Civil Aviation Authority
United Kingdom

TYPE-CERTIFICATE DATA SHEET

UK.TC.A.00079

for

L-13 “BLANÍK”

Type Certificate Holder

Blanik Aircraft CZ s.r.o.

Beranových 65

199 00 Praha 9 - Letňany

Czech Republic

Model(s):	L-13 “BLANÍK” L-13 AC BLANÍK L 13 A Blaník
Issue:	1
Date of issue:	12 April 2023

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Section 1 L-13 "BLANÍK"

I. General

1.	Model:	L-13 "BLANÍK"
2.	Airworthiness Category:	Acrobatic Cloud flying
3.	Type Certificate Holder:	Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic
4.	Manufacturer:	Strojírny první pětiletky n.p. Uherské Hradiště - Kunovice 1177 Czechoslovakia: from S/N 170101 to S/N 170920, from S/N 171001 to S/N 171930, from S/N 172001 to S/N 172960, from S/N 173001 to S/N 173930, from S/N 174001 to S/N 174930, from S/N 175001 to S/N 175330, LET, n.p. Uherské Hradiště - Kunovice 1177 Czechoslovakia: from S/N 025401 to S/N 025930, from S/N 026001 to S/N 026960, from S/N 027001 to S/N 027361, from S/N 817401 to S/N 817403, from S/N 827404 to S/N 827421,
5.	Certification Application Date:	---
6.	CAA CZ Type Certification Date:	May 29, 1959
7.	EASA Type Certification Date:	February 4, 2005

8. The EASA Type Certificate replaced Czech Type Certificate No. 2725-59.

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II. Certification Basis

1.	Reference Date for determining the applicable requirements:	---
2.	Certification Basis:	- Bauvorschriften für Segelflugzeuge (BSV) issued August 1939 - BCAR, Section E, issued June 16, 1966
3.	Airworthiness Requirements:	see (2) above
4.	Requirements elected to comply:	CS 22.627, Amdt. 2 (Fatigue strength) for sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.
5.	EASA Special Conditions:	None
6.	EASA Exemptions:	None
7.	EASA Equivalent Safety Findings:	None

III. Technical Characteristics and Operational Limitations

1.	Type Design Definition:	Drawing No. A 101 310 N
2.	Description:	All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder.
3.	Equipment:	Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 four-point safety harnesses (symmetric) 2 parachutes or backrests (approx. 10 cm thick when compressed)
4.	Dimensions:	
	Span	16.2 m
	Length	8.4 m
	Height	2.1 m

Wing Area	19.15 m²
Aspect Ratio	13,7

5. Launching Hooks:

Nose tow release Dwg. No. A 740 210 N, or

nose tow release "E85", LBA Type Certificate No. 60.230/1

Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P

"Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or

"Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or

"Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

6. Weak links:

Ultimate Strength:

- for winch launching	max. 6300 N
- for aero-tow	max. 6300 N

7. Air Speeds:

Manoeuvering Speed V_A	145 km/h IAS	78 kts IAS
Never Exceed Speed V_{NE}		
up to 2500 m MSL	253 km/h IAS	136 kts IAS
2500 to 3000 m MSL	245 km/h IAS	131 kts IAS
3000 to 4000 m MSL	230 km/h IAS	123 kts IAS
4000 to 5000 m MSL	215 km/h IAS	115 kts IAS
5000 to 6000 m MSL	201 km/h IAS	108 kts IAS
6000 to 7000 m MSL	187 km/h IAS	100 kts IAS
7000 to 8000 m MSL	174 km/h IAS	93 kts IAS
8000 to 9000 m MSL	161 km/h IAS	86 kts IAS
9000 to 10000 m MSL	150 km/h IAS	80 kts IAS
Rough Air Speed V_{RA}	145 km/h IAS	78 kts IAS
Max. Aerotow Speed V_T	140 km/h IAS	76 kts IAS
Max. Flap Extended Speed V_{FE}	110 km/h IAS	60 kts IAS
Max. Winch-launch Speed V_W	120 km/h IAS	65 kts IAS
Max. Landing Gear Operating Speed V_{LO}	---	

8. Operational Capability:

VFR Day

Cloud flying

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9. Maximum Weights:

