

# TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.598

for Breezer B600

Type Certificate Holder Breezer Aircraft GmbH & Co. KG

> Sönke-Nissen-Koog 58 25821 Reußenköge Germany

For models: B600



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#### SECTION A: BREEZER B600

#### A.I General

- 1. Type/ Model/ Variant
  - 1.1 Type: Breezer B600
  - 1.2 Model: B600
  - 1.3 Variant: N/A
- 2. Airworthiness Category: Restricted
- 3. Manufacturer: Breezer Aircraft GmbH & Co. KG,
- 4. EASA Type Certification Application Date: 10 September 2012
- 5. EASA Type Certification Date: 22 January 2016

#### A.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements: 10 September 2012
- Airworthiness Requirements: Certification Specifications for Light Sport Aeroplanes CS-LSA, Initial Issue, June 27, 2011 For detailed information see CRI-A01
   Special Conditions: none
   Exemptions: none
   Elected to Comply: CS-23 acc. CC: 001-M-12-E-CC
   Equivalent Safety Findings: none
   Environmental Protection: ICAO Annex 16, Volume I (for more details see EASA TCDSN A.598)



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#### A.III Technical Characteristics and Operational Limitations

1.	Type Design Definition:	001-M-12-MZ-TDD, Issue 1, Revision 2 or later approved revisions.			
2.	Description:	Single engine,	two seats side-b	py-side, all metal, low-wing,	
		fixed tricycle landing gear with steerable nose wheel.			
3.	Equipment:	For Minimum Equipment List refer to POH, Chapter II.11			
4.	Dimensions:	Wingspan	8.74 m	(344 in)	
		Length	6.73 m	(265 in)	
		Height	2.14 m	(84.3 in)	
		Wing area	11.89 m²	(127.97 sq ft)	

#### 5. Engine

5.1 Model: ROTAX 912 ULS2

5.2	Type Certificate:	Certified as part of the aircraft
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5.3 Limitations:

Power:	Max. take-off: 73,5 kW at 5800 rpm (max. 5 min)
	Max. continuous: 69,0 kW at 5500 rpm
Oil pressure:	Min: 0,8 bar
	Normal: 2,0 – 5,0 bar
	Max. 7,0 bar cold engine starting
Oil temperature	: Max. 130 °C
	Min. 50 °C
Fuel pressure:	Max. 0,4 bar
	Min. 0,15 bar
Coolant temp.	Max. 120 °C

- 6. (Reserved)
- 7. Propeller
  - Propeller 1

- 7.2 Type Certificate: Certified as part of the aircraft
- 7.3 Number of blades: 3 blades
- 7.4 Diameter: 1700 mm
- 7.5 Sense of Rotation: Clockwise in flight direction

#### Propeller 2

- 7.6
   Model:
   NEUFORM CR3-75-(IP)-47-101,6
- 7.7 Type Certificate: Certified as part of the aircraft
- 7.8 Number of blades: 3 blades
- 7.9 Diameter: 1750 mm
- 7.10 Sense of Rotation: Clockwise in flight direction



8.	Fluids							
	8.1 Fuel:	Use a minimum ROZ 95 gas, MOGAS, AVGAS UL 91 or AVGAS						
		100 LL. Refer to ROTAX Op			(Operat	Operating Manual.		
		The use	e of E10/	/E85 fue	el is proh	ibited.		
	8.2 Oil:	Engine	oil as pe	er ROTA	X Opera	ting Manual and	l SI-912-016:	
		min. gr	ade API	"SG" or	higher,	chose viscosity a	according to	
		climate	conditi	ons. Red	commen	ded oil brands li	isted in	
		ROTAX	SI-912-(	016.				
	8.3 Coolant	Conven	tional c	ooling f	luid with	a mixture ratio	of 1:1 with	
		water a	is per RO	ΟΤΑΧ ΟΙ	perating	Manual and SI-9	912-016.	
		Recom	mended	cooling	g fluids li	sted in ROTAX S	I-912-016.	
9.	Fluid capacities							
	9.1 Fuel:	76 litre	s (74 litr	es usab	le)			
	9.2 Oil:	Min: ap	prox. 2.	5 litres				
		Max: ap	oprox. 3	.0 litres				
	9.3 Coolant system capacity:	Approx	. 2.2 litr	es				
10.	Air Speeds:	$V_{FE}$ :	126 km	h IAS	(68 KIA	S)		
		V <sub>A</sub> :	179 km	h IAS	(97 KIA	S)		
		V <sub>NO</sub> :	197 km	h IAS	(106 KI	AS)		
		V <sub>NE</sub> :	252 km	h IAS	(136 KI	AS)		
11.	Maximum Certified Operating Alti	tude:	13.000	ft				
12.	Approved Operations Capability:		Day-VF	R				
13.	Maximum Masses:		Take-of	ff: 600 k	g (1320	lbs)		
			Landing	g: 600 k	g (1320 l	bs)		
14.	Centre of Gravity Range:	Empty	weight:	Forwa	rd limit:	280 mm	(20.6 % MAC)	
				Rear li	mit:	300 mm	(22.0% MAC)	
		Flight:		Forwar	d limit:	269 mm	(19.8 % MAC)	
				Rear li	nit:	435 mm	(32.0% MAC)	
15.	Datum:	Referer	nce datu	ım is the	e wing le	ading edge.		
16.	(Reserved)							
17.	Levelling Means:	A Horiz	ontal pla	ain thro	ugh the	upper edge of t	he	
		canopy guardrail						
18.	3. Minimum Flight Crew: 1 (Pilot)							
19.	Maximum Passenger Seating Capa	city:	1					
20.	Baggage/ Cargo Compartments:		ggage co aggage l	•	nent is lo 15 kg (3	ocated behind th	ne seats.	
21	Wheels and Tyres:		heel tyr		5.00-5			
			heel tyr		4.00-4			
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22. (Reserved)



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### A.IV Operating and Service Instructions

1. Flight Manual:	, РОН-В600-Е-Е		
	German: POH-B600-A-	D, POH-B600-E-D	
2. Maintenance Manual:	ММ-В600-Е		
3. Structural Repair Manual:	Refer to Maintenance	Manual	
4. Weight and Balance Manual:	Refer to Flight Manual		
5. Illustrated Parts Catalogue:	N/A		
6. Operating Manual and Maintenance	e Manual for Engine:	Operating Manual:	OM-912
		Maintenance Manual:	MMH-912,
			MML-912
7. Operating Manual and Maintenance	e Manual for Propeller:	NEUFORM-Adjustable	Pitch Propeller
		Operating manual,	
		NEUFORM–3-Blade-Va	riable Pitch
		Propeller R2 Series Ope	erating Manual

#### A.V <u>Notes</u>

n/a



## SECTION B: ADMINISTRATIVE SECTION

#### B.I Acronyms & Abbreviations

CS-LSA	Certification Specification for Light Sport Aeroplanes
EASA	European Aviation Safety Agency
IAS	Indicated Airspeed
KIAS	Indicated Airspeed [knots]
LSA	Light Sport Aeroplane
NAAC	

MAC Mean Aerodynamic Chord

TCDS Type Certificate Data Sheet

#### B.II <u>Type Certificate Holder Record</u>

Breezer Aircraft GmbH & Co. KG Sönke-Nissen-Koog 58 25821 Reußenköge Germany

#### B.III Change Record

Issue	Date	Changes	TC Issue No. & Date
Issue 01	22 January	Initial Issue	Initial Issue,
	2016		22 January
			2016

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