

21 September 2015 Reference: F0002462

Dear XXXX

I am writing in respect of your recent request of 5 September 2015, for the release of information held by the Civil Aviation Authority (CAA).

Your request:

Occurrence reports for S-92, EC225 or AW189 aircraft that have made a diversion, air turn back or otherwise declared an emergency while in flight between 1 Jan 2014 and 31 Aug 2015.

Following our telephone conversation on 11 September, you agreed to limit your request to S-92 and EC225 aircraft.

Our response:

Having considered your request in line with the provisions of the Freedom of Information Act 2000 (FOIA), we are able to provide the information below.

Incident reports are provided to the CAA under the terms of the Mandatory Occurrence Reporting (MOR) scheme, as described under Article 226 of the Air Navigation Order 2009 (ANO). Each report made is reviewed and, where appropriate, further investigation carried out and action taken.

We have carried out a search of the CAA database for any report involving either helicopter type S92 or EC225, between 1 January 2014 and up to all processed reports as at 14 September 2015, which involves either a diversion, air turn back or the declaration of an emergency and provided a summary of those reports in the attachment.

We have not included identifying information in these summary reports as this information is exempt from disclosure under Section 44(1)(a) of the FOIA.

Section 44(1)(a) provides that information is exempt information if its disclosure is prohibited by, or under, any enactment. Under Section 23 of the Civil Aviation Act 1982, information which relates to a particular person (which includes a company or organisation) and has been supplied to the CAA pursuant to an Air Navigation Order is prohibited from disclosure (a copy of this exemption can be found below).

If you are not satisfied with how we have dealt with your request in the first instance you should approach the CAA in writing at:-

Caroline Chalk Head of External Information Services Civil Aviation Authority Aviation House Gatwick Airport South Gatwick RH6 0YR

caroline.chalk@caa.co.uk

The CAA has a formal internal review process for dealing with appeals or complaints in connection with Freedom of Information requests. The key steps in this process are set in the attachment.

Should you remain dissatisfied with the outcome you have a right under Section 50 of the FOIA to appeal against the decision by contacting the Information Commissioner at:-

Information Commissioner's Office FOI/EIR Complaints Resolution Wycliffe House Water Lane Wilmslow SK9 5AF www.ico.gov.uk/complaints.aspx

If you wish to request further information from the CAA, please use the form on the CAA website at http://www.caa.co.uk/application.aspx?catid=286&pagetype=65&appid=24.

Yours sincerely

Mark Stevens External Response Manager

CAA INTERNAL REVIEW & COMPLAINTS PROCEDURE

- The original case to which the appeal or complaint relates is identified and the case file is made available;
- The appeal or complaint is allocated to an Appeal Manager, the appeal is acknowledged and the details of the Appeal Manager are provided to the applicant;
- The Appeal Manager reviews the case to understand the nature of the appeal or complaint, reviews the actions and decisions taken in connection with the original case and takes account of any new information that may have been received. This will typically require contact with those persons involved in the original case and consultation with the CAA Legal Department;
- The Appeal Manager concludes the review and, after consultation with those involved with the case, and with the CAA Legal Department, agrees on the course of action to be taken;
- The Appeal Manager prepares the necessary response and collates any information to be provided to the applicant;
- The response and any necessary information is sent to the applicant, together with information about further rights of appeal to the Information Commissioners Office, including full contact details.

Freedom of Information Act: Section 44

(1) Information is exempt information if its disclosure (otherwise than under this Act) by the public authority holding it-

- (a) is prohibited by or under any enactment,
- (b) is incompatible with any Community obligation, or
- (c) would constitute or be punishable as a contempt of court.

(2) The duty to confirm or deny does not arise if the confirmation or denial that would have to be given to comply with section 1(1)(a) would (apart from this Act) fall within any of paragraphs (a) to (c) of subsection (1).

File number	UTC date	Manufacturer/model [Make]	Manufacturer/model [Model]	Location name	Headline	Nari
201400214	01/01/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	Master Caution illuminated, with AC GEN 2 FAIL illuminated.	On r AP 2 once line.
201401608	11/02/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	PAN declared due chip warning on intermediate gearbox.	Airc hold The help Sup Duri only redu land indic
201402175	22/02/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	Collective trim runaway.	On o pitch atter bega colle Sup I wa 1900 grou depa ackr initia park term

rative text

reaching 3000 carrying out cruise checks the Master Caution illuminated, with AC GEN 2 FAIL illuminated. The 2 light also illuminated indicating off line followed by AVC dropping off line. EOPs carried out and gen reset e. Subsequently failed again, this time with the above plus Capts PFD screen (MFD#4) temporarily dropping off EOPs carried out and aircraft returned to base.

raft inbound, called PAN due chip warning on intermediate gearbox. I acknowledged the PAN, cancelled his and speed restrictions (due multiple fixed wing and helicopter inbound and routed him to the ADN for a priority pilot advised me that he would be at 80kts. He was then transferred to INT and landed safely at 1037. WM bed me inform all the relevant agencies.

plementary 11/02/14:

ing flight, we had caption IGB CHIP. Consulting the emergency operating procedures (EOP) checklist, we had one indication of IGB CHIP. Following the EOP, immediate actions were nil, subsequent actions were to uce speed to 80kias and land as soon as possible. We informed ATC with a PAN call and requested priority for ling. During the flight we monitored for any unusual noise, vibrations and pedal kicks to which there were no cations of anything further. We briefed the passengers and continued for an ILS runway 16.

departure for Track and Balance air test, we engaged the upper modes IAS and ALT-P. IAS engaged in the h axis and ALT-P in the collective. Aircraft continued to climb through the selected alt, decoupled ALT-P and mpted to descend to 1200feet. Reduced power and recoupled ALT-P in the descent, aircraft immediately an to climb again. Torque was increasing steadily at approx 2%per sec. Deselected ALT-P and decoupled the ective trim, consulted the ECL and declared a PAN, RTB. Landed RWY 16 no further incident.

as on duty as the ADC with Air and GMC positions band boxed. Prior to me taking over control at approximately 0z, the aircraft had started for an air test, requesting a wide visual circuit. This followed a series of engine und runs and an earlier sortie which had been abandoned during the pre-departure hover-check. Aircraft arted RW16; when approximately 3 nm SE of airport, at 1912z a PAN was declared. The PAN was nowledged, the pilot confirmed a control trim runaway issue which was not serious, 4 SOB. PAN aircraft was ructed to join downwind left hand for either RW16 or H23; the pilot elected for RW16. A Local Standby was ated at 1914z, and the ATC Watch Manager informed. PAN aircraft landed at 1919z, and was able to taxi to king area, contacting Fire on 121.6. Thereafter Fire 1 reported that the aircraft was safe: the local standby was hinated at 1922z.