Maximum Take-Off Weight:	500 kg
Maximum Weight of non-lifting parts:	355 kg
Empty Weight:	292 kg ± 2% original unreinforced design
	306 kg ± 2% reinforced as per MB L13/117a

10. Centre of Gravity Range: Fore most c.g. limit aft of reference plane 112 mm
Aft most c.g. limit aft of reference plane 300 mm
[MAC is 1253 mm]

11. Datum: Wing leading edge at root rib

12. Levelling Means: Leveling points on fuselage

13. Minimum Flight Crew: 1 (Pilot)

14. Maximum Passenger Seating Capacity: 1

15. Lifetime limitations: Refer to Maintenance Manual

16. Deflection angles of control surfaces:

Elevator	up	32° + 2°
	down	25° ± 1°
Rudder	left, right	30° + 1°
Ailerons	up	34° + 2°
	down	13° + 2°
Wing flaps	down	08° ± 1°
Elevator trim tab	up	12° ± 1°
	down	35° ± 1°

IV. Operating and Service Instructions

1. Flight Manual:

In Czech language	
Do-L13-1111.1	Směrnice pro pilota větroně L 13
In English language	
Do-L13-1111.3	Pilots Notes for the L-13 Sailplane
In German language	
Do-L13-1111.2	Fluganweisung fur das Segelflugzeug L-13
In Russian language	
Do-L13-1111.5	Rukovodstvo po letnoj ekspluatacii planera L 13
In Spanish language	

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Do-L13-1111.4	Planeador L 13 Blaník Instructiones Para el Piloto
In English language	
Do-L13-1111.6	L-13 „BLANÍK“ Sailplane Flight Manual *)
*)	For sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.

2. Maintenance Manual:

In Czech language	
Do-L13-1132.1	Technická příručka větroně L 13
Do-L13-1131.1	Příručka pro provoz a údržbu větroně L 13 Blaník bez generálních oprav
In English language	
Do-L13-1132.3	Technical Manual of the L 13 Sailplane,
Do-L13-1131.3	Manual for Operation and Maintenance of the L 13 Blaník Sailplane without overhauls
In German language	
Do-L13-1132.2	Technisches Handbuch fur das Segelflugzeug L-13
In Russian language	
Do-L13-1131.5	Rukovodstvo po techničeskoj ekspluatacii L 13 Blaník bez kapitalnych remontov

3. Illustrated Parts Catalogue:

In Czech language	
Do L13-2121.6	Kusovník větroně L 13 Blaník (C-A-N)
In English language	
Do-L13-2121.6	Spare Parts Catalogue L 13 Blaník (C-A-N)
In German language	
Do-L13 2121.6	Katalog der Bestandteile L-13 Blaník

4. Overhaul Manual

In Czech language	
Do-L13-3031.1	Příručka pro generální opravu kluzáku L 13, L 13A
In English language	
Do-L13-3031.3	Overhaul Manual for L 13, L 13A Gliders
In Russian language	
Do-L13-3131.5	Rukovodstvo po kapitalnomu remontu planera L 13

5. Operating Manuals for Tow Releases

In German and in English language

Operating Manual for Nose Tow Releases TOST “Europa E 85”

Operating Manual for Safety Tow Releases TOST “Europa G 88”

Operating Manual for Safety Tow Releases TOST “Europa G 72” and “Europa G 73”

V. Notes

1. Six-position Serial Numbers starting with 17 precede the six-position Serial Numbers starting with 02.
2. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
3. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
4. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

Section 2 L-13 AC BLANÍK

I. General

1. Model:	L-13 AC BLANÍK
2. Airworthiness Category:	Acrobatic Cloud flying
3. Type Certificate Holder:	Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic
4. Manufacturer:	LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC from S/N 988601 to S/N 988604 from S/N 008605 to S/N 008606 LETECKÉ ZÁVODY a.s. 686 04 Kunovice 1177 CZECH REPUBLIC S/N 018901 and from S/N 028902 to S/N 028905 and S/N 029101
5. Certification Application Date:	---
6. CAA CZ Type Certification Date:	July 15, 1999
7. EASA Type Certification Date:	February 4, 2005

