Temporary Revisions (TRs) apply to this MMEL, which have been placed at the front of the document for convenience. All TRs overwrite and supersede the corresponding entry in the MMEL, and therefore must be incorporated in the document.

Please follow the instructions on each TR carefully, ensuring that the TR pages are inserted facing the effective page(s) in the MMEL.

The TRs should be incorporated in the order in which they were issued, as it is possible that a TR may be superseded by a later one.

Additionally please incorporate/amend the temporary revision record page and amend the list of effective pages accordingly.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: **L188**

18 November 1996

APPLICABLE TO NORMAL REVISION No: TWO

ATA 35 - OXYGEN

Insert in Master Minimum Equipment List facing page 35-2

ACTION: Record the incorporation on the temporary revision record page and

amend the list of effective pages accordingly.

REASON FOR ISSUE: Items 2,3 and 4 omitted from revision 2 in error. Revision 2 highlights

page correct - item 2 unchanged from revision 1, Note added to item

3 and item 4 is new.

2. Crew Oxygen System and Supply	1	0	As required by Air Navigation Legislation. Refer to Flight Manual for crew requirements.
3. Portable Oxygen Dispensing Units (bottle and Mask)	-	-	As required by Air Navigation Legislation. Any in excess of those required by legislation may be inoperative.
			NOTE The portable supplies required by scales L1 and L2 are totally separate from the requirements of scale R2.
4. Protective Breathing Equipment	-	-	As required by Air Navigation Legislation.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

REASON FOR ISSUE:

Amended various entries in line with current CAA MMEL Policy. Corrected entries for Fuelling Bay Fuel Cap and Engine Air Scoop and Inlet Vane Icing Lights.

Added entry for Navigation Database.

ACTION:

- 1. Insert page 1 of this TR after the Temporary Revision record page.
- 2. Insert page 2 of this TR facing page 23-1 of the MMEL.
- 3. Insert page 3 of this TR facing page 23-4 of the MMEL.
- 4. Insert page 4 of this TR facing page 26-1 of the MMEL.
- 5. Insert page 5 of this TR facing page 28-3 of the MMEL.
- 6. Insert page 6 of this TR facing page 30-1 of the MMEL.
- 7. Insert page 7 of this TR facing page 31-1 of the MMEL.
- 8. Insert page 8 of this TR facing page 33-2 of the MMEL.
- 9. Insert page 9 of this TR facing page 34-5 of the MMEL.
- 10. Insert page 10 of this TR facing page 34-5 of the MMEL.11. Insert page 11 of this TR facing page 34-6 of the MMEL.
- 12. Insert page 12 of this TR facing page 34-6 of the MMEL.
- 13. Insert page 13 of this TR facing page 34-7 of the MMEL.
- 14. Record the incorporation on the Temporary Revision record page and amend the list of effective pages accordingly.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 23 - COMMUNICATIONS

Insert this page facing page 23-1 of the MMEL.

2. Passenger Address System (If installed)	1	0	(O) As required by Air Navigation Legislation. May be inoperative provided:
			(a) The flight deck / cabin interphone system (including chime system) is operative,
			(b) Alternate normal and emergency procedures are established and used, and
			(c) Repairs or replacements are carried out within three calendar days.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 23 - COMMUNICATIONS

Insert this page facing page 23-4 of the MMEL.

11. Cockpit Voice Recorder (CVR)	1	0	(O) As required by Air Navigation Legislation. May be inoperative provided:
			(a) It is not reasonably practical to repair or replace before commencement of flight,
			(b) The aircraft shall not exceed eight (8) consecutive flights with the CVR unserviceable beginning with the first flight after the CVR was last operating throughout the flight,
			(c) Not more than 72 hours have elapsed since the CVR became unserviceable, and
			(d) The Flight Data Recorder (if required by
			Air Navigation Legislation) must be operating normally.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 26 - FIRE PROTECTION

Insert this page facing page 26-1 of the MMEL.

- 5. Toilet Compartment Smoke Detection System (If installed)
- (M) May be inoperative provided:
 - (a) Toilet compartment is electrically isolated (including toilet flush motor and all other high voltage devices) as applicable,
 - (b) Toilet waste bin is empty,
 - (c) Toilet door is locked closed and placarded,
 - (d) Toilet is not used for any other purpose, and
 - (e) Repairs or replacements are carried out within ten calendar days.
- | | May be inoperative provided:
 - (a) Toilet fire extinguishers are fitted and operating normally,
 - (b) The toilet is checked at regular and frequent intervals for evidence of fire and smoke, and
 - (c) Repairs or replacements are carried out within three calendar days.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 28 - FUEL

Insert this page facing page 28-3 of the MMEL.

7. Fuelling Bay Fuel Cap	1	0	May be missing provided:
			(a) Refuelling receptacle is verified to be free of contamination prior to each refuelling,
			(b) Receptacle is verified to have no leakage afterwards,
			(c) All fuel tank quantity indicators are operative, and
			(d) Repairs or replacements are carried out within three calendar days.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 30 - ICE AND RAIN PROTECTION

Insert this page facing page 30-1 of the MMEL.

2. Engine Air Scoop and Inlet Vane Icing Lights	4	2	(M) Two may be inoperative provided:
			(a) Associated fuel flow indicators are operative,
			(b) Associated horsepower indicators are operative, and
			(c) Repairs or replacements are carried out within three calendar days.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 31 - INDICATING AND RECORDING SYSTEMS

Insert this page facing page 31-1 of the MMEL.

2. Flight Data Recorder (FDR)	1	0	As required by Air Navigation Legislation. May be inoperative provided:
			(a) It is not reasonably practical to repair or replace before commencement of flight,
			(b) The aircraft shall not exceed eight (8) consecutive flights with the FDR unserviceable beginning with the first flight after the FDR was last operating throughout the flight,
			(c) Not more than 72 hours have elapsed since the FDR became unserviceable, and
			(d) The Cockpit Voice Recorder must be operating normally.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 33 - LIGHTS

Insert this page facing page 33-2 of the MMEL.

insert this page racing page 33-2 of the MMLL.					
5. Anti-Collision Lights	-	1	 (O) As required by Air Navigation Legislation. Any in excess of one may be inoperative provided: (a) A high intensity strobe light system is installed and operative, (b) The light(s) is (are) repaired at the earliest practicable opportunity, and (c) Repairs or replacements are carried out within ten calendar days. 		
	-	0	 (O) As required by Air Navigation Legislation. All may be inoperative for daylight operations provided: (a) Operations are limited to flights within the UK FIR only, unless prior permission is obtained from the appropriate Air Traffic Control Unit for the intended route to be flown, (b) The light(s) is (are) repaired at the 		
			earliest opportunity, and (c) Repairs or replacements are carried out within ten calendar days. Note: If the red anti-collision light is inoperative, alternative procedures must be developed and used when the aircraft is on the ground with the engine(s) running.		
(1) Strobe Lights	-	0	All may be inoperative provided repairs or replacements are carried out within ten calendar days.		

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 34 - NAVIGATION

Insert this page facing page 34-5 of the MMEL.

