

European Aviation Safety Agency

EASA
TYPE-CERTIFICATE
DATA SHEET

VSO 10

Type Certificate Holder:

Schempp-Hirth výroba letadel spol. s r.o
U Dvořiska 1733
565 01 Choceň
CZECH REPUBLIC

EASA TCDS No. A.442

For variants: VSO 10
VSO 10 C (see A.V. Notes)

Issue 01, 14 February 2007

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0.III. Change Record

Issue	Date	Changes
01	14 February 2007	Transfer from Czech Type Certificate No.: 79-01 to the EASA Type Design

Section A: VSO 10

A.I. General

1. Data Sheet No.: EASA.A.442
2. a) Type: VSO 10
b) Variant: VSO 10 C (see A.V. Notes)
3. Airworthiness Category: Utility
4. Type Certificate Holder: Schempp-Hirth výroba letadel spol. s r.o.
U Dvořiska 1733
565 01 Choceň
CZECH REPUBLIC
5. Manufacturer:
From S/N 150 001 to S/N 150 206
Orličan, n.p.
565 37 Choceň
CZECH REPUBLIC
From S/N 150207 to S/N 150 225
Orličan, s.p.
565 37 Choceň
CZECH REPUBLIC
6. CAA CZ Type Certification Date: 15 May 1979
7. This TCDS replaces Czech Type Certificate No.: 79-01

A.II. Certification Basis

1. Certification Basis: L 8/O Regulation, effective July 7, 1976
2. Airworthiness Requirements: L 8/O Regulation, effective July 7, 1976
3. Requirements elected to comply: None
4. Special Conditions: None
5. Exemptions:
 - 2.1.4.b Control forces are higher than required.
 - 2.1.7 Static friction force of aileron control is higher than required.
 - 2.8.1 Spin recovery after five turns was not demonstrated. Sailplane is not capable to complete more than 4 turns.
 - 2.9.1 During side slips with airbrakes extended and greater rudder deflection higher empennage vibrations occur. Side slips with max. 1/2 rudder deflection enclosed in the operation limitations.
 - 3.5.4 The requirement of 50 percent elongation of rudder control circuit not met
6. Equivalent Safety Findings: None

A.III. Technical Characteristics and Operational Limitations

- | | | | | |
|-----|--------------------------|--|---------------------|----------|
| 1. | Type Design Definition: | Parts list S-VSO 10. | | |
| 2. | Description: | VSO 10 sailplane is single-seat cantilever shoulder-wing monoplane with closed cockpit and retractable landing gear with shock absorber. Two-piece wing of wooden structure with plywood sandwich skin and airbrakes on the upper and lower sides. T-tail empennage of aluminium alloy structure with control surfaces coated by fabric. Front part of the fuselage is made from glass composite and the rear one from aluminium alloys. | | |
| 3. | Equipment: | Minimum equipment:
1 Airspeed indicator
1 Altimeter
1 Variometer
1 Magnetic compass
1 Turn and bank indicator
Four-point safety harness
Parachute | | |
| 4. | Dimensions: | Span | 15,00 m | |
| | | Length | 6,10 m | |
| | | Height | 1,38 m | |
| | | Wing area | 9,30 m ² | |
| | | Aspect Ratio | 18,75 | |
| 5. | Launching Hooks: | Nose | VSO 10.417-01 | |
| | | Side | VSO 10.417.03/04 | |
| 6. | Weak links: | Ultimate Strength: | max.500 daN | |
| 7. | Air Speeds: | Never Exceed Speed | V_{NE} | 250 km/h |
| | | Maximum permitted speeds | | |
| | | - in rough air | V_{RA} | 160 km/h |
| | | - in aero-tow | V_T | 160 km/h |
| | | - in winch-launch | V_W | 120 km/h |
| 8. | Load Factors: | At $V_A = 160$ km/h | n | = +5,5 |
| | | | n | = -3,5 |
| | | At $V_{NE} = 250$ km/h | n | = +4,6 |
| | | | n | = -2,6 |
| 9. | Operational Capability: | Approved for VFR-Day. | | |
| 10. | Maximum Masses: | Maximum permitted take-off mass | 380 kg | |
| | | Empty mass | 250 kg +- 3% | |
| 11. | Centre of Gravity Range: | Datum: Leading edge of the wing root rib. Mean Aerodynamic Chord 0,824 m, front point on the datum. | | |
| | | Leveling means: Wedge 323:15 on the rear top fuselage, horizontal. | | |
| | | 26 – 48 % MAC (231 až 379 mm behind the datum). | | |
| 12. | Seating Capacity: | 1 | | |
| 13. | Lifetime limitations: | See "Návod k obsluze kluzáku VSO 10" | | |

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|--|----------|------|----------|
| 14. Deflection of control surfaces:
Ruderausschläge | Ailerons | down | +12° ±1° |
| | | up | -30° ±2° |
| | Elevator | down | +16° -1° |
| | | up | -17° +1° |
| | Rudder | | ±30° -3° |
15. Wheels and Tyres: Light alloy wheel with brake HP 4741.
Tyre 350 × 135.

A.IV. Operating and Service Instructions

1. Sailplane Flight Manual:
- in Czech language: "Letová příručka", vydání I
"Letová příručka", vydání II (year 1982)
"Letová příručka", vydání III (year 1987)
2. Sailplane Operating Instructions:
- in Czech language: "Návod k obsluze kluzáku VSO 10", vydání II (1982)
"Návod k obsluze kluzáku VSO 10", vydání III (1987)
3. Sailplane Log Book:
- in Czech language: "Záznamník kluzáku VSO 10"
4. Sailplane Maintenance Directives:
- in Czech language: "Směrnice pro provádění prohlídek po 500 hodinách
VS-VSO10-011", vydání 2004

A.V. Notes

1. The VSO 10 C variant differs from VSO 10 Model only in application of fixed landing gear. (see A III. Par.2) It was manufactured exclusively by Orličan n.p. Choceň (see A.I. Par.4.)