8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59

II. Certification Basis

1. Reference Date for determining the applicable requirements:	August 31, 1998

2.	Certification Basis:	CRI-A-01, issue 2, issued August 31, 1998
3.	Airworthiness Requirements:	BCAR, Section E, issued June 6, 1966
4.	Requirements elected to comply:	None
5.	EASA Special Conditions:	Appendix H, Joint Airworthiness Requirements, Sailplanes and Powered Sailplanes, Change 5 of October 28, 1995
6.	EASA Exemptions:	None
7.	EASA Equivalent Safety Findings:	None

III. Technical Characteristics and Operational Limitations

1.	Type Design Definition:	Drawing No. A 500 020 N
2.	Description:	L - 13 AC BLANÍK sailplane is all-metal, cantilever, highwing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake and a tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder. The S/N 018901 and from S/N 039102 there are installed ailerons outer stops, left aileron balance tab and there exists an option of wing tip extensions installation. Among the standard equipment there belongs AMU-1B recording unit.
3.	Equipment:	Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 five-point safety harnesses (symmetric) 2 accelerometers 2 parachutes or backrests (approx. 10 cm thick when compressed) 1 AMU-1B recording unit
4.	Dimensions:	
	Span	14.2 m without wing-tip extensions 16.2 m with wing-tip extensions
	Length	8.4 m
	Height	2.09 m
	Wing Area	17.44 m^2 without wing-tip extensions

	19.15 m^2 with wing-tip extensions
Aspect Ratio	11.186 without wing-tip extensions
	13.7 with wing-tip extensions

5. Launching Hooks:

Nose tow release Dwg. No. A 740 210 N, or

nose tow release "E85", LBA Type Certificate No. 60.230/1

Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P

"Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

6. Weak links:

Ultimate Strength:

- for winch launching	max. 6230 N
- for aero-tow	max. 6230 N

7. Air Speeds:

Air speeds of sailplane without wing tip extensions:

Manoeuvering Speed V_A	160 km/h IAS
Never Exceed Speed V_{NE}	
up to 2500 m MSL	230 km/h IAS
2500 to 3000 m MSL	223 km/h IAS
3000 to 4000 m MSL	209 km/h IAS
4000 to 5000 m MSL	195 km/h IAS
5000 to 6000 m MSL	182 km/h IAS
6000 to 7000 m MSL	170 km/h IAS
7000 to 8000 m MSL	158 km/h IAS
8000 to 9000 m MSL	147 km/h IAS
9000 to 10000 m MSL	136 km/h IAS
Rough Air Speed V_{RA}	160 km/h
Max. Aerotow Speed V_T	150 km/h
Max. Winch-launch Speed V_W	120 km/h
Max. Landing Gear Operating Speed V_{LO}	---

Air speeds of sailplane with wing tip extensions:

Maneuvering Speed V_A	150 km/h IAS
Never Exceed Speed V_{NE}	
up to 2500 m MSL	230 km/h IAS
2500 to 3000 m MSL	223 km/h IAS
3000 to 4000 m MSL	209 km/h IAS
4000 to 5000 m MSL	195 km/h IAS
5000 to 6000 m MSL	182 km/h IAS

6000 to 7000 m MSL	170 km/h IAS
7000 to 8000 m MSL	158 km/h IAS
8000 to 9000 m MSL	147 km/h IAS
9000 to 10000 m MSL	136 km/h IAS
Rough Air Speed V_{RA}	150 km/h
Max. Aerotow Speed V_T	150 km/h
Max. Winch-launch Speed V_W	120 km/h
Max. Landing Gear Operating Speed V_{LO}	---

8. Operational Capability: VFR Day
   Cloud flying

9. Maximum Weights:

Maximum Weight:	500 kg (without extensions)
	510 kg (with extensions)
Maximum Weight of non-lifting parts:	355 kg
Empty Weight:	305 kg ± 2% (without extensions)
	315 kg ± 2% (with extensions)

10. Centre of Gravity Range: Fore most c.g. limit aft of reference plane 143 mm
    Aft most c.g. limit aft of reference plane 337 mm
    [MAC is 1295 mm without wing tip extensions]
    [MAC is 1252.3 mm with wing tip extensions]
11. Datum: Wing leading edge at root rib