23. Ground Proximity Warning System (GPWS)			
(1) Terrain Avoidance Modes 1-4	-	0	May be inoperative provided: (a) Two navigation systems (not including
			ADF and marker beacons) are operative, and (b) Repairs are made within 6 further flights
(2) Test Mode	_	0	or 25 flying hours or 2 calendar days, whichever occurs first.
(2) Test Nade			May be inoperative provided:
			(a) The GPWS is considered inoperative, and
			(b) Repairs are made within 6 further flights or 25 flying hours or 2 calendar days,
(3) Glideslope Deviation (Mode 5)	-	0	whichever occurs first.
`	-	0	May be inoperative provided repairs or replacements are made within three calendar days.
			May be inoperative for day VMC only provided repairs or replacements are made within ten calendar days.
			(Cont)

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 34 - NAVIGATION

Insert this page facing page 34-5 of the MMEL.

23. Ground Proximity Warning System (GPWS) (Cont)			
(4) Advisory Callouts (If installed)	-	0	(O) May be inoperative provided alternate procedures are established and used, and repairs or replacements are carried out within ten calendar days.
			Note: Check Flight Manual Limitations for approach minima.
(5) Windshear Mode (If installed)	-	0	(O) May be inoperative provided alternate procedures are established and used, and repairs or replacements are carried out within ten calendar days.
(6) Enhanced Mode (EGPWS)	-	0	May be inoperative provided the GPWS functions are operative.
			Note: Particular circumstances may require the use of additional or alternate procedures. The alternate procedures would require the operator to consider the routes over which he is flying and ensure that the pilot adopted a flight path which would give him the protection which would otherwise be afforded.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 34 - NAVIGATION

Insert this page facing page 34-6 of the MMEL.

24. Airborne Collision Avoidance System (ACAS) (If installed)			
(1) ACAS System	-	0	(O) (M) May be inoperative provided the system is deactivated and secured and
			(a) It is not reasonably practicable to repair before the commencement of flight, and
			(b) If the aeroplane is intended to be flown in airspace in which ACAS operation is required it may not fly for more than 10 calendar days with the equipment completely inoperative.
			Note: Local airspace requirement may impose a more restrictive rectification interval.
(2) Combined TA and RA Dual Displays	-	1	(O) May be inoperative on the non-flying pilot side provided:
			(a) TA and RA elements and audio functions are operative in flying pilot side,
			(b) TA and RA display indications are visible to the non-flying pilot, and
			(c) Repairs or replacements are carried out within ten calendar days.
			(Cont)

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 34 - NAVIGATION

Insert this page facing page 34-6 of the MMEL.

24. Airborne Collision Avoidance System (ACAS) (If installed) (Cont)			
(3) Resolution Advisory (RA) Display System(s)	-	1	(O) One may be inoperative on the non-flying pilot side provided repairs or replacements are carried out within ten calendar days.
	-	0	(O) One or more may be inoperative provided:
			(a) All Traffic Alert (TA) display elements and voice command audio functions are operative,
			(b) TA only mode is selected by the crew, and
			(c) Repairs or replacements are carried out within ten calendar days.
(4) Traffic Alert (TA) Display System(s)	-	0	(O) One or more may be inoperative provided:
			(a) All installed RA display and audio functions are operative, and
			(b) Repairs or replacements are carried out within ten calendar days.

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

AIRCRAFT TYPE: LOCKHEED L-188 ELECTRA

APPLICABLE TO NORMAL REVISION No: TWO

23 May 2001

ATA 34 - NAVIGATION

Insert this page facing page 34-7 of the MMEL.

29. Navigation Database (If installed)

- (O) May be out of currency provided:
- (a) Current aeronautical information is used to verify Navigation Fixes prior to dispatch,
- (b) Procedures are established to verify the status and suitability of Navigation Facilities used to define the route of flight, and
- (c) The navigation database is updated to the current standard within 10 calendar days.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

APPLICABLE TO CAA MMEL FOR THE FOLLOWING AIRCRAFT TYPES:

AIRCRAFT TYPE:	MMEL NORMAL REVISION No:
Airbus Industrie A300-600	2
Airbus Industrie A319/A320/A321	2
ATR 42	4
ATR 72	Initial issue
BAC 1-11	2
BAe (HS) 125 series B up to 800B	Initial issue
BAe (HS) 748	Initial issue
Beech F90/200/B200/B200C series	1
Beech B90/C90/C90A/E90	Initial issue
Beech 100/A100	Initial issue
Beechjet 400/400A and MU300	3
Boeing 707-300 series	Initial issue
Boeing 727-100 and 200 series	1
Boeing 737-100/200/300/400/500 series	3
Boeing 747-100/200 series	2
Boeing 747-400	3
Boeing 757	12
Boeing 767	Initial issue
Canadair Challenger	2
Cessna Citation CE-500 series	Initial issue
Cessna CE-525	Initial issue
Cessna Citation CE-650	Initial issue
Cessna CE-208/208A/208B	1
Cessna 401/402/404/411	Initial issue
Reims / Cessna 406/F406	Initial issue
Cessna 414/421	Initial issue
Cessna 425/441	Initial issue
Dassault Aviation Fan Jet (Falcon 20)	1
Dassault Aviation Mystere Falcon 900	Initial issue
Dassault Aviation Falcon 900EX	Initial issue
De Havilland DHC-6	3

Cont...

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

APPLICABLE TO CAA MMEL FOR THE FOLLOWING AIRCRAFT TYPES:

AIRCRAFT TYPE:	MMEL NORMAL REVISION No:
De Havilland DHC-7	3
De Havilland DHC-8	1
Dornier 228	1
Embraer EMB-110	2
Embraer EMB-120	2
Fokker F27	1
Fokker F100/F70	2
Gulfstream Aerospace Gulfstream IV	3
Islander BN-2A/BN-2B	1
Learjet 35/36/55	Initial issue
Lockheed L-188 Electra	2
Lockheed L-1011 Tristar	1
MCDonnell Douglas DC-10 (Models 10 and 30)	Initial issue
McDonnell Douglas DC-3	Initial issue
Piper PA31	3
Saab SF340A and 340B	1

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ACTION:

Insert pages 1, 2 and 3 of this TR after the TR Record page. Insert page 4 of this TR at the front of the Preamble section. Insert page 5 of this TR at the front of the Definitions section. Insert page 6 of this TR immediately before and facing page 23-1. Insert page 7 of this TR immediately before and facing page 25-1. Insert page 8 of this TR immediately before and facing page 31-1. Insert page 9 of this TR immediately before and facing page 34-1. Insert page 10 of this TR immediately before and facing page 34-1.

Record the incorporation on the temporary revision record page and amend the list of effective pages accordingly.

REASON FOR ISSUE:

The TR reflects current CAA MMEL Policy for Cockpit Voice Recorders, Emergency Locator Transmitters, Flight Data Recorders, ACAS II and GPWS.

The Definitions and Preamble sections have also been updated to reflect current CAA MMEL Policy.

NOTES

- 1. This TR replaces any existing alleviation given in the MMEL normal revision and/or any previous TR on the same subject.
- The existing MMEL numbering should be retained where applicable. In the absence of an applicable MMEL entry, the alleviation given in this TR should be added at the end of the relevant ATA chapter in the MMEL.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

PREAMBLE

Insert this page facing at the front of the Preamble section in the MMEL.