201402419	28/02/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to 'Eng' caution and 'Chip 2' warning light.	At 1 redu spec airci Sup In ci tran gate
201404798	15/04/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to two spurious engine fire warnings during flight.	X2 s cons fault
201404300	10/04/2014	SIKORSKY	S92	Brent D platform	PAN call due to severe vibration. Aircraft returned.	I wa wish relev Sup Uni deci eme vibra inve retu and Sup The ther and norr four roto adju

12:25 aircraft approached the coast and declared a PAN call stating he had a chip warning light and was ducing one engine back to idle and therefore reducing his speed. This was acknowledged. A previously placed eed restriction was lifted and a priority routing offered direct to final rwy16 under a direct VFR clearance. the craft was transferred to the tower and landed at 12:46 without further incident.

cruise ' Eng' caution plus 'CHIP 2'. ECL actions carried out, PAN call. Pax briefed. Engine #2 left at idle for nsit. No further signs of engine distress. Uneventful single engine running landing carried out. Pax debriefed in

spurious No2 engine fire warnings reported by aircrew in flight. Aircraft declared PAN and landed safely. After isultation, with manufacturer rep, No2 outboard flame detector replaced with new item due to known historic It with detector P/N: 92552-04112-042.

as radar controller when aircraft reported severe vibration was returning to platform. I asked aircraft if they hed to declare and they declared a PAN. I instructed aircraft to squawk 7700 when able and informed all evant agencies. Aircraft report landed safely and shut down.

oplementary 10/4/14:

nusual vibration felt after take-off from platform. Climb continued to 2000ft to establish if the vibration would crease with increased airspeed. The vibration continued. A pan call was made declaring the nature of the ergency, number of POB and intention to return. Pan acknowledged and transponder code 7700 selected. The ration continued during the descent. Tail rotor authority confirmed before landing. Aircraft shutdown for further estigation. Engineers winched down later in the day. After investigation and air-test the next day, aircraft urned without passengers. Further investigation to aircraft continued.OIM and passengers briefed by both crew d engineers.

plementary 14/4/14:

e vibration was described as a rumbling cobblestone type vibration normally associated with a 4p vibration and re were no other abnormal indication or captions. The mornings flights were downloaded from HUMS laptop d no obvious increase in readings but it was noted that the vertical 1p vibration was slightly higher than we would mally aim for in certain regimes. The MRH and MRB were inspected for any apparent defects but none were nd but it was noted that two dampers were nearing the fill mark. Main rotor dampers were filled and bled and a pr smoothing data gathering flight carried out. Air tests were unable to reproduce the vibration. Recommended ustments were carried out and the 1p vibration levels smoothed to a more comfortable level.

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201401048	29/01/2014	SIKORSKY	S92	EGPB (LSI): Sumburgh	Intermittent 'ENG PWR LIMIT' caption.	Duri capt inter
201401321	03/02/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared and aircraft returned, after shutting down one engine, due to 'CAUT FUEL' caption during air test.	Duri the I carri and with engi new pres Sup Duri First pres
201402781	09/03/2014	SIKORSKY	S92	EGPB (LSI): Sumburgh	PAN declared due small cabin fire from the Captain's TCAS module. Aircraft made precautionary landing.	I rec a pre aircr my \ dow from Sup Whil from prec dissi retui out.

ing descent to winch at cliff, eng 2 TGT display showed red dashes with associated 'ENG 2 PWR LIMIT' tion. EOP actions carried out and decided to curtail sortie. Whilst recovering to base, caption illuminated rmittently. On two occasions, the 'FADEC NO DISPATCH' caption also illuminated intermittently.

ing the cruise, whilst conducting a HUMS air test, 'CAUT FUEL' illuminated with an associated 'FILT' caption on Nr1 engine side of the fuel panel. The emergency checklist was consulted and 5/2 filter clogging warning ried out which results in shutting down the engine. Aircraft turned back toward base, procedure 2/7 carried out I nr1 engine shut down. A PAN was declared and aircraft routed back for a running landing. Aircraft landed nout further incident. Test of pressure drop switch carried out, no fault apparent. Test of clogging indication nr1 ine fuel differential pressure switch found to function intermittently Pressure switch and fuel filter replaced with v component. Flight test carried out satisfactorily. This was the first flight after replacement of the differential ssure switch.

plementary 03/02/14:

ing test flight, pilot had to shut down nr1 engine due to fuel filter clogging indication on fuel management panel. t flight since the fuel pressure switch replaced. Fault confirmed to be intermittent indication caused by fuel ssure switch and further air test carried out with no further fault.

ceived a PAN call advising me that they had a small cabin fire from the Captain TCAS module and were making recautionary landing, they passed the position in Lat and Long. I acknowledged their call and observed the raft on radar, confirming their position. The next call saying that they had landed safely on the ground. I advised WM and ADC who called the local emergency services. Subsequently aircraft called airborne and rigraded/cancelled the pan as they had isolated the TCAS module and were en route to pick up a crewman n a vessel.

plementary 09/03/14:

ilst carrying out SAR winch training with vessel, smoke was detected in the cockpit and found to be emanating in the TCAS 2 display in front of the Captain. The aircraft was flown to a nearby island and landed as a caution whilst a PAN call was made to ATC. The TCAS circuit breaker was identified, pulled and smoke sipated. The unit was not hot to touch hence aircraft was repositioned to recover winch man from vessel and rn to base. TCAS/VSI found to be cause of smoke, removed and to be replaced. Function checks to be carried

201403028	13/03/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	PAN declared and aircraft returned due to possible engine fire.	In C cons extir and Supp Aircr orbit 2A, aske engi advis had this ADC airpo
201409319	12/07/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to transmission chip warning.	Aircu retur shep Sup Note we v EOF cont call - no foun clea chip dete
201403428	22/03/2014	SIKORSKY	S92	EGPB (LSI): Sumburgh	PAN declared due to nr1 engine chip warning. Aircraft returned.	Aircr engi arriv Supj At 10 an e was rema Supj Duri caut ENC engi ferrc and

Cruise at 3000 ft, a spurious ENG #2 FIRE warning was noticed and Aural warning was heard. Decision to sult ECL and to start a right hand turn was commenced. As the turn was initiated the fire warning signs nguished. No signs of smoke or erratic engine indications were noticed. Blue sky quick position was engaged ATC was in formed with a PAN call. RTB without further problem.

