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# TYPE-CERTIFICATE DATA SHEET

No. EASA.A.451

for

SZD-56 "Diana"

Type Certificate Holder

## Avionic Spółka Jawna Bolesław Kawik - Leszek Matuszek

ul. Stary Dwór 9 43-436 Górki Wielkie POLAND

Models: SZD-56-1 "Diana"

SZD-56-2 "Diana-2" (Restricted Type Certificate)

Issue 03, 05/04/2017

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## Section A: SZD-56-1 "Diana"

#### A.I. General

1. Data Sheet No.: EASA.A.451

2. a) Type: SZD-56 "Diana"b) Model: SZD-56-1 "Diana"

c) Variant:

3. Airworthiness Category: Sailplane - Utility (U) Category

4. Manufacturer: 1. Przedsiębiorstwo Doświadczalno-Produkcyjne

Szybownictwa (PDPSz) "PZL - Bielsko"

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2. Biuro Projektowe "B" Bogumił Bereś

5. Polish CAA Certification Date: 23 June 1997 (TC No. BG-203)

6. The EASA TC replaces Polish Type Certificate No. BG-203/1, which replaced the BG-203 on 05 February 2001, due to TC transfer from PDPSz "PZL-Bielsko" to Biuro Projektowe "B" Bogumił Bereś.

#### A.II. Certification Basis

1. Reference Date January 1995

for determining the applicable requirements:

2. Airworthiness Requirements: JAR-22, Change 4, issued on 7 May 1987, with

Amendments up to No. 22/94/1 dated on 10

November 1994

3. Special Conditions: None

4. Exemptions: None

5. Deviations: None

6. Equivalent Safety Findings: None

7. Requirements elected to comply: None

8. Environmental Standards: None

### A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: List of Drawings for sailplane SZD-56-1 "Diana"

2. Description: Single-seat high-performance sailplane of 15-meter

flapped class.

Cantilever mid-wing monoplane with T-shaped tail unit.

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All composite carbon-aramid-epoxy structure.

Bipartite monocoque wing of multi-tapered contour and

constant aerofoil section NN-27-13. Wings mounted on fixed fuselage spar.

Plate airbrakes protruding only from upper surface.

Flaperons over the whole wing span. Water ballast tanks in wings and fin.

Retractable main landing gear and fixed tail wheel. Main wheel with drum brake and shock absorber.

3. Equipment: Standard equipment:

- airspeed indicator,

altimeter,compass,

- rate-of-climb indicator,

- towing hook,

- pilot safety belts (4-point),

- first aid kit.

4. Dimensions: Span: 15,00 m

Wing area: 8,16 m<sup>2</sup>
Aspect ratio: 27,57
Length: 6,88 m
Height: 1,35 m

5. Launching Hook: SZD-III P or TOST E85

6. Weak Links: Nominal strength: 677 daN (±10%)

7. Load Factors: +5.3 / -2.65 (up to  $V_A$ )

+4.0 / -1.5 (up to  $V_{NE}$ )

Flaps down deflected +4.0 (up to  $V_{FE}$ ) Airbrakes extended +3.5 (up to  $V_{NE}$ )

8. Air Speeds (IAS): Never Exceed Speed V<sub>NE</sub> 270 km/h

Manoeuvring Speed V<sub>A</sub> 195 km/h

Maximum permitted speeds:

- flaps up (from -4° to 0°)  $V_{FE} = 270 \hspace{0.1cm} km/h$  - flaps down (over 0° up to +14°)  $V_{FE} = 175 \hspace{0.1cm} km/h$  - in rough air  $V_{RA} = 195 \hspace{0.1cm} km/h$  - in aero-tow  $V_{T} = 140 \hspace{0.1cm} km/h$ 

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9. Operational Capability: Approved for VFR-day conditions.

10. Masses: Max. mass with water ballast: 410 kg

> Max. mass without water ballast: 297 kg

> Empty mass: 182 kg

11. Centre of Gravity Range: Empty glider with standard equipment:

Forward and Rearward Limits depend on empty glider mass

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and are shown in Flight Manual - point 2.5.

Centre of Gravity operational limits:

Forward limit: 105 mm aft of the datum

(19,0 % MAC)

257 mm aft of the datum Rearward limit:

(45,0 % MAC)

0% MAC is on the same coordinate MAC is 570,6 mm;

22° ±1°

along longitudinal axis as the datum.

Leading edge and wing-fuselage division plane intersection. 12. Datum:

Leading and trailing points of the root chord (670 mm) 13. Levelling Means:

at the same level.