The CAA MMELs and Supplements are produced in conjunction with a base document, generally either the MMEL issued/approved by a Foreign Airworthiness Authority or the aircraft manufacturer at a specific quoted revision number and date. There may be occasions whereby the CAA MMEL or Supplement has not been updated to consider later revisions of the base document. This could lead to instances where there are alleviations in the base MMEL which have either been revised or deleted and are now more restrictive than the corresponding CAA MMEL or Supplement entry. Operators are invited to review all new base document MMEL revisions and where necessary advise the CAA MMEL section of any significantly more restrictive alleviations introduced by the revision. The CAA will then expedite review of these variations and, where required, issue amendments to the CAA MMEL or Supplement.

New or amended alleviations given in later issues of the base document shall not be used until the CAA MMEL or Supplement has been updated to confirm that issue of the base document is acceptable.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

DEFINITIONS

Insert this page facing at the front of the Definitions section in the MMEL.

"As required by Air Navigation Legislation / Operating Requirements": The associated item must comply with legal provisions such as the Air Navigation Order or any other legislation (JAR-OPS 1) in force during the flight.

Operators should refer to the JAR-OPS 1 MEL Policy document (Temporary Guidance Leaflet number 26) for suitable alleviations based upon the required equipment identified within JAR-OPS 1, subparts K and L (published in the JAA Administrative and Guidance, section four, Operations, part three).

<u>"It is not reasonably practicable for repairs or replacements to be made"</u>: This statement is intended to cover situations whereby there is a lack of a replacement part(s), inadequate engineering resources or manpower to enable the defect to be rectified.

<u>Flight</u>: For the purpose of a MEL, a flight is the period of time between the moment when an aeroplane begins to move by its own means, for the purpose of preparing for take-off, until the moment the aeroplane comes to a complete stop on its parking area, after the subsequent landing (and no subsequent take-off).

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ATA 23 - COMMUNICATIONS

Insert this page facing page 23-1 of the MMEL.

Cockpit Voice Recorder (CVR)	-	-	-	As required by Operating Requirements.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ATA 25 - EQUIPMENT / FURNISHINGS

Insert this page facing page 25-1 of the MMEL.

Emergency Locator Transmitter (ELT) (If installed)	A	-	-	May be inoperative provided repairs or replacements are made within 6 further flights or 25 flying hours, whichever occurs first.
	D	-	-	Any in excess of those required may be inoperative.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ATA 31 - INDICATING / RECORDING SYSTEMS

Insert this page facing page 31-1 of the MMEL.

Flight Data Recorder (FDR)

- - - As required by Operating Requirements.

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ATA 34 - NAVIGATION

Insert this page facing page 34-1 of the MMEL.

Airborne Collision and Avoidance System (ACAS II) (If installed)				
(1) ACAS II System	A	-	0	(O) (M) As required by Air Navigation Legislation. May be inoperative provided the system is deactivated and secured, and
				(a) The aircraft may continue the flight or series of flights but shall not depart an airport where it is reasonably practicable for repairs or replacements to be made, and
				(b) Repairs or replacements must be carried out within 10 calendar days.
				Note: Local airspace requirements may require a permission to proceed or impose a more restrictive rectification interval.
(2) Combined Traffic Alert (TA) Resolution Advisory (RA) Dual Displays	С	-	1	(O) May be inoperative on the non- flying pilot side provided TA and RA elements and audio functions are operative on the flying pilot side.
				(Cont)

29 October 2001

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ATA 34 - NAVIGATION

Insert this page facing page 34-1 of the MMEL.

Airborne Collision and Avoidance System (ACAS II) (If installed) (Cont.)				
(3) Resolution Advisory (RA) Display System(s)	С	-	1	(O) One may be inoperative on the non-flying pilot side .
				OR
	С	-	0	(O) May be inoperative provided:
				(a) All Traffic Alert (TA) display elements and voice command audio functions are operative, and
				(b) TA only mode is selected by the crew.
(4) Traffic Alert (TA) Display System(s)	С	-	0	(O) May be inoperative provided all installed RA display and audio functions are operative.
Ground Proximity Warning System (GPWS) (including TAWS)	-	-	-	As required by Operating Requirements.

20 March 2002

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

TR-G6 APPLICABLE TO CAA MMEL FOR THE FOLLOWING AIRCRAFT TYPES:

GLOBAL TEMPORARY REVISION INDEX

AIRCRAFT TYPE:	G1	G2	G3	G4	G5	G6
Airbus Industrie A300-600				√	V	√
Airbus Industrie A319/A320/A321 Supplement ATR 42				√ √	٧	
ATR 72				√	√	
BAC 1-11		J		√ √	٧	٦
BAe (HS) 125 series B up to 800B		•		1		1
BAe (HS) 748		√		√		1
Beech F90/200/B200/B200C	V			√		√
series Beech B90/C90/C90A/E90	√			√		√
Beech 100/A100	√			√		1
Beechjet 400/400A and MU300				√		√
Boeing 707-300 series				√		√
Boeing 727-100 and 200 series				√		
Boeing 737-100/200/300/400/500 series Supplement Boeing 747-100/200 series				√ √	√ √	
Boeing 747-400 Supplement				V	Ž	
Boeing 757 Supplement				Ž	Ž	
Boeing 767 Supplement				Ž	Ž	V
Canadair Challenger				V	,	V
Cessna Citation CE-500 series Supplement				1		,
Cessna CE-525 Supplement				√		
Cessna Citation CE-650 Supplement				√		
Cessna CE-208/208A/208B	1			√		√
Cessna 401/402/404/411	1			√.		√.
Reims / Cessna 406/F406	√.			√.		√.
Cessna 414/421	√.			√.		√.
Cessna 425/441	√			√		√

20 March 2002

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

GLOBAL TEMPORARY REVISION INDEX (Cont.)

AIRCRAFT TYPE:	G1	G2	G3	G4	G5	G6
Dassault Aviation Fan Jet				1		√
(Falcon 20) Dassault Aviation Mystere Falcon 900		√		1		√
Dassault Aviation Falcon 900EX				√		
De Havilland DHC-6	√			√		\checkmark
De Havilland DHC-7	√	\checkmark		\checkmark		√
De Havilland DHC-8				√	√	
Dornier 228	√			√		√
Embraer EMB-110	√			√		\checkmark
Embraer EMB-120				√		
Fokker F27	√	√		√	√	√
Fokker F100/F70 Supplement				√	√	
Gulfstream Aerospace				√		√
Gulfstream IV Islander BN-2A/BN-2B	J			N		J
Learjet 35/36/55	•			٠ ما		1
Lockheed L-188 Electra				٨		۷ ما
				Ŋ		Ŋ
Lockheed L-1011 Tristar				٧		٧
MCDonnell Douglas DC-10 (Models 10 and 30)				√	√	√
McDonnell Douglas DC-3				√		
Piper PA31	√			√		√
Saab SF340A and 340B Supplement				1	1	

Note: The TR-G prefix designates a global Temporary Revision which is a policy change applicable to several aircraft types. Please note that revisions of the MMEL may have incorporated (and superseded) the Temporary Revisions previously issued.