The pilot advised that they intended to return to departure airport and requested a descent to which was granted routing direct to GSE or the ATF as required. The routing to GSE for VFR was accepted. I ed the pilot if they wished to declare an emergency. Aircraft declared at PAN due to an indication of a possible ine fire. The PAN was acknowledged and aircraft was instructed was to squawk A7700. Which they did. Pilot ised that they believed the fire indication was spurious and they were operating normally on both engines, they not seen any smoke trail during the tight orbit. Another aircraft inbound for the ILS, approximately 8nm north of one, offered to escort the aircraft to the field but pilot declined the escort. The duty watch manager, INT and C were briefed on the situation and of the aircraft intentions. Aircraft was cleared to enter CAS routing direct to ort not above 2A, transferred to airport ADC and landed safely.

raft declared a Pan with a chip warning and requested immediate descent to 1A. He elected to continue rning and declined a□

pherd aircraft. He reduced speed to 80kts. At time of report a/c still has 40 minutes to land.□ plementary 12/7/14:□

ed aircraft was on a monitor for main gearbox chip every five flight hours. Fifteen minutes into the inbound leg were presented with a CAUT XMSN CHIP. EOP actioned, chip pulse activated but CHIP remained on. law. P speed reduced to 80 kts (Vy). On speed reduction thru 90 kts, the CHIP light extinguished. law. EOP, flight tinued to base with us cautiously increasing speed to 100 kts. As a precaution we descended to 1000 ft. Pan was made at 16:07 to radar. No further reoccurrence. All MGB magnetic chip detectors removed and inspected o debris or particles found, all chip detectors clean. MGB oil filter removed and inspected - no particles or debris and, filter clean. Oil cooler magnetic plug removed and inspected - no debris or particles found, magnetic plug an. Due to the nature of the reported defect (warning extinguished as power and speed reduced) all electrical o detectors tested for correct electrical properties. Flared housing chip detector found U/S. Flared housing chip ector replaced and aircraft released to service after appropriate ground runs etc.

raft was operating low level on a VFR training sortie when at time 1038 they called PAN PAN declaring an ine malfunction and return. They were instructed to squawk 7700, which they did and returned for a VFR val. One engine was shut down but no further assistance was required. They landed safely at time 1101.

038 radar called to inform me aircraft was north of the CTR returning to the field after declaring a PAN due to engine malfunction. A full emergency was initiated. I was then informed aircraft had shut down one engine and a flying on his number 2 engine only, which was passed onto the fire chief. Aircraft landed safely with the AFS aining in position until he shut down. The full emergency was stood down at 11.09.

ring a continuation training sortie as the aircraft came to a relative hover alongside a local ferry the ENG 1 CHIP ution illuminated intermittently. We immediately diverted the aircraft and during the transition to cruise flight the G 1 CHIP caption remained on. The EOP was consulted which resulted in the aircraft returning with the affected gine at idle. We declared a pan call and landed via a running landing. Chip detector inspected and cleaned no rous debris but small amount of carbon in chip detector basket. GE inspection and clean inspection carried out a ground run leak check carried out satis.

201403494	23/03/2014	SIKORSKY	S92	EGPB (LSI): Sumburgh	PAN declared and aircraft returned due to 'ENG 1 CHIP' caution.	At 1 sing posi Sup On I engi aircr repla
201407589	11/06/2014	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	PAN declared during taxi out due to passenger medical emergency.	On r very and pass Ove inop
201407490	10/06/2014	EUROCOPTER	EC225	Oil Platform	Aircraft returned due to torque split and mismatched power indications.	On a Tota desp climi setti CAA Roo

11:20, shortly after departure, the aircraft declared a PAN due a chip warning light and would be landing with gle engine on runway 27. Full emergency was initiated. Aircraft landed safely at 11:26 with the AFS remaining in sition until he shut down. The emergency was stood down at 11:30. □

levelling, shortly after take-off, 'ENG 1 CHIP' caution illuminated. Actions carried out iaw EOPs, retarding nr1 gine to idle and a single engine running landing was carried out with no further incident. Nr1 engine secured and craft ground taxied to dispersal for shutdown. Chip detector inspected, no debris found, suspect chip detector laced iaw AMM. Ground run and leak check carried out satisfactorily.

reaching holding point M5 a passenger came forward and reported that another passenger appeared to be y unwell. Surrounding passengers gave immediate support to unwell individual. Crew issued a Pan Pan Call d taxied back into the apron. Medical assistance and an ambulance was requested which met the unwell ssenger on spot/inside terminal. Flight later departed without further incident (minus unwell passenger). erheating of individual is/was initially suspected. Cabin Vents were on however the air conditioning was perative and listed as a B defect

approach, the crew observed CHQ TRQ and Diff Pwr indications. Delta N1 gauges appeared unaffected. tal Torque was around 65%. No1 torque: 23%, No2 torque: 42% approximately. The mismatch remained spite the application / reduction of collective and indications were intermittent. The crew went around and nbed to MEA. Indications stabilised during the climb and the aircraft returned to base at a reduced power ting for further investigation.

A Closure: 🗆

ot cause was the failure of the No 2 torque transmitter. Component replaced.

201405259	29/04/2014	EUROCOPTER	EC225	En route	PAN declared and aircraft diverted due to transmission	l wa
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201407798	14/06/2014	SIKORSKY	S92	En route	Fire warning, engine nr2	Wh
201101100	1 1/00/2011		002	2		actu
						repe
						CĂ
201409873	21/07/2014	SIKORSKY	S92	En route	Aircraft returned due to Display Control Panel (DCP)	The
					failure.	was
						cou
201408437	24/06/2014	SIKORSKY	S92	EGPM (SCS): Scatsta	PAN declared due to passenger medical emergency.	
					Aircraft returned and was met by emergency services.	
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as on duty as the Radar controller supervising a controller who is re validating on the sector. This aircraft was route at FL070. At 0937 the pilot declared a PAN, reporting 'a minor gearbox warning' and that they were erting, commencing descent and reducing speed. The PAN was acknowledged and the pilot was advised that are was no traffic to affect their descent and to route direct to diversion airport. Aircraft was instructed to squawk 700, which they did, confirming their SOB as 20. Their ETA was 10:00. The Watch Manager and adjacent ctors controllers advised all relevant agencies of the situation. We briefed ADC of the diversion. 0943 aircraft quested the PB weather, which was passed and confirmed they would make a VFR approach, a VFR clearance uting direct to the field was issued and read back. 0954 The aircraft was transferred to the airport ADC frequency 8.250mhz. 1004 Aircraft landed safely. All agencies informed.