14. Control surface Aileron: - up deflections:

12° ±1° - down

34° ±1° Elevator: - up

> 18° ±1° - down

30° ±1° Rudder: - left

> 33° ±1° - right

Flaps: 4° ±1° - up

> 14° ±1° - down

15. Minimum Flight Crew: 1 pilot

16. Maximum Passenger

Seating Capacity:

No Passengers

17. Baggage/ Cargo

Compartments:

Refer to the Flight Manual

18. Lifetime Limitations: Refer to the Technical Service Manual

19. Other Limitations: The following is prohibited:

- night flying,

cloud flying,

- flights in icing conditions,

- aerobatics,

flights with water ballast at outside temperature

below 0°C,

- winch launching.

A.IV. Operating and Service Instructions

1. Flight Manuals:

Polish • Instrukcja Użytkowania w Locie Szybowca

SZD-56-1 "Diana", wydanie II - wrzesień 2001

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• Flight Manual of Sailplane SZD-56-1 "Diana",

Issue II - September 2001

2. Maintenance Manuals:

Polish • Instrukcja Obsługi Technicznej Szybowca

SZD-56-1 "Diana", wydanie I - maj 1997

English
 Technical Service Manual of Sailplane SZD-56-1

"Diana", Issue I - May 1997

#### A.V. Notes:

1. This TCDS, Section A applies to the following S/N:

X-147; X-152,

561197001, 561101002, 561101003

2. All glider outside surfaces must be white painted. Neither registration number nor any colour marks on the wings and stabilizer upper surfaces are allowed

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## Section B: SZD-56-2 "Diana-2"

#### **B.I.** General

1. Data Sheet No.: EASA.A.451

2. a) Type: SZD-56 "Diana"b) Model: SZD-56-2 "Diana-2"

c) Variant:

3. Airworthiness Category: Restricted Category

Sailplane - Utility (U) Category

4. Manufacturer: 1. Biuro Projektowe "B" Bogumił Bereś

Avionic Spółka Jawna,
 Bolesław Kawik - Leszek Matuszek

5. Polish CAA Certification Application Date:

26 April 2004

6. EASA Certification Application Date:

25 October 2011

7. EASA Certification Date: 21 December 2015

The SZD-56-2 "Diana-2" model has got

Restricted Type Certificate

#### **B.II.** Certification Basis

 Reference Date for determining the applicable requirements: 26 April 2004

2. Airworthiness Requirements:

JAR-22, Amendment 7, dated 1 September 2003

3. Special Conditions:

None

4. Restrictions:

- Normal landing with water ballast is banned.
   In case of emergency landing with water ballast, an inspection is required.
- Maximum crosswind component during take-off with water ballast – 5 m/s
- Take-offs with water ballast from runway with high grass – prohibited
- Operation is limited to holder of an appropriate valid license for sailplanes
- Operation is limited to pilots who have accumulated more than 100 hours of flight experience on other flapped sailplanes.

5. Deviations: None

6. Equivalent Safety Findings: • JAR 22.201(g), 221(a) – CRI B-01 Stall and spin

characteristics with water ballast asymmetry

• JAR 22.473, 479, 485, 723, 731 – CRI C-01

Landing gear loads

7. Requirements elected to comply: None

8. Environmental Standards: None

#### B.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: List of Drawings, Document No. 562-DSG-06,

issued in December 2004, revision 1 dated 15.01.2005

or later approved revision.

2. Description: Single-seat high-performance sailplane of 15-meter

flapped class.

Cantilever mid-wing monoplane with T-shaped tail unit.

All composite carbon-aramid-epoxy structure.

Bipartite monocoque wing of curvilinear contour and variable along span aerofoil section of KL-002-12\*F/17

family, with winglets.

Wings mounted on fixed fuselage spar.

Plate airbrakes protruding only from upper surface.

Flaperons over the whole wing span. Water ballast tanks in wings and fin.

Retractable main landing gear and fixed tail wheel. Main wheel with drum brake and shock absorber.

3. Equipment: Standard equipment:

- airspeed indicator,

altimeter,compass,

- rate-of-climb indicator,

- towing hook,

- pilot safety belts (4-point),

outside air temperature gauge.