20 March 2002

MASTER MINIMUM EQUIPMENT LIST TEMPORARY REVISION

ACTION: Insert pages 1 and 2 of this TR immediately after the TR

record page.

Insert page 3 of this TR immediately before and facing page 34-1 of the MMEL (or S34-1 for MMEL Supplements).

Record the incorporation on the temporary revision record page and amend the list of effective pages accordingly.

REASON FOR ISSUE: Update MMELs to include current CAA MMEL Policy on Radio

Altimeters. Two notes have been introduced in order to ensure

that the applicable dispatch deviations are used if the GPWS/TAWS and ACAS systems are also inoperative.

If either of these notes already exists in the current MMEL entry (as a note or as part of the alleviation), the existing wording in the MMEL should remain. These notes should be incorporated only if the current MMEL entry does not refer to these systems. If the MMEL entry refers to GPWS but not ACAS, then only the note for ACAS need be incorporated.

ATA 34 – NAVIGATION

Insert this page facing page 34-1 of the MMEL.

The following notes should be added to the entry for Radio Altimeters:

Note 1: If the loss of the radio altimeter prohibits normal operation of the

GPWS/TAWS, the dispatch deviation and rectification interval for an

inoperative GPWS/TAWS must be observed.

Note 2: If the loss of the radio altimeter prohibits normal operation of the ACAS, the

dispatch deviation and rectification interval for an inoperative ACAS must be

observed.

MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

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MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

Revision 2 25 October 1996

REVISION 2

This Master Minimum Equipment List (MMEL) is issued by the Civil Aviation Authority at the above revision and is approved as the basis for the preparation and approval of individual operator's Minimum Equipment Lists (MELs) for aircraft of this Type.

Correspondence concerning this document should be addressed to the office listed below:-

Civil Aviation Authority Safety Regulation Group Aviation House South Area Gatwick Airport Gatwick West Sussex RH6 0YR

Attention:

Aircraft Projects MMEL Section

MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

Revision 2 25 October 1996

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MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

Revision 2 25 October 1996

REVISION RECORD

REVISION No.	ISSUE DATE	INCORPORATED BY	DATE
Revision 1	7 January 1992		
Revision 2	25 October 1996		

MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

Revision 2 25 October 1996

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MASTER MINIMUM EQUIPMENT LIST

LOCKHEED L-188 ELECTRA

Revision 2 25 October 1996

TEMPORARY REVISION RECORD

TR No.	Date	Page Affected	Incorporated By	Date Incorporation	Superseded By
1	18/11/96	35-2			
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MASTER MINIMUM EQUIPMENT LIST

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PREAMBLE

- 1. The CAA approved Master Minimum Equipment List (MMEL) provides owners/operators of United Kingdom registered aircraft, of the relevant type, with the basis for the preparation of their individual Minimum Equipment List (MELs). In the case of holders of Air Operators Certificates the MEL will be included in that Company's Operations Manual.
- 2. The approved MMEL represents a list of items of equipment which, under particular circumstances, can, to the satisfaction of the CAA, be unserviceable when the aircraft is despatched, while still retaining the required level of safety.
- 3. The CAA recognises that in some respects the standard and scale of equipment provided in the aircraft may exceed the minimum required to satisfy airworthiness or Air Navigation Legislation requirements. Where necessary to achieve a satisfactory level of safety with an inoperative item, appropriate limitations are imposed or the function transferred to another component.
- 4. The MMEL does not include items such as wings, engines and landing gear that are always required, nor is reference made to equipment such as passenger convenience and entertainment items which when inoperative obviously do not affect airworthiness. It is important to note therefore that ANY ITEM WHICH IS RELATED TO THE AIRWORTHINESS OF THE AIRCRAFT AND WHICH IS NOT INCLUDED IN THE MMEL IS ALWAYS REQUIRED TO BE OPERATIVE BEFORE A FLIGHT IS DESPATCHED. Likewise items required by Air Navigation Legislation. Additional Certification Requirements as appropriate, which are not listed must be operative.
- 5. The MMEL may not waive a limitation or an emergency procedure which is given in the Flight Manual (FM) or override an Airworthiness Directive (AD) /Mandatory Inspection unless the FM/AD provides otherwise. Similarly any Additional Certification Requirements, or other special provisions, as appropriate which have been determined as necessary by the CAA shall not be waived unless otherwise agreed or varied by the CAA.
- 6. An Owner/Operators MEL must receive CAA approval which thereby conveys the permission, required by the UK Air Navigation Order, to the Commander, for operation of the aircraft with specified items of equipment unserviceable.
- 7. The MEL may not be less restrictive than the MMEL therefore the number of items required for despatch shall not be less than the corresponding number in column 3 of the MMEL and any associated conditions shall be at least as severe as those specified in column 4.
- 8. The MMEL does not anticipate the effects of combinations of apparently unrelated unserviceabilities or allow for situations where systems are made inoperative for special purposes such as demonstration, test or crew training.
- 9. The MEL should indicate that a decision to operate the aircraft with multiple unserviceabilities should only be made after due consideration of possible interrelated or additive effects and, if necessary, following consultation with appropriate engineering specialists.

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PREAMBLE (Cont...)

10. It is not the purpose of the MMEL to allow defects of other than optional items to remain unrectified indefinitely. The operational flexibility provided under the MMEL policy is justified only within a framework of controlled and sound programmes of repairs, replacement and servicing. Defects should be rectified expeditiously thus retaining the intended overall level of safety and reducing the possibility of a subsequent failure necessitating the removal of the aircraft from service. Some particular items in the MMEL may be subject to a limitation of flight hours, number of flights or consecutive calendar days, and these must be transferred into the MEL.

A limit of three calendar days for completion of repairs or replacements has been applied to some items. Other time limits for rectification, such as those specified by the ANO, may also be applied as appropriate. Operators with established routes shall specify in the MMEL at which stations, in additional to the main maintenance base, repair facilities exist.

11. This MMEL is based upon UK legislation and some of the alleviations it provides may not therefore necessarily comply with foreign legislation.

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DEFINITIONS

- 1. In this list, the items of equipment are classified in systems according to the ATA 100 specification. Individual items within a given ATA classification are numbered sequentially.
- 2. "Item" (Column 1): The equipment, system, components or function as listed in Column 1.

NOTE: Items annotated in UPPER CASE letters indicates the precise flight deck legend used.

3. "Number Installed" (Column 2): The number of the specified items normally installed in the aircraft. This number identifies the aircraft configuration considered in developing the MMEL.

NOTE: The operator's MEL should list the number installed in a particular aircraft.

- 4. <u>"Number Required for Despatch"</u> (Column 3): The minimum number of the specified items required for operation provided the conditions defined in Column 4 are met.
- 5. <u>"Remarks or Exceptions"</u> (Column 4): This column includes a statement prohibiting operation or permitting operation with a specific number of items inoperative, provisos (conditions and limitations) for such operation and appropriate notes.
- 6. <u>Dash (-)</u>: This symbol indicates a variable quantity when used in Columns 2 or 3.