pplementary 01/05/14:

the cruise, a transmission chip warning illuminated on the CWP accompanied by a CHIP caption on the VMS. EOPs were consulted and two attempts were made to burn the chip with no success. Power was reduced and a/c diverted. Engineer's Report: During the flight the amber transmission (XMSN) caution warning light came in the cockpit together with a 'CHIP' warning. The aircraft landed to enable engineering to carry out an pection. Warning System This cockpit 'CHIP' warning indicates the possible presence of metal particles on one the magnetic chip detectors fitted to the main, intermediate or tail transmission assemblies. In normal operation, magnetic chip detectors will indicate a cockpit warning when the metal contamination Is attracted to the gap on e detector. When the contamination bridges the gap an electrical circuit is made and the 'CHIP' warning light in e cockpit illuminates. With reference to the manufacturer's maintenance manual, the presence of particles on a p detector does not necessarily mean there is a problem with the transmission. Consideration is given to the: ape of the particle, the quantity and size, the nature of the occurrence and the nature of the material. On this casion, a hair-like thread was found on the detector of the intermediate transmission. If contamination is found en we have to strictly follow the aircraft manufacturer's recommendations as detailed in their maintenance nuals. A serviceability check in accordance with the manufacturer's recommendations was carried out. This luded ground runs and a hover. The detector was checked again with no further issues. Further Inspections: aircraft will be under close monitor and this means that the transmission chip detector will be checked after ery flight for a period of 25 flying hours. If no further issues then the aircraft will be returned to normal iintenance. 🗆

pplementary 01/05/14:

duty as ADC I heard a PAN being declared on Radar frequency (0939z). At that moment I had no details,

nilst on task, FIRE ENG 2 aural alert and fire warning caption came on. Initial actions were carried out and an cual fire was not confirmed. Aural warning and caption went out after a few seconds but came on again beatedly. Aircraft RTB and precautionary twin engine running landing.

A Closure:

e number 1 DCP failed during cruise flight. The crew called for weather at destination through Log and the cloud s reported as overcast at 500 feet, the MEL states 1 DCP for dispatch but VFR flight only. It was clear that VFR uld not be maintained after take-off from the rig and therefore the aircraft aborted the flight and returned to base.

201409973	23/07/2014	SIKORSKY	S92	En route	Aircraft returned due to minor technical problem.	We
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						retu
201410250	20/07/2014	SIKOPSKY	S02	Bridge of Dop	PAN declared due to pose dear retracted on approach	Airc
201410250	29/01/2014	SINONSKI	592	Bridge of Don	FAN declared due to hose gear retracted on approach.	tho
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201406148	16/05/2014	EUROCOPTER	EC225	EGPD (ABZ):	HUMS MOD 45 fail message. Aircraft returned.	En-
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e received a call from the coastguard informing us that the aircraft was returning to the field with a minor hnical problem, not declaring an emergency. After confirming this with radar a local standby was initiated for the urning flight. Aircraft landed safely.

craft was inbound and I had instructed them to report at the Bridge of Don. On reaching BoD the crew stated by had an issue with their nose gear, and asked to orbit at BoD to go through the checklist. Having completed is and not resolved the problem, the captain declared a PAN and advised they would need to hover on the field to try to further clear the problem. Since the a/c was in the CTR, I decided against instructing the crew to ect 7700 on their transponder and I cleared the a/c to hover above runway 23. Visual confirmation ascertained at the nose wheel was still retracted. As runway 34 was in use, I decided to stop all runway movements until I ew what actions would be required by the captain/ company. When it become clear that the subject aircraft was cablished in the hover on runway 23 and remaining to the east of E6, I was satisfied that I could recommence erations on the main runway. I then handed over the watch to the incoming controller. pplementary 12/08/14:

tween Gorse and the Bridge of Don on approach to destination, the before landing checks were carried out. The se wheel indication light failed to illuminate, therefore drills were carried out IAW the ECL. This did not resolve e situation, so a PAN was declared and an approach to R/way 23 made, where we remained in the hover East of . ATC were able to confirm visually that the nose wheel had not extended. After numerous calls between gineering and Fire 1(airport fire service), it was decided that an engineer would walk underneath the aircraft and empt to free the nose wheel. After 2 attempts, the engineer successfully freed it. With a confirmation of this from a engineer and a green indication in the cockpit, the aircraft was landed and immediately shut down. It should be ted that this sequence of events took 50 minutes from time of notifying ATC to a resolution being reached. It ickly became apparent that there is no contingency procedure for this event; even to the point that the engineers d no hand held radios to communicate with us on. During this time, a passenger came forward expressing owing concern over the situation and an ever increasing temperature as the air-conditioning was u/s. It should o be noted that we were carrying sufficient fuel to cater for the time taken for a resolution to be found. A Closure: A Closure:

restigations found that the root cause was that the nose wheel was not centred when retracted. Nose landing ar shock strut seals and main landing gear actuator solenoid valve seals replaced. Retraction/extension checks ried out satisfactorily and aircraft returned to service.

-route approximately 60nm out we got a HUMS Caption and a MOD 45 FAIL displayed on the HUMS panel. hergency checklist consulted and aircraft slowed down to 120Kts and torque reduced to less than 65% in cordance with drill 7/10 of the Emergency Checklist. This was maintained for 15 minutes as instructed and fail lication remained. Approx 100nm ATC were informed of the situation and an RTB was requested and osequently initiated. The aircraft initially descended to 2000ft routing direct and subsequently after a request m ATC descended to 1000ft to avoid conflicting traffic still maintaining 120kts and less than 65%tq. As we were t of range for radio contact with Ops we tried to make contact 4 times, however each time we were unsuccessful e to connection failure. We then monitored company frequency in the hope another company aircraft was talking Ops and could relay however, again this was not the case. We finally got hold of Ops about 30nm out. Aircraft ded without further incident.

A Closure:

nfirmed random failure. Appropriate actions carried out iaw AMM CH 05-53-00-237 and EDR raised with unufacturer. NDI inspection of shaft confirmed no fault and aircraft returned to service. No re-occurrence since.

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201406128	09/05/2014	EUROCOPTER	EC225	En route	Aircraft returned due to main transmission cowling caption on climb out.	On o CW Trar cont depa
201501286	02/02/2015	SIKORSKY	S92	En route	PAN declared and aircraft returned due to hydraulics issues.	Airc
201411446	18/08/2014	SIKORSKY	S92	EGPM (SCS): Scatsta	Aircraft returned due to engine torque fluctuations	Duri

n climb out, Main Transmission Cowling caption illuminated on Door/Cowl Panel with associated Door/Cowl on VP and CAUT. The caption was intermittent and continued flickering in the cruise phase at MCP. The ansmission Cowling was visually checked in the mirrors and observed to be secure but we could not sensibly ntinue the flight. Aircraft returned to base and passengers debriefed by crew on PA and Engineering in parture gate.

craft declared a PAN due to hydraulic issues. Handling not affected. Returned to departure airport and cancelled N at 0853 as warnings had cleared but continued back. Landed at 0900.

ring climb engine torque fluctuations noted. After levelling and cruise power set fluctuations increased up to 10 rcent between engines with corresponding Ng , TGT and NR indications. Power reduced and aircraft RTB.

