	One may	be inoperative provided:	
					(a)	All oth associat	er engine indicating instrum ed engine operate normally,	ents for the
					(b)	Fan free engine s	edom of movement is verified start,	l before each
					(c)	Fan syn	chronisation system is not use	ed,
					(d)	Flight N	Ianual Limitations are observ	ed, and
					(e)	Repairs three ca	or replacements are carried lendar days.	d out within
2. N1 Indicators		2	1	(M) (0)	One may	be inoperative provided:		
	(Oas O	enerator Rotor)			(a)		her engine indicating inst ed engine operate normally,	ruments for
					(b)	Compre before e	essor freedom of movement each engine start, and	t is verified
					(c)	Approp Manual are appl	riate alternate approved proce Limitations, and performanc ied.	dures, Flight e decrements
3.	EGT Ir	ndicators	2	1	(0) One	may be i	noperative provided:	
					(a)		er engine indicating instrum ed engine operate normally, a	
					(b)		riate alternate approved proce Limitations, and performanc ied.	
4.	N2 Ind	licators			NOT US	SED (See	e Item 77-1)	

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT AVIATION FAN JET FALCON (FALCON 20)				REVISION NO: REVISION 1	PAGE:
	FAN JET FALCON (FALC	ON 20)		DATE: 1 September 1994	78-1
(1) Syst	em & Sequence Numbers	(2) Numbe	er Installe	ed	
	Item] r	(3) Nur	nber required for despatch	
				(4) Remarks or Exceptions	
<u>77</u>	ENGINE INDICATING				
5.	NOT USED.				
6.	NOT USED.				
7.	Engine Instruments (N1, N2, EGT and ITT Indicators)				
	(1) Digital Display	-	0	May be inoperative provided associated anal operates normally.	ogue pointer
<u>78</u>	EXHAUST				
1.	Thrust Reversers (If Installed)	2	0	 (M) One or both may be inoperative provided: (a) No damage to the thrust reverser which would adversely affect oper aircraft, and (b) A procedure is established and used i with the applicable Flight Manual sidetermine that the associated thrust disabled and pinned in the stowed (for position. 	system exits ration of the n accordance upplement to reverser(s) is
2.	Thrust Reverser Indicating Lights (If Installed)	2	0	(M) One or both may be inoperative provided associated reverser(s) is disabled and pinned i (forward thrust) position.	the n the stowed

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: DASSAULT		REVISIC	NNO: REVISION 1	PAGE:
FAN JET FA	LCON (FALCON 20)	DATE:	1 September 1994	79-1
(1) System & Sequence Numbers	(2) Numbe			
Item		(3) Number require	d for despatch	
		(4) Remar	ks or Exceptions	
<u>79 OIL</u>				
1. Oil Pressure Warning L	ights 2		One may be inoperative pro	
		(a)	The malfunction is verifi- system,	ed to be in the warning
		(b)	Oil pressure and oil ten operating normally and during flight, and	perature indicators are are monitored closely
		(c)	A light that remains deactivated.	illuminated must be

MASTER MINIMUM EQUIPMENT LIST

AIRC	RAFT:	DASSAULT AVIATIO FAN JET FALCON (FA			REVISION NO: REVISION 1	PAGE:
					DATE: 1 September 1994	80-1
(1) Syste	m & Seque	ence Numbers	(2) Num	ber Install	ed	
		Item		(3) Nu	mber required for despatch	
					(4) Remarks or Exceptions	
80	STAR ⁻	TING				
1	A					· · · · · · · · · · · · · · · · · · ·
1.	Automat	ic Starter Cutouts	2	0	One or both may be inoperative pro disengaged manually at 41% N1 during st	art.
			I	I	I	

MASTER MINIMUM EQUIPMENT LIST