NOTE: The operator's MEL should list the numbers appropriate to his particular aircraft in Columns 2 and 3.

7. "Placarding" Each inoperative item must be placarded to inform and remind the crew members and maintenance personnel of the equipment condition. To the extent practicable, placards should be located adjacent to the control or indicator for the item affected such that it is clear to the operating crew that it or its associated system is inoperative.

NOTE: The practice of specifying which items must be placarded, by means of an asterisk (*), has been discontinued.

- 8. <u>"Inoperative"</u>: A system or item of equipment is deemed inoperative if it malfunctions such that it does not accomplish its intended purpose and/or is not consistently functioning within it's designed operating limit(s) or tolerance(s).
- 9. "(0)": The use of this symbol in Column 4 indicates that an appropriate operating procedure (or change to an existing procedure) must be established, published and utilised to maintain the required level of safety while operating under the terms of the (M)MEL.

Normally, these procedures are accomplished by the flight crew. However, other personnel may be qualified and authorised to perform certain functions.

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DEFINITIONS (Cont...)

10. "(M)": The use of this symbol in Column 4 indicates that an appropriate maintenance procedure must be established, published and utilised prior to the first flight undertaken following discovery of the defect and, if necessary, repeated at specified intervals during operation under the terms of the (M)MEL to maintain the required level of safety.

Normally, these procedures are accomplished by maintenance personnel. However, other personnel may be qualified and authorised to perform certain functions.

NOTE: Where an item is annotated (0)/(M), the "/" is defined as "and/or", which shows that there may be different options available in respect of the MEL procedures.

- 11. <u>"As required by Air Navigation Legislation"</u>: The associated item must comply with legal provisions such as the Air Navigation Order or any other legislation in force during the flight.
- 12. "VMC" and "IMC": The definitions of these terms are those used in Section 2 of the Air Navigation Order Rules of the air.
- 13. <u>"Icing Conditions"</u>: An atmospheric condition that may cause ice to form on the aircraft or in the engines.
- 14. <u>"Visible Moisture"</u>: An atmospheric environment containing water in any form that can be seen in natural or artificial light, i.e. clouds, fog, rain, sleet, hail, snow.
- 15. <u>"Flight Hour"</u>: The time from the moment an aircraft leaves the surface of the earth until it touches it at the next point of landing.
 - NOTE: The definition differs from that given in the Air Navigation Order.
- 16. <u>"ETOPS"</u>: Refers to "extended range" operations which may be defined as "operation of a two-engined aeroplane over a route that contains a point farther than one hour flying time at the normal one-engined inoperative cruise speed (in still air) from an adequate airport".
 - In the MEL, for an operator who has received approval to extend maximum diversion time from 120 minutes to 138 minutes, unless otherwise stated, "120 minutes" may be interpreted as "138 minutes".
- 17. <u>"Flight day"</u>: A 24 hour period (from midnight to midnight) during which at least one flight is scheduled for the affected aircraft.
- 18. <u>"Authority"</u>: The competent regulatory authority according to the country of registry; for aircraft registered in the UK this is the Civil Aviation Authority.
- 19. <u>"Deleted"</u>: When applied to an item number, indicates that the item was previously listed but is now required to be operative.

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DEFINITIONS (Cont...)

20. "Combustible (Material)": is defined as material which is capable of catching fire and burning.

When an MMEL item specifies the condition that only non-combustible materials are to be carried, it is the operator's responsibility to determine that all material (including containers, packing material and pallets etc) in the associated compartments is of a non-combustible nature.

If it cannot be determined whether any proposed cargo is non-combustible, it must not be loaded in compartments where combustible materials are prohibited.

- 21. <u>"System"</u>: System means the group of directly related components which together performs a specified function, for example 'RPM indication system' would include the RPM indicator, tachometer generator, circuit breaker and associated circuitry.
- 22. <u>"Extended Overwater Flight"</u>: Refers to an operation overwater at a horizontal distance of more than 50 nautical miles from the nearest shoreline.
- 23. Repair Intervals

Calendar Day

A period of 24 hours elapsed time, commencing at midnight on the day of discovery and recording of a malfunction in the aircraft's maintenance record/logbook and ending at midnight on the next day. For example, if it were recorded at 10 am on January 26th that a malfunction had occurred, and the MMEL allowed three calendar days for completion of repairs or replacements, the three day interval would commence at midnight on 26th January and end at midnight on 29th January.

24. <u>"Despatch"</u>: The point at which an aircraft first moves under its own power for the purpose of commencing a flight.

NOTE:

The definition above is in accordance with that given in Article 118(2)(a) of the ANO and it is at the point of despatch that the provisions of the MMEL cease to apply. They come into effect again when the aircraft next comes to rest at the end of its flight. In the case of a helicopter which comes to rest without stopping rotors, it is deemed to have ended its flight and the provisions of the MMEL then apply until it is next despatched.

- 25. <u>NOT USED</u>: An item which appeared in the base document (e.g. FAA MMEL) but which has not been included in the CAA MMEL. The base document item number is retained to maintain continuity.
- 26. Base documents used for the preparation of this MMEL are:
 - (1) FAA L-188 ELECTRA MMEL Revision 14 dated 16 March 1994.
 - (2) CAA Policy Statements, as effective at end September 1996.

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General

- 1. The proviso "The aircraft may continue the flight or series of flights not to exceed 3 calendar days but shall not depart an airport where repairs or replacements can be made" has been replaced by the simpler proviso "Repairs or replacements are carried out within three calendar days".
- 2. In accordance with a recent policy change the "*" has been removed See DEFINITIONS 7.

21 AIR CONDITIONING

21-1	Cabin Compressors	3 day repair policy applied.
21-2	Cabin Pressure Control System NOTE added.	Alleviation clarified 2nd scenario of (1) deleted and
21-3	Cabin Air Compressor Dump Switches	Revised to give alleviation for one or both switches inoperative (M) and (O) added.
21-5	Cabin Air Compressor Disconnect Switch	Revised to give alleviation for one or both switches inoperative (M) added.
21-7	Cabin Altitude Indicator	Proviso (c) added.
21-8	Cabin Differential Pressure Indicator	Proviso (c) added.
21-9	Cabin Rate of Climb Indicator	Proviso (d) added.
21-10	Cabin Air Compressor Warning Lights	(M) added and revised.
21-12	Vapour Cycle (Freon) System	Revised.
21-13	Cooling Turbines	Revised.
21-15	Recirculation Fans	Expanded and (M) added.
21-16	Duct Heaters	Expanded and (M) added.
21-17	Program Position Indicator	Revised i.a.w. FAA MMEL.