201501981	17/02/2015	SIKORSKY	S92	EGPM (SCS): Scatsta	AC generator No.1 failure.	On leve
201417123	10/12/2014	SIKORSKY	S92	En route	PAN declared and aircraft returned due to passenger medical emergency.	Para
201418191	30/12/2014	SIKORSKY	S92	Unknown	Flight crew illness/Incapacitation.	Duri feeli take adrr with

n climb out after departure when the AC Gen No.1 Failure caution came on. Climbed out on rwy heading and velled off at altitude 2000ft. Carried out SOP and EOPs and aircraft returned to base.

ramedics met the aircraft on arrival.

uring the climb the co-pilot (PF) started to complain that he was not feeling well. He complained of numb hands, eling hot and dizzy. I took control and levelled the aircraft. The symptoms persisted so I RTB. The co-pilot was ken to A&E by another Captain whilst I completed the post flight□ dministration. Further details from the co-pilot to follow in due course. Passengers were debriefed post landing

hout further comment.

201410351	30/07/2014	EUROCOPTER	EC225	ADN	XMSN CHIP' caution illuminated during cruise. Aircraft diverted.	In ti uns On plug win
201409447	15/07/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared and aircraft returned due to torque split and nr2 engine 'GOV' light illuminated.	Airc out, dec the Sup Wh 2 er PAI 2 to sigr we ther MC Blee Pas
201406421	21/05/2014	EUROCOPTER	EC225	En route	PAN declared due chip warning.	PAI squ Sup CHI 1 se belo war sys war mag the plug The repr oil, carr con mar plug

the cruise at 4000ft Caut, XMSN and Chip illuminated. The crew followed the EOP and attempted two successful burns on the chip detector. The speed was reduced to Vy and the aircraft diverted (nearest location). arrival, the pax were disembarked and debriefed and engineering consulted. On inspection of the MGB sump g a substantial amount of metal debris was found. The crew and pax returned to intended destination on a fixed ng flight.

craft outbound stated that he had a minor technical problem and wished to return to base from about 105 miles t, asked if he wished to □

clare but declined saying that it was unnecessary at the time. About 84 miles from base declared a PAN due to problem, was offered routing to an installation but was happy to continue. Landed safely. pplementary 15/07/14:

The in cruise flight, the VMS "Check TQ" illuminated and torque split of 5% was noted. Shortly after this, number engine GOV light illuminated and torque split noted as circa 15%. Aircraft turned back to departure airport and a N was declared. Subsequent analysis during the return flight caused us to believe it was an erroneous number orque indication over-reading. One consequence of this failure was that the FLI MCP limit red line reduced nificantly and did not reflect the engine power settings showing on the VMS (photos available if required). Had followed the guidance on the FLI, our total power available would have been reduced significantly. We erefore relied on the VMS indications to set a reduced cruise power but this was still above the FLI indicated CP setting so the autopilot kept trying to reduce collective. Collective trim was therefore manually deselected. eeds were offset manually prior to landing and an uneventful run on landing made with no further consequences. ssengers debriefed after landing.

N declared due chip light number 1 engine reduced to idling requiring no further assistance. Aircraft asked to Jawk 7700 when able. Aircraft landed. □

IIP1 light illimunated in cruise. EOPS actioned but light returned for the third time after clearing twice, so engine et to idle. PAN call made. Continued flight to destination. IHUMS (No Mod 45) illuminated subsequently as ow 60% torque in cruise. Running landing was carried out. Ts and Ps normal throughout. No1 engine chip rning during the flight back to base. The engine has 5 magnetic plugs and a chip detector situated within the oil tem. The engine chip detector is designed to attract metal particles in the oil system and will illuminate a rning light in the cockpit when the quantity or size of the particles are sufficient to bridge a gap between two ignetised sections on the detector, thus creating electrical continuity. After landing, the aircraft was taken into hangar to enable engineering to carry out the required inspections. This included checking all the magnetic gs and the electrical chip detector. There were signs of hair-like contamination on the electric chip detetctor. contamination was sent to a laboratory for analysis and the report forwarded to the engine manufacturer field resentative. Serviceability checks were carried out iaw the AMM recommendations. Ths includes draining the inspecting the cleaning mag plugs, chip detectors, stariners and replacing the oil filter. A ground run was ried out and further checks on the mag plugs and chip detector. No contamination found. The amount/type of tamination has been assessed to be within limits as specified in the engine manufacturer's maintenance nual. This will allow the aircraft to return to service under close monitoring, whcih entails a check of the mag gs and chip detector after 5 flying hours for a period of 25 flying hours. If no further contamination, then the craft will be returned to normal maintenance.

plementary 22/05/14:

201417673	18/12/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to landing gear indication light failure.	I was train they RW ¹ they Supp Left- decla
201413592	26/09/2014	SIKORSKY	S92	Claymore A platform	PAN declared due to engine surge/stall warnings.	Aircr to cli climi squa back infor esco Was Supj On i spike retar almo decla runn tank CAA Duri caus also aircr
201501456	06/02/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to engine chip warning.	Aircr instr Supp Chip brou land inclu elect engi amo a sin out f this o repla the E and

as working as the tower controller, the weather was good VMC and the traffic was very busy. Aircraft was a ning flight following a company aircraft to final for RWY, with another helicopter on final for RWY. At this point / called a PAN with unsafe undercarriage indications. They stated that they would like to make an approach for Y. I got the other traffic out of the way and instructed them to land. On landing they remained in the hover until / got their gear checked by an engineer before landing safely.

plementary 18/12/14:

-hand undercarriage light did not illuminate with travel light still illuminated. Check list followed, emergency lared, landing carried out with engineering assistance (indication issue).

raft was doing deck training. Appeared to be climbing out and tracking as normal when he called at 1A looking limb to 2A. He said he had an engine problem and had idled back an engine. Was identified given a ROT and bed to 2A routing direct. When asked if he was declaring an emergency he declared a PAN, he was told to awk 7700 and confirm SOB. WM was informed. When queried he confirmed it was the #1 engine that was idled k due to surge/stall warnings but when asked did not indicate any other problems with the aircraft. He was remed of traffic which was 14nm SE of him that would be available to escort him. He said he didn't need the ort. Aircraft 2 was transferred and given the aircraft headings to converge with and escort the PAN aircraft. Is at approx 65nm and routing to the field at 2A when I was relieved.

plementary 26/9/14:□ initial climb out we heard compressor bangs surging and stalling with the associated Engine pod was showing tes in Ng and TGT and sporadically turned yellow. We levelled the aircraft and consulted the EOPs we then inded the throttle of the affected engine and the stalling ceased we then tried to return the throttle to fly and ost immediately the surge/stall returned. We then idled the engine and decided as a crew to start the APU. We lared a pan call who had the flight watch and subsequently to radar. We returned with the engine at idle to a ning landing. We elected not to cross feed as we were going to land with sufficient fuel on the good engines

and the number one wasn't going to be beyond 700lb imbalanced.