4. Dimensions: Span: 15,00 m

Wing area: 8,64 m<sup>2</sup>
Aspect ratio: 26,04
Length: 6,88 m
Height: 1,35 m

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5.	Launching Hook:	SZD-III A 56	or TOST	E85
J.	Lauriching Hook.	32D-III A 30 (	01 1031	ᆮ

6.	Weak Links:	Nominal strength:	677 daN	(±10%)	)

7.	Load Factors:	+5,3 / -2,65	(up to $V_A$ )
		, ,	\ 1 '''

$$+4.0 / -1.5$$
 (up to  $V_{NE}$ )

+4,0 / 0,0 (up to  $V_{FE}$ ) Flaps down deflected

(over  $+8^{\circ}$  up to  $+28^{\circ}$ )

Airbrakes extended +3,5 / -1,5 (up to  $V_{NE}$ )

#### 8. Air Speeds (IAS): Never Exceed Speed $V_{\mathsf{NF}}$ 275 km/h

198 km/h Manoeuvring Speed  $V_A$ 

Maximum permitted speeds:

- flaps up (from -2° to +8°) 275 km/h  $V_{\mathsf{FE}}$ - flaps down (over +8° up to +28°)  $V_{\mathsf{FE}}$ 212 km/h - in rough air  $V_{\mathsf{RA}}$ 198 km/h  $V_T$ 139 km/h - in aero-tow

9. Operational Capability: Approved for VFR-day conditions.

10. Masses: Max. mass with water ballast: 500 kg

> Max. mass without water ballast: 297 kg Max. mass for landing: 297 kg Min. mass without water ballast: 237 kg

11. Centre of Gravity Range: Empty glider with standard equipment:

Forward and Rearward Limits depend on empty glider mass

and are shown in Flight Manual - point 2.5.

Centre of Gravity operational limits:

Forward limit: 202 mm aft of the datum

(19,0 % MAC)

Rearward limit: 362 mm aft of the datum

(45,0 % MAC)

MAC is 615 mm; 0% MAC lies 85 mm aft of the datum.

12. Datum: Leading edge and wing-fuselage division plane intersection.

13. Levelling Means: Upper contour of the tail boom in horizontal position

14. Control surface 18° ±1° Aileron: - up deflections: 12° ±1° - down

Elevator: 30° ±1° - up

- down 27° ±1°

Rudder: - left 30° ±1° 33° ±1°

- right 2° ±1° - up

Flaps: - down 28° ±1° TCDS No. EASA.A.451 Issue 03 SZD-56 "Diana" Section B: SZD-56-2 "Diana-2"

15. Minimum Flight Crew: 1 pilot

16. Maximum Passenger Seating Capacity:

No Passengers

17. Baggage/ Cargo Compartments:

Refer to the Flight Manual

18. Lifetime Limitations: Refer to the Technical Service Manual

19. Other Limitations: The following is prohibited:

night flying,cloud flying,

- flights in icing conditions,

- aerobatics,

- flights with water ballast at outside temperature

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below 0°C,

- winch launching.

#### **B.IV.** Operating and Service Instructions

1. Flight Manuals:

Polish • Instrukcja Użytkowania w Locie Szybowca

SZD-56-2 "Diana-2", wydanie II, lipiec 2015

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English
 Flight Manual for a Sailplane SZD-56-2 "Diana-2",

Issue II, July 2015

2. Maintenance Manuals:

Polish • Instrukcja Obsługi Technicznej Szybowca

SZD-56-2 "Diana-2", wydanie II, lipiec 2015

• Technical Service Manual for a Sailplane SZD-56-2

"Diana-2", Issue II, July 2015

#### B.V. Notes:

 This TCDS, Section B applies to S/N XB-01 and 5621yynnn, starting from 562105001 where:

yy - the year of the aircraft manufacture, nnn - the successive aircraft number.

2. All glider outside surfaces must be white painted. Neither registration number nor any colour marks on the wings and stabilizer upper surfaces are allowed

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## **Administrative section**

### <u>Acronyms</u>

CRI Certification Review Item
MAC Mean Aerodynamic Chord
S/N Aircraft Serial Number
VFR Visual Flight Rules

#### **Type Certificate Holder Record**

1. since 2007 till 2017:

Biuro Projektowe "B" Bogumił Bereś ul. Goleszowska 4/136 43-308 Bielsko-Biała POLAND

2. since 2017:

Avionic Spółka Jawna, Bolesław Kawik - Leszek Matuszek

ul. Stary Dwór 9

43-436 Górki Wielkie

**POLAND** 

According to the bilateral agreement, actions and obligations required from the TC Holder by paragraph 21.A.44 (Annex I to Commission Regulation (EU) No 748/2012) are undertaken by:

Zakłady Lotnicze Margański & Mysłowski S.A. (DOA No. EASA.21J.117)

ul. Strażacka 60

43-300 Bielsko-Biała

**POLAND** 

#### **Change Record**

Issue	Date	Changes
Issue 01	28 Mar 2007	Transfer from Polish Type Certificate No. BG-203/1 to the EASA Type Design
Issue 02	21 Dec 2015	The SZD-56-2 "Diana-2" model added
Issue 03	05 Apr 2017	Type Certificate transfer to the new TC Holder