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HIGHLIGHTS TO REVISION 2

21	AIR CONDITIONING (Cont)	
21-20	Automatic Temperature Controls	Revised, item title becomes Automatic Temperature Controls.
21-21	Freon Condenser Pressure Indicator	Deleted i.a.w. FAA MMEL.
21-23	Manual Temperature Control Switches.	Deleted i.a.w. FAA MMEL.
21-24	16 Point Temperature Selector and Indicator	Additional Alleviation.
21-27	Temperature Control Panel Advisory Lights	(M) and provisos added.
22	AUTO FLIGHT	
22-3	Autopilot Disconnect Indicator Light	New Item.
23	COMMUNICATIONS	
23-2	Passenger Address System	New proviso (b) added.
23-3	Communication Systems	UHF now "may be inoperative".
23-7	Cabin Attendants Interphone System	New proviso (a) added.
23-9	Radio Rack Cooling Fan	Proviso (b) now 20 minutes and proviso (d) added.
23-10	Crewmember Alerting System	New item, previous item 10 moved to 11.
23-11	Cockpit Voice Recorder	Moved from item 10, previous proviso (c) deleted, 24 hours becomes 48.
23-12	Headsets	Relocated from 11.
NOTE	D : : 10 I I I I	16 4 1000

NOTE: Previous item 12 Loudspeaker has been removed from the MMEL at revision 2, existing item 1 makes "Loudspeaker" superfluous.

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24	ELECTRICAL POWER	
24-1	AC Generator System	Revised, provisos (c) and (d) added, now considers a/c with 3 generators.
24-2	Generator Warning Lights	Revised.
24-11	Generator System Annunciator Lights	Now 4 required i.a.w. FAA MMEL.
24-14	Generator Disengage Switch	Revised.
24-15	Frequency Meter	3 day repair policy applied.
25	EQUIPMENT/FURNISHINGS	
25-1	Megaphones	Latest CAA policy applied.
25-2	Flight Crew Shoulder Harness Inertia	3 day repair policy applied. Reels
26	FIRE PROTECTION	
	Toilet Compartment Smoke Detection	
26-5	Tonet Compartment Smoke Detection	Latest CAA policy applied, third alleviation added. System
26-526-7	APU Fire Detection System	
	•	System
26-7	APU Fire Detection System	System New item.
26-7 26-8	APU Fire Detection System APU Fire Extinguishing System	System New item.
26-7 26-8 <u>27</u>	APU Fire Detection System APU Fire Extinguishing System FLIGHT CONTROLS	System New item. New item.
26-7 26-8 <u>27</u> 27-2	APU Fire Detection System APU Fire Extinguishing System FLIGHT CONTROLS Take-off Warning Horn System	System New item. New item.

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<u>28</u>	FUEL (Cont)	
28-4	Fuel Dump System	Revised, 3 day repair limit applied.
28-5	Fuel Quantity Indicators	Revised and (M) and (O).
28-7	Fuelling Bay Fuel Cap	New item.
28-8	Fuel Crossfeed Valve Position Lights	New item.
29	HYDRAULIC POWER	
29-4	Hydraulic System Pressure Indicators	Revised.
30	ICE AND RAIN PROTECTION	
30-1	Engine Air Scoop and Inlet Vane Anti-ice Valves.	Revised.
30-2	Engine Air Scoop and Inlet Vane Icing Lights	3 day repair limit applied.
30-3	Propeller Anti-Icing	Revised, ICEX alleviation added.
30-4	Wing and Empennage De-icing and Anti-icing	(M) and proviso (a) added.
30-5	Ice Detector	Reference to item 33-6 removed.
30-7	Engine Bleed Air Valve Lights	(M) deleted.
30-10	Leading Edge Temperature Indicator	Revised.
30-11	Pitot Heaters	Revised, now either side may be inoperative, 3 day repair limit applied. Annunciators now in 30-15.
30-12	Electric Windshield Heat	(M) (O) added, 3 day repair policy applied.
30-14	Anti-Ice Manifold Leakage Test System	Revised, 3 day repair limit applied.
30-15	Pitot Heat Annunciator Lights	New item.

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<u>31</u>	INDICATING AND RECORDING SYSTEMS	
31-2	Flight Data Recorder	Proviso (c) deleted, new proviso (c) now 48 hours.
32	LANDING GEAR	
32-3	Emergency Air Brakes	Revised to say "Must be operative".
33	<u>LIGHTS</u>	
33-1	Cockpit and Instrument Lighting Systems	Revised to reflect latest CAA policy.
33-2	Cabin Interior Lighting	Proviso (c) added.
33-3	Passenger Notice System	Revised to reflect latest CAA policy.
33-6	Wing Illumination Lights	Revised to reflect latest CAA policy.
33-7	Landing Lights	NOTE removed, three day repair limit applied.
33-10	Interior Emergency Exit Lighting System	Revised, may be inoperative if no passengers are carried.
33-11	Exterior Emergency Illumination System	Revised, passengers are not carried added.
33-12	Floor Proximity Lighting	Revised, passengers are not carried added.
<u>34</u>	<u>NAVIGATION</u>	
been modif		evised to include consideration of aircraft which have Captain/ Co-Pilots ADI's/ HSI's. Other items have
34-3	Altimeters	Revised i.a.w. FAA MMEL, 3 day repair limit applied.
34-5	Attitude Display Indication	Revised to reflect latest CAA policy.
34-6	Vertical Gyro Systems	Revised to reflect latest CAA policy.

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34	NAVIGATION (Cont)	
34-7	Standby Attitude Indicator	Revised to reflect latest CAA policy.
34-8	Turn and Bank Indicators	Revised to reflect latest CAA policy.
34-9	Horizontal Situation Indication	Revised to reflect latest CAA policy.
34-10	Standby Compass	3 day repair policy applied.
34-23	GPWS	Note added.
34-24	TCAS	New item, aligns CAA MMEL with FAA MMEL.
34-25	Altitude Alerting System	New item.
34-26	Radio Altimeter System	New item.
34-27	Microwave Landing System	New item, aligns with FAA MMEL.
34-28	Transfer Switching System	New item.
<u>35</u>	OXYGEN	
35-1	Passenger Oxygen System	Latest CAA policy applied.
35-3	Portable Oxygen Dispensing Units	NOTE added.
35-4	Protective Breathing Equipment	New item.
49	AIRBORNE AUXILIARY POWER	
49-2	APU Generator	New item
49-3	APU Bleed Air	New Item
49-4	APU Caution and Warning Lights	New Item

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52	<u>DOORS</u>	
52-1	Forward Air Stair	Revised, provisos a) and b) added.
52-2	Door Warning Light System	NOTE added.
52-3	Main Cabin Cargo Door Warning Light	New item, previous item 3 moved to 52-4. System
52-4	Lockable Flight Deck Door	3 day repair policy applied.
52-5	Main Cabin Cargo Door	New item.
61	PROPELLERS	
61-4	Autofeather-TSS System	(M), (O) added 3 day repair policy applied.
61-8	Low Oil Level Warning Light	3 day repair policy applied.
73	ENGINE FUEL & CONTROL	
73-1	Temperature Datum System	Relocated from ATA 77, (M) added.
73-2	Fuel Flow Indicators	Relocated from ATA 77, 3 day repair policy applied.
74	<u>IGNITION</u>	
74-1	Speed Sensitive Control (Switch)	New item.