ng borescope inspection, the engineer noticed damage to the axial compressor blades suspected to be sed by FOD. Further investigation showed considerable damage to the centrifugal compressor and the turbines showed signs of damage. The engine was removed and sent for inspection and repair. An inspection of the raft was carried out to try and identify the source of the FOD. No hardware (screws / nuts) were found to be

raft declared a PAN stating he had a technical problem with one of his engines. I acknowledged the PAN and ructed aircraft to set 7700 and informed the Watch Manager.

plementary 6/2/15:

o number 1 engine on return leg we confirmed indications and followed EOP's (engine reduced to idle). Engine ught to flight on finals and a precautionary run on landing was made, engine secured after landing. After ding, the aircraft was taken into the hangar to enable engineering to carry out the required inspections. This uded checking all the magnetic plugs and the electrical chip detector. A hair-like particle was found on the triccal chip detector. This particle will be sent to a laboratory for analysis and the result will be forwarded to the ine manufacturer field representative. In accordance with details specified in the maintenance manual, the but and type of contamination is regarded to be within acceptable limits. However, there was an occurrence of milar chip approximately 50 flying hours previously. A close monitor for a period of 25 flying hours was carried following this with no recurrence of the warning and the aircraft reverted to normal maintenance. As a result of chip warning and the previous one, the engine manufacturer representative has confirmed a module 4 accement is required. To do this module replacement, the engine will be removed from the aircraft and taken to Engine Workshops. A serviceable replacement engine will be fitted to the aircraft. Post fitment, ground runs air tests will be carried out. If everything is satisfactory, the aircraft will return to service.

201501190	0 30/01/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared and aircraft returned due to engine chip warning.	Hav saic inst pho offe earl Sup Dur puls whic furtl OEI grou run sev
201415197	7 28/10/2014	EUROCOPTER	EC225		PAN declared due tail rotor gearbox overheating during cruise.	Free traft they soo thrc adv Sup Dur dec con the shu eith on i TGI Imp
201416780	01/12/2014	SIKURSKY	592	EGPB (LSI): Sumburgh	PAN declared and aircraft returned due to gearbox oil pressure warning.	Airc pres retu and Sup Airc dec Sup Dur illur

ving already requested a return "as a precaution", aircraft declared a PAN with an engine chip warning; the crew d that the engine had been reduced to idle and they would be returning on a single engine. Aircraft was tructed to squawk 7700 which was complied with. The Watch manager was informed and made all relevant one calls. A company aircraft that was in the vicinity, offered to act as escort if required; after a brief pause this er was turned down. On establishing that he could take a frequency change was transferred to the next sector rly so as not to risk losing 2-way contact as the RT link was OOS.

pplementary 30/1/15: Iring cruise CAUT, ENG and CHIP 1 lights illuminated. ECL consulted and Chip Detector control switch set to lse. CHIP 1 caption cleared. After a few seconds, the same indications appeared. Chip pulsed a second time ich cleared again. As a precaution the power was reduced to below safe single engine while the crew discussed ther actions. It was decided to return to base. Again the CHIP 1 light illuminated so crew set engine 1 to idle and ELLO on engine 2, descended to 2000 feet and declared a PAN. Pax were briefed in the aircraft and on the

I LO on engine 2, descended to 2000 feet and declared a PAN. Pax were briefed in the aircraft and on the bund subsequent to the flight. The aircraft was returned and carried out a VFR approach and running landing on hway 34 without further incident. Engineering informed. It should be noted that the CHIP 1 light illuminated veral times during the return.

equency very quiet, aircraft was approaching the 80 miles mark. I had already advised him there was no known ffic offshore and they could report to me once they transferred the flight watch offshore. Before they could do so, ey called me with a PAN PAN due to a tail rotor gearbox overheat indication and stated they intended to land as on as possible. As they descended low level, I lost radio contact with them on frequency, and managed to relay ough another aircraft to make sure they were speaking to the offshore frequency (which they were). We were vised that they landed safely,

pplementary 28/10/14:

ring the cruise 'WARN' illuminated with an associated 'TGBT' caption on the VMS. Aircraft immediately celerated to Vy, four-axis coupled. EOP consulted, drill 7/8 actioned. Line test returned 'Normal', high TGB temp infirmed. PAN call made and aircraft landed and immediately shut down. Engineering assistance sought. During a flight out the crew observed a 'TGB.T' indication on the VMS. Following their flight procedures they landed, ut down and contacted engineering back. On the information given by the crew the fault was diagnosed to be here the TGB temp probe or associated wiring. Myself and another engineer where sent out recover the aircraft, investigation it was found that the wiring to the TGB temp probe was at fault. The wire was repaired and the B indication system was tested 'satis'. The aircraft was released and returned back. As a precaution provement of IGB and TGB temperature probes and wiring has now been carried out on all aircraft type.

craft was performing low level work approximately 35 miles west of airport and called a PAN due to a gearbox essure warning. The pilot requested a return and was advised that there was no know traffic to affect a direct urn, and to select SSR code 7700. The pilot later reported that the indication was improving, but still wished to urn as a precautionary measure. The aircraft was cleaned to join the Control Zone VFR, with no level restriction d was transferred to Tower approximately 12 miles west of the airfield. The aircraft landed safely at airport. pplementary 01/12/14:

craft was returning to airfield after declaring a PAN PAN due to a gearbox oil pressure warning. Full emergency clared.□

pplementary 01/12/14:

ring prolonged hover 7.5deg nose up, oil pressure observed to decrease from 58 PSI to 31. Main oil pressure minated. PAN Call made, aircraft recovered. During transit oil pressure recovered. No further indications seen.

2014	15360	30/10/2014	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	Birdstrike to flight deck window.	Crui bird) Bloc
2014	16044	14/11/2014	SIKORSKY	S92	En route	PAN declared due to fluctuating gear box oil pressure during cruise.	Airc for t Airc Sup Duri MGI PRE com initia recc leve defe CAA
2014	16290	20/11/2014	EUROCOPTER	EC225	EGPB (LSI): Sumburgh	Aircraft diverted due to cracked windscreen. Local standby initiated.	At 1 eme

uising 1500ft offshore, when there was a thump and the centre screen was hit by a bird(probably a duck sized d). Screen remained complete but crazed.

ood smeared on screen. no other evidence of damage or remains. Returned back.

craft was approximately 7 miles nw of airport routing for a 5 mile point for rwy to carry out a low &slow approach training when he declared a PAN for a fluctuation in the gear box oil pressure. Full emergency was declared. craft landed safely on rwy. Incident was stood down by fire chief.

pplementary 14/11/14:□ ring recovery to base, MRGB OIL PRES amber caption illuminated momentarily. This drew crews attention to

GB oil pressure on EICAS which was observed to be decaying from 60PSI. On reaching 45PSI, MGB OIL RESS amber caption illuminated and associated pressure indications stabilised at approximately 40PSI. Drills mpleted iaw EOP 5/3. Aircraft speed reduced to 80kts and turned towards nearest point of land. PAN call itated. After approximately 45 seconds, pressure observed to steadily increase and caption extinguished. Aircraft covered to base via running landing. MGB visually inspected for leaks and cracks , no leaks apparent. MGB oil rel normal and filter bowl button not popped. Pressure switches and wiring inspected with no sign of any visual fect. A/C sent for 30 minute hover/flight test, A/C assessed as serviceable.□

A Closure: 🗆

estigation was inconclusive and suspected to be a transient defect.