12. Levelling Means: Leveling points on fuselage in horizontal position.
13. Minimum Flight Crew: 1 (Pilot)
14. Maximum Passenger Seating Capacity: 1
15. Lifetime limitations: Refer to Maintenance Manual

16. Deflection angles of control surfaces:

Elevator	up	32° + 2°
	down	27° ± 1°
Rudder	left, right	29° + 1°
Ailerons	up	34° + 2°
	down	13° + 2°
Left aileron balance tab	up	20° ± 2°
	down	15° ± 2°
Elevator trim tab	up	12° ± 1°
	down	35° ± 1°

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IV. Operating and Service Instructions

1. Flight Manual:

In Czech language	
Do-L13AC-1013.1	Letová příručka L 13 AC Blaník (to S/N 008606 from S/N 028902 to 029101)
Do-L13AC-1014.0	Letová příručka L 13 AC Blaník (S/N 018901 from S/N 039102)
In English language	
Do-L13AC-1013.3	Sailplane Flight Manual L 13AC Blaník to S/N 008606 from S/N 028902 to 029101
Do-L13AC-1014.2	Sailplane Flight Manual L 13AC Blaník (S/N 018901 from S/N039102)

2. Maintenance Manual:

In Czech language	
Do-L13AC-1032.1	Provozně technická příručka kluzáku L 13 AC
In English language	
Do-L13AC-1032.3	Maintenance Manual for the Sailplane L 13 AC

3. Illustrated Parts Catalogue:

In Czech language	
Do-L13AC-2051.0	Katalog dílů a montážních jednotek kluzáku L-13AC Blaník (C/A)
In English language	
Do-L13AC-2051.0	Illustrated parts catalogue for the sailplane L 13 AC Blaník (C/A)

4. Operation book

In Czech language	
Do-L13AC.1015.02	L 13AC Blanik Záznamník provozu kluzáku (C/A)
In English language	
Do-L13AC.1015.02	L 13 AC Blanik Sailplane operation book of records (C/A)

5. Operating Manuals for Tow Releases

In German and in English language

Operating Manual for Nose Tow Releases TOST "Europa E 85"

Operating Manual for Safety Tow Releases TOST "Europa G 88"

V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

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Section 3 L 13 A Blaník

I. General

1.	Model:	L 13 A Blaník
2.	Airworthiness Category:	Acrobatic Cloud flying
3.	Type Certificate Holder:	Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic
4.	Manufacturer:	LET, n.p. 686 04 Kunovice 1177 CZECH REPUBLIC from S/N 817401 to S/N 827421 LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC from S/N 968501 to S/N 968505
5.	Certification Application Date:	---
6.	CAA CZ Type Certification Date:	December 16, 1981
7.	EASA Type Certification Date:	12 August 2005
8.	The EASA Type Certificate replaces Czech Type Certificate No. 2725-59	

II. Certification Basis

1.	Reference Date for determining the applicable requirements:	---
2.	Certification Basis:	BCAR, Section E, issued June 6, 1966

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3.	Airworthiness Requirements:	see (2) above
4.	Requirements elected to comply:	None
5.	EASA Special Conditions:	None
6.	EASA Exemptions:	None
7.	EASA Equivalent Safety Findings:	None

III. Technical Characteristics and Operational Limitations

1.	Type Design Definition:	Drawing No. A 101 310 N
2.	Description:	All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder.
3.	Equipment:	Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 four-point safety harnesses (symmetric) 2 parachutes or backrests (approx. 10 cm thick when compressed)
4.	Dimensions:	
	Span	16.2 m
	Length	8.4 m
	Height	2.1 m
	Wing Area	19.15 m^2
	Aspect Ratio	13.7
5.	Launching Hooks:	Nose tow release of type SR-L13.225, or nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1 Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or

"Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or

"Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

6. Weak links: Ultimate Strength:

- for winch launching	max. 6230 N
- for aero-tow	max. 6230 N

7. Air Speeds:

Manoeuvering Speed V_A	145 km/h IAS
Never Exceed Speed V_{NE}	
up to 2500 m MSL	253 km/h IAS
2500 to 3000 m MSL	245 km/h IAS
3000 to 4000 m MSL	230 km/h IAS
4000 to 5000 m MSL	215 km/h IAS
5000 to 6000 m MSL	201 km/h IAS
6000 to 7000 m MSL	187 km/h IAS
7000 to 8000 m MSL	174 km/h IAS
8000 to 9000 m MSL	161 km/h IAS
9000 to 10000 m MSL	150 km/h IAS
Rough Air Speed V_{RA}	145 km/h
Max. Aerotow Speed V_T	140 km/h
Max. Winch-launch Speed V_W	120 km/h
Max. Landing Gear Operating Speed V_{LO}	---

8. Operational Capability: VFR Day

Cloud flying

9. Maximum Weights:

Maximum Weight:	500 kg
Maximum Weight of non-lifting parts:	355 kg
Empty Weight:	306 kg ± 2%

10. Centre of Gravity Range: Fore most c.g. limit aft of reference plane 112 mm

Aft most c.g. limit aft of reference plane 300 mm

[MAC is 1253 mm]

11. Datum: Wing leading edge at root rib

12. Levelling Means: Leveling points on fuselage in horizontal position.

13. Minimum Flight Crew: 1 (Pilot)

14. Maximum Passenger Seating Capacity: 1

15. Lifetime limitations: Refer to Maintenance Manual

16. Deflection angles of control surfaces:

Elevator	up	32° + 2°
	down	25° ± 1°
Rudder	left, right	30° + 1°
Ailerons	up	34° + 2°
	down	13° + 2°
Wing flaps	down	8° ± 1°
Elevator trim tab	up	12° ± 1°
	down	35° ± 1°

IV. Operating and Service Instructions

1. Flight Manual:

In Czech language	
Do-L13A-1011.1	Letová příručka L 13 A
In English language	
Do-L13A-1011.3	Flight Manual of the L 13 A
In German language	
Do-L13A-1011.2	Flughandbuch fur das Segelflug L 13 A

2. Maintenance Manual:

In Czech language	
Do-L13A-1031.1	Příručka pro obsluhu, údržbu a opravy kluzáku L 13
In English language	
Do-L13A-1031.3	Technical Manual of the L 13 A Sailplane
In German language	
Do-L13A-1031.2	Handbuch fur die Reparatur, Instandhaltung und Wartung des Segelflugzeuges L 13 A

3. Illustrated Parts Catalogue:

In Czech language	
Do-L13-2126.6	Kusovník větroně L 13 Blaník (C-A-N)
In English language	
Do-L13-2126.6	Spare Parts Catalogue L 13 Blaník (C-A-N)

In German language

Do-L13-2126.6	Katalog der Bestandteile L-13 Blaník (C-A-N)

4. Overhaul Manual

In Czech language

Do-L13-3031.1	Příručka pro generální opravu kluzáku L 13, L 13A

In English language

Do-L13-3031.3	Overhaul Manual for L 13, L 13A Gliders

5. Sailplane Technical Description L 13A

In Czech language

Do-L13A-1021.1	Sailplane Technical Description L 13A

6. Operating Manuals for Tow Releases

In German and in English language

Operating Manual for Nose Tow Releases TOST "Europa E 85"

Operating Manual for Safety Tow Releases TOST "Europa G 88"

Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73"

V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

Section 4 Administration

I. Acronyms and Abbreviations

Acronym / Abbreviation	Definition
CAA	Civil Aviation Authority
CAA CZ	Civil Aviation Authority Czech Republic
EASA	European Union Aviation Safety Agency
Kg	Kilogram
L	Litres
TC	Type Certificate
TCDS	Type Certificate Data Sheet
TCH	Type Certificate Holder
VFR	Visual Flight Rules

II. Type Certificate Holder Record

TCH Record	Period
Blanik Aircraft CZ s.r.o. Beranových 65 199 00 Praha 9 - Letňany Czech Republic Beranových 65	Present. No changes.

III. Amendment Record

TCDS Issue No.	TCDS Issue Date	Changes	TC Issue and Date
1	12 Apr 2023	This certificate supersedes EASA.A.024 in the UK. All technical data as per EASA.A.024 Issue 11. Correction of S/N per manufacturer and correction of Maximum Empty Weight.	Issue 1 12 Apr 2023

– END –