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77	ENGINE INDICATORS	
77-1	TIT Indicators	Revised i.a.w. FAA source, alleviation now for digital display only.
77-2	Torque Indicating System	Deleted.
77-3	Temperature Datum System	Moved to ATA 73.
77-4	Fuel Flow Indicators	Moved to ATA 73.
79	ENGINE OIL	
79-1	Engine Oil Pressure Indicators	Deleted i.a.w. FAA revision 14.
79-2	Engine Oil Pressure Indicators	Deleted i.a.w. FAA revision 14.
79-3	Engine Oil pressure Warning Lights	Revised, 3 day repair policy applied.
79-4	Engine Oil Temperature Indicator	Deleted i.a.w. FAA revision 14.
79-6	Oil Cooler Flap Control System	Revised, alleviation for auto system, where fitted.
79-7	Oil Quantity Indicators	Revised and 3 day repair policy applied.
79-8	Oil Cooler Inducers	Revised and 3 day repair policy applied.

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		Item		(3) Nui	mber required for despatch	
				, ,	(4) Remarks or Exceptions	
					(4) Normania di Exceptiona	
24	ΛID	CONDITIONING				
<u>21</u>	AIR	CONDITIONING				
1.	Cabin	Compressors	2	1	(O) One may be inoperative provided:	
					(a) Flight altitude is limited to FL 15	0, and
					(b) Sufficient oxygen is carried which continuous use for the duration of FL 100.	
					OR	
			2	0	(O) (M) Both may be inoperative for unproprovided:	ressurised flight
					(a) Flight altitude is limited to FL100),
					(b) Outflow valve is secured open, ar	nd
					(c) Repairs or replacements are carrie three calendar days.	ed out within
2.	Cabin	Pressure Control System				
	(1)	Automatic (Pneumatic)	1	0	(O) May be inoperative provided:	
					(a) Manual pressurisation is operatin	g normally, and
					(b) Cabin altitude, rate and differenti operating normally.	al indicators are
	(2)	Manual (Electric)	1	0	(O)(M) May be inoperative for unpressuring provided operations are conducted in according to the fight Manual limitations.	
	(3)	Automatic and Manual	2	0	Both may be inoperative provided:	
					(a) The aircraft is flown unpressurise operations are conducted in accor Flight Manual Limitations, and	
					Cont	

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		Item		(3) Nur	mher require	ed for despatch	
			1	(3) Nui			
					(4) Remai	rks or Exceptions	
<u>21</u>		ONDITIONING					
	(Cont	<u>)</u>					
2.	Cabin F (Cont	Pressure Control System)					
	(3)	Automatic and Manual (Cont)					
		(Cont)			(b)	Flight altitude is limited to FL100	
					NOTE	If the outflow valve has been secu cannot be closed extended overwaprohibited.	
3.		Air Compressor Dump	2	1	(M) (O)) One may be inoperative provided:	
	Switche	es			(a)	Remaining Cabin Air Compressor is operative, and	Dump Switch
					(b)	Cabin Air Compressor associated Dump Switch is operative.	with operative
			2	0) May be inoperative provided flightepted unpressurised configuration.	t is conducted in
4.	Safety '	Valve Control	1	1	Must be	e operative.	
5.	Cabin A Switch	Air Compressor Disconnect	2	1	(M) On compre	ne may be inoperative provided assorssor is verified disconnected and is	ciated not used.
			2	0	(M) Ma	ay be inoperative provided:	
					(a)	Both compressors are verified disc	connected, and
					(b)	Flight is conducted in an accepted configuration.	unpressurised
						<u></u>	

AIR	CRAFT:	VED A		REVISION	ON NO: REVISION 2	PAGE:
	LOCKHEED L-188 ELEC	TRA		DATE:	25 OCTOBER 1996	21-3
(1) Sys	stem & Sequence Numbers	(2) Numl	ber Installe	ed		
	Item		(3) Nur	mber require	ed for despatch	
				(4) Rema	rks or Exceptions	
21	AIR CONDITIONING					
	(Cont)					
6.	Cabin Compressor Indicator (Compressor Inlet/ Discharge Pressure)	2	-	May be	e inoperative for an inoperative cabi	n compressor.
7.	Cabin Altitude Indicator	1	0	(O) Ma	by be inoperative provided:	
					Cabin differential pressure indicate	tor is aparating
				(a)	normally, and	or is operating
				(b)	A chart is provided to crew to corpressure to cabin altitude.	vert differential
					OR	
				(c)	Flight is conducted in an accepted configuration.	unpressurised
8.	Cabin Differential Pressure Indicator	1	0	(O) Ma	by be inoperative provided:	
				(a)	Cabin altitude indicator is operation	ng normally, and
				(b)	A chart is provided to crew to coraltitude to differential pressure.	vert cabin
					OR	
				(c)	Flight is conducted in an accepted configuration.	unpressurised
9.	Cabin Rate of Climb Indicator	1	0	(O) Ma	y be inoperative provided:	
				(a)	Cabin altitude indicator operates i	normally, and
				(b)	Cabin differential pressure indicat normally, and	or operates
					-	
				Cont		
		I	I	I		

AIRCRAFT:				REVISION NO: REVISION 2 PAGE:				
	LOCKHEED L-188 ELEC	CIRA		DATE:	25 OCTOBER 1996	21-4		
(1) Sys	tem & Sequence Numbers	(2) Numb	er Installe					
	Item		(3) Nur	mher require	ed for despatch			
		1	(5) 1401					
				(4) Remai	rks or Exceptions			
<u>21</u>	AIR CONDITIONING (Cont)							
	<u>(0011:)</u>							
9.	Cabin Rate of Climb Indicator							
<i>)</i> .	(Cont)							
				(c)	Pressurisation system operates nor automatic and manual modes.	mally in		
					OR			
				(d)	Flight is conducted in an accepted configuration.	unpressurised		
10.	Cabin Air Compressor Warning	4		May ba	e inoperative provided associated con	mproggar ig		
10.	Lights (Low Oil Pressure/High	4	-		nected and secured.	iipiessoi is		
	Oil Temp)							
11.	Cabin Altitude Warning System	1	0	(O) Ma	y be inoperative for flights below 10	0,000 feet MSL.		
	<i>3</i>				, ,	,		
12.	Vapour Cycle (Freon System)	1	0	(O) Ma	y be inoperative provided:			
				(a)	Departure, enroute and terminal ai	rnort		
					temperatures are not forecast to ex (80° F).			
					OR			
				(b)	Above 27° C (80° F) if the Freon	System is		
					inoperative, both Cooling Turbine	s are operative.		
13.	Cooling Turbines	2	1	One ma	ay be inoperative provided:			
	C					,		
				(a)	Departure, enroute and terminal ai temperatures are not forecast to ex (80° F).	rport ceed 27° C		
					OR			
				(b)	Freon System is operative.			

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	LOCKHEED L-188 ELEC	DATE: 25 OCTOBER 1996 21						
(1) System & Sequence Numbers (2			ber Installe	ed				
	Item	(3) Number required for despatch						
		1	(3) Nui					
				(4) Remarks or Exceptions				
<u>21</u>	AIR CONDITIONING							
	(Cont)							
14.	Freon Condenser Door (If Installed)	1	0	(M)(O) Door and/or actuator may be inope	rative provided:			
				(a) Door is secured in FLIGHT OPEN	N position,			
				(b) Circuit breaker in centre section lo pulled and tagged,	oad centre is			
				(c) Freon condenser (if installed) disc and temperature are monitored du operations and throughout the dur flight, and	ring ground			
				(d) Operations are conducted in accor Flight Manual limitations.	dance with			
15.	Recirculation Fans (If Installed)	-	0	(M) May be inoperative provided system is accepted procedure.				
16.	Duct Heaters	-	0	(M) May be inoperative provided system is accepted procedure.	secured by an			
17.	Program Position Indicator (If Installed)	1	0	(M) May be inoperative provided:				
	`			(a) 16 Point Selector is operative.				
				OR				
				(b) Associated system is secured by a method.	n approved			
18.	Cabin Temperature Indicator (If Installed)	1	0	May be inoperative.				