1057Z radar, pre noted 60 miles north west of the field diverting with a cracked windscreen, not declaring an ergency. A local standby was initiated for runway 15 landing safely at 11:46.

201417950	27/12/2014	EUROCOPTER	EC225	En route	PAN declared and aircraft returned due to lightning strike.	Aircl repc ackr Sup Aircl clou typic aircr safe of a agre and appl for a from pass VMC thro path insta
201500867	22/01/2015	SIKORSKY	S92	En route	PAN declared and aircraft returned due to Alternating Current (AC) generator failure.	App poin 7700 whe othe Sup No. ² Sup UTC Eme term
201414356	10/10/2014	SIKORSKY	S92	En route	PAN declared and aircraft returned due to LH cowling loose.	On f and assis had trans Sup Afte engi we f whils turn aircr from depa

craft outbound reported a lightning strike and requested immediate descent and return. I acknowledged the ort and advised of no known traffic to affect. Aircraft reported systems appeared normal and declared a PAN. I nowledged the PAN and instructed aircraft to squawk 7700. I handed over the sector to the next controller.

raft was cruising at standard outbound altitude of 3000ft, intermittent IMC with light rime ice. Main structure of ud was Scattered Cu with bases at approx 1500ft and tops estimated at 5000ft. Aircraft was intermittent IMC in cal 'good outside showers' weather. Some showers became evident on radar approximately 15nm ahead of the raft but were tracked crossing left to right with no 'red centres' outside 2-3nm. Aircraft track appeared to pass ely around the radar showers. At approximately 042-ADN-116nm, whilst intermittent IMC, the crew were aware flash and a sharp 'tap', which was audible above the background noise of the aircraft. Both crew immediately eed that a possible lightning strike of the aircraft had occurred and immediately checked all systems for correct safe operation. No failures or sub-optimal operations on any system were detected, however, after roximately 5 minutes, whilst in a subsequent descent, TAWS failed but recovered soon after. ATC were asked an immediate descent to 1000ft to ensure continuous VMC and a left 90 degrees turn to separate the aircraft n its original track. Once level at 1000ft and a subsequent assessment of all systems had been made, the sengers were informed. After the aircraft had achieved approximately 5nm track separation and remained good C the crew elected to declare an emergency via a PAN call and route direct back. Flight continued WFI, VMC ughout. Aircraft landed safely and shutdown. Engineering informed. On the Triggered Lightning print, a clear outside all amber and red sectors, direct to the destination, was evident out to the vicinity of the Harding allation, where an amber band was evident. Furthermore, with an aircraft ETA, the Triggered Lightning print cated that this amber area would move South and clear of the destination for the aircraft's arrival and this d by the print M/hen Op rox 40 NE aircraft called a PAN with an EC Generator fault and requested a diversion back (a/c departure nt). Aircraft was given a L or R turn back to the a/d. Having turned towards aircraft was requested to squawk 00 and report SOB (16). Aircraft requested a descent in about 10 miles so a clearance to descend was issued en ready. Aircraft was later given the Wx and a VFR clearance to enter CAS. Aircraft landed safely on RWY. No

er a/c were delayed.□

plementary 22/01/15:□ 1 AC Generator Fail. Actions iaw EOPs & RTB.□

plementary 24/01/15:

C radar informed ADC that aircraft was declaring a PAN due to EC generator failure and was returning. A full ergency was declared as per local instructions. The helicopter landed safely and taxied without incident to ninal to drop off passengers before returning to operator hangar to shut-down.

hand over the previous controller had told me that the aircraft had declared that he had a technical problem I requested to descend to altitude 2000ft and return. When asked the pilot replied that he did not require istance at that time. At 1006 aircraft declared a PAN, stating that a passenger had indicated to them that they I a loose panel. He was given a direct routing back to the field and instructed to squawk A7700. The aircraft was insferred to Tower and landed safely at 1024.

plementary 10/10/14:

er take-off, levelled in the cruise and part of the cruise checks we did an EPAC for both engines. The left hand gine did give us strange readings and we discussed that we do another EPAC on the way back. Shortly after this had both FADEC 1 and FADEC 2 fault warning coming on in flight. We went into the Emergency checklist and ilst we are discussing the problem it cleared itself. Five minutes later again a FADEC fault, now we decide to n back to base. In the turn I tell the passengers that we are turning back because of a minor problem with the craft. Shortly after a passenger comes forward and tells us that the left side cowling is open. I could not see this m the cockpit but I treated this as true and declared a PAN call, slowed down our airspeed and landed back at parture airport. Company investigation underway.

201500092	03/01/2015	SIKORSKY	S92	Scatsta	On lifting into the hover, an unusual high pitched metallic noise became apparent.	At th offlo CAA Inve shea no a no fi
201504267	05/04/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	Transmission chip warning. Aircraft returned.	At to air to infor
201504268	06/04/2015	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	Dual FADEC fault captions illuminated.	Duri com entry Sup Aircu retu safe

he same time a passenger approached the cockpit to inform us of the same. Aircraft landed, passengers baded and aircraft shut down. Engineering advice sought.□ A Closure:□

estigations into the noise revealed that a cabin heating supply hose under the forward cabin floor section had eared from a union. The hose was removed and replaced. No further evidence that this is a common failure and additional incidents of this type within the operator's reliability reporting system. Aircraft returned to service with further issues.

top of climb out on takeoff - approx 1,000ft : CAUTION- XMSN - CHIP illuminated on post engine replacement test. EOP Drill 7/7 actioned, CHIP light remained illuminated, all other parameters remained normal. Air traffic ormed. Joined circuit downwind to land R/w 34. Uneventful landing. Engineering informed. Tech log entry mad.

ring cruise checks at the top of the first climb, both FADEC FAULT captions illuminated for ECU 1 & 2. ECL npleted, decision taken to RTB. Pan call made, priority landing conducted without further incident. Tech log ry made sheet 267.□

craft on a standard IFR departure called a PAN with FADEC failure on both engines. Requested immediate urn and was given VFR clearance to enter the zone DCT to the field and transferred to Tower. Aircraft landed ely with no further complications.

plementary 06/4/15:

201504020	01/04/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared and aircraft returned due to cowl warning light.	Airc avoi retu was
201506382	12/05/2015	SIKORSKY	S92	EGPM (SCS): Scatsta	Aircraft returned due to failure of active vibration control (AVC) system during flight.	AVC
201507228	28/05/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to smoke warning in the cargo hold during approach.	I wa aircr clea visu situa esco Sup On a actio to sp

craft northbound climbed to FL85 north to avoid weather. At FL85 he requested lateral weather avoidance to bid towering CU. At 0755 he declared a pan call due to a cowl warning light and requested a descent to 2A and urn. The aircraft turned back and descended. Further reports from the pilot indicated that the cowl warning light s still flickering. As the aircraft approached at 2A he cancelled his pan but continued back regardless.

C system failed. Switched off and on as per EOPs no change. Tried IBIT no change. Although this aircraft is nnically serviceable without AVC the vibration level was unacceptable for passengers and crew. RTB.

as on duty as ADC when at approximately 1558 the aircraft declared a PAN with smoke in the cargo hold. The craft was already on a right base, Number 3 overall, for RW32, with another aircraft on short final RW34 already ared to land. The aircraft was immediately cleared to final RW32 Number 1, and the previously number 2 on a ual approach was instructed to enter the ATF hold and maintain A2500': INT were appraised of the uation. GMC initiated a Full Emergency. PAN aircraft landed RW32 at 1600, and taxied to park with AFS cort.

plementary 28/05/15:

approach, the smoke warning started to flicker and then remained steady on. The checklist was consulted and ioned. Due to the proximity of the airfield, a PAN was declared and priority landing was granted. Aircraft taxied spot and was shut down. Passengers were briefed in the terminal building.

201507282	29/05/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to nr1 engine chip caution. Engine power reduced to idle until finals and reinstated for running landing.	At 1: pron ackr I bel leav arise Sup Duri secc reap runn
201507482	04/06/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to nr2 engine chip warning. Engine shut down.	In th actic The were
201507901	15/06/2015	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	PAN declared due to nr2 engine failure on approach.	Whi 2 en A ru the / Sup I plu in or for r whe the r aircr the 0 0950

1514 I was just completing a call to GMC when Helicopter inbound from Montrose rig called on frequency. I omptly finished the call and responded to A/c. She immediately called PAN with an engine shutdown. I knowledged the PAN, told her the service and weather, ascertained the problem and if she could maintain 2000'. elieve it was a precautionary shut down, she was slowing slightly but there weren't any other issues. I elected to we her on her own squawk. There were two helicopter options following her for shepherding should the need se. The helicopter landed safely at 1554.□

plementary Rep from Operator 01/06/15

ring the cruise at approx 90 miles from ADN, CAUT, ENG and CHIP 1 illuminated then extinguished. 30 conds later, the same lights illuminated and EOPS were followed. On both pulses, CHIP 1 extinguished then ppeared so engine 1 brought back to IDLE until finals for runway 34. At this point, engine 1 reinstated for ning landing

the cruise at 2000ft returning the #2 engine chip warning light illuminated. The ECL (Emergency Checklist) was ioned, leading to a shutdown of the affected engine. Pan declared to Air Traffic Control. Passengers briefed. e flight continued back where a single engine landing was performed with no further issues. The passengers re debriefed by the Aircraft Commander and given an opportunity to ask questions.

ilst turning finals, the power was reduced. Almost immediately the No 2 fuel pressure caution came on, the No ngine failed. The engine failed and relit repeatedly. A PAN call was declared and full length runway requested. unning landing was completed and then the No 2 Engine selected to STOP. ATC reported a puff of smoke from A/c. A/c was shut down without further incident.□

plementary 15/06/15:

lugged in on GMC at 0900 with a UCE whilst undergoing a competency check. I then handed over and plugged on ADC at 0930. At 0935, the aircraft checked in approaching the field. He was told to report downwind left hand runway 34, and then subsequently to finals. As I cleared another aircraft for take off, the first was on a left base en he declared a PAN with engine malfunction. I let the second aircraft continue to depart as it would clear off e runway quicker than taxiing off. I initiated a Full Emergency (however I pressed the Omni-Crash Line) as the craft was on a 1 mile final. He landed safely at 0945. On landing it was observed that smoke had come out of e engines, but the pilot confirmed that it maybe because they used their fire suppressors. The fire service sponded and followed the aircraft back to stand. The Full emergency was downgraded to a local standby at 56, then subsequently to cancel the local standby at 1003.

201507971	12/06/2015	EUROCOPTER	EC225	EGPD (ABZ): Aberdeen/Dyce	MAYDAY due to nr1 engine fire.	Airc
201508413	19/06/2015	SIKORSKY	S92	EGPB (LSI): Sumburgh	Aircraft returned due to a minor technical issue. Local standby initiated.	
201508745	26/06/2015	EUROCOPTER	EC225	Transocean Prospect	During missed approach XMSN and IGB-T light warnings illuminated. Aircraft returned.	Duri inter abn cont

rcraft was ground running on the apron, the pilot called a Mayday reporting an engine fire on number 1 engine. I tiated an Aircraft Ground Incident.

uring GA the crew were presented with a caution light XSMN and IGB-T on the VMS, all illuminations were termittent. Once the GA was complete the crew joined the hold and carried out the EOPs. Line test proved phormal. As the weather at the Rig had resulted in a GA, the crew elected to return. As the indications were philontinually intermittent the crew elected to request priority assistance via ATC. Pax were briefed on the situation ver the PA and then face to face by the captain after shutdown.

	4 0 / 0 0 / 5 5 / -		0.00			1
201011337	19/08/2015		092	EGNJ (HUY): Humberside		hear hand impa shut
201512084	20/08/2015	SIKORSKY	S92	EGPD (ABZ): Aberdeen/Dyce	Engine 2 Chip light illuminated.	Inbo flow Engi Iand As a

on reaching cruising altitude of 2000ft aircraft was levelling and accelerating to a cruise speed. A loud bang was ard in the cabin and rear crew reported the main cabin door emergency exit had detached from the door. Aircraft adling and controls were checked and with safe flight assured crew attempted to inspect aircraft for any sign of bact or damage. As no issues were visible and the aircraft was operating correctly R912 routed direct back, to utdown. Engineering informed en route and inspection carried out on shutdown.

ound from the Rig, Engine Chip on engine 2 came on. Checklist was followed, PAN declared and aircraft was on back at 90 KIAS, engine number 2 at idle. A fuel transfer was required to keep the C of G within balance. gine was brought back to full power on short final, fuel switched back to direct and a single engine rolling ding performed. Passengers were debriefed in the aircraft □

a result of Analysis, metal found in chip detector and engine replaced