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		LOCKHEED L-188 ELEC	IKA		DATE: 25 OCTOBER 1996	21-6
(1) Sys	stem & Sec	uence Numbers	(2) Numl	ber Installe	ed	-
		Item		(3) Nun	nber required for despatch	
					(4) Remarks or Exceptions	
					(4) Remarks of Exceptions	
<u>21</u>	AIR (CONDITIONING t)				
19.		Condenser Door Position or (If Installed)	1	0	(M)(O) May be inoperative provided:	
	marcut	or (ir instance)			(a) Door is secured in FLIGHT OPEN p	osition,
					(b) Circuit breaker in centre section load pulled and tagged, and	d centre is
					(c) Freon condenser discharge pressure temperature are monitored during groperations and throughout the duratiflight.	ound
20.		atic Temperature Controls and Flight Station) alled)	2	0	(O) May be inoperative provided Manual Cor are operative.	ntrols
21.	Freon (If Inst	Condenser Pressure Indicator alled)			Deleted - Revision 2.	
22.		Compartment Overheat Lights rd and Rear Baggage Bays)				
	(1)	Aircraft with Cargo Compartment Heating	2	0	(M) Both may be inoperative provided:	
		Compartment Heating			(a) Cargo heater switch is selected "OFI	F", and
					(b) Circuit breakers for cargo compartm control are pulled and tagged.	ent heat
	(2)	Aircraft Without Cargo Compartment Heating	2	0	Both may be inoperative.	

AIRCRAFT:			REVISION NO: REVISION 2	PAGE:		
LOCKHEED L-188 ELEC	DATE: 25 OCTOBER 1996					
stem & Sequence Numbers	(2) Num	ber Installe	ed			
Item		(3) Nur	mber required for despatch			
	1					
			(4) Nemarks of Exceptions			
AID CONDITIONING						
<u>, commy</u>						
Manual Temperature Control Switches			Deleted - Revision 2.			
16 Point Temperature Selector and						
Indicator						
(1) Cabin Evap Freon IN/OUT (If Installed)	1	0	May be inoperative.			
(2) Flt. Sta. Evap Freon IN/OUT (If Installed)	1	0	May be inoperative.			
(3) Air Cycle IN/OUT Left/Right	2	0	May be inoperative provided associated Comp Inlet/Discharge Indicator is operative.	pressor		
(4) Cabin Heater/Evap. Air	1	0	May be inoperative provided:			
			(a) Associated Overheat Light is operative	ve, and		
			(b) Cabin Temperature Indicator is opera	tive.		
(5) Flight Station Heater/Evap. Air	1	0	May be inoperative provided Associated Over operative.	heat Light is		
(6) Cargo Compartment Heat Aft/Forward	1	0	(O) May be inoperative provided Cargo Heat	is not used.		
(7) Cond. Freon/Cond. Fan Motor	2	0	May be inoperative.			
	AIR CONDITIONING (Cont) Manual Temperature Control Switches 16 Point Temperature Selector and Indicator (1) Cabin Evap Freon IN/OUT (If Installed) (2) Flt. Sta. Evap Freon IN/OUT (If Installed) (3) Air Cycle IN/OUT Left/Right (4) Cabin Heater/Evap. Air (5) Flight Station Heater/Evap. Air	AIR CONDITIONING (Cont) Manual Temperature Control Switches 16 Point Temperature Selector and Indicator (1) Cabin Evap Freon IN/OUT (If Installed) (2) Flt. Sta. Evap Freon IN/OUT (If Installed) (3) Air Cycle IN/OUT Left/Right (4) Cabin Heater/Evap. Air 1 (5) Flight Station Heater/Evap. Air 1 (6) Cargo Compartment Heat Aft/Forward	AIR CONDITIONING (Cont) Manual Temperature Control Switches 16 Point Temperature Selector and Indicator (1) Cabin Evap Freon IN/OUT (If Installed) (2) Flt. Sta. Evap Freon IN/OUT (If Installed) (3) Air Cycle IN/OUT	Item Sequence Numbers Item (2) Number Installed (3) Number required for despatch (4) Remarks or Exceptions AIR CONDITIONING (Cont) Manual Temperature Control Switches Deleted - Revision 2. Deleted - Revision 2. Deleted - Revision 2. May be inoperative. (1) Cabin Evap Freon IN/OUT (If Installed) (2) Fit. Sta. Evap Freon IN/OUT (If Installed) (3) Air Cycle IN/OUT Left/Right (4) Cabin Heater/Evap. Air 1 0 May be inoperative provided associated Complinet/Discharge Indicator is operative. (4) Cabin Heater/Evap. Air 1 0 May be inoperative provided: (a) Associated Overheat Light is operative (b) Cabin Temperature Indicator is operative. (5) Flight Station Heater/Evap. Air 1 0 May be inoperative provided Associated Over operative.		

AIRCRAFT:				REVISION NO: REVISION 2 PAGE:
	LOCKHEED L-188 ELE	CTRA		DATE : 25 OCTOBER 1996 21-8
(1) Sys	(1) System & Sequence Numbers (2) Number Ins			lled
Item (3)			(3) Nur	umber required for despatch
			(-)	(4) Remarks or Exceptions
				(4) Nemarks of Exceptions
<u>21</u>	AIR CONDITIONING (Cont)			
25.	Freon Compressor Inlet Pressure Indicator (If Installed)	1	0	May be inoperative provided:
				(a) Freon condenser pressure indicator operates normally, and
				(b) Freon condenser door and position indicator operates normally.
26.	Rapid Heat Doors and Controls (If Installed)	-	0	May be inoperative.
27.	Temperature Control Panel Advisory Lights	6	0	(M) May be inoperative provided:
	Turnot Jugate			(a) 16 Point Selector is operative.
				OR
				(b) Associated system is secured.

AIRCRAFT:				REVISIO	ON NO:	REVISION 2	PAGE:
	LOCKHEED L-188 ELEC	CTRA		DATE:	25 OCT	OBER 1996	22-1
(1) Sys	tem & Sequence Numbers	(2) Numb	er Install			02211770	
	Item		(3) Nu	mber require	ed for desna		
		1	(5) 140				
				(4) Remai	rks or Excep	otions	
22	AUTO FLIGHT						
1.	Automatic Pilot System	1	0			erative except components ass sed must be operative.	sociated with
2.	Control Wheel Autopilot Disengage Switch	2	1	(O) One (a) (b) (c)	Aircraft The dise	noperative provided: is flown from the non-affecte ngage switch on the non affect to be operating normally, and opilot is not used below 1,500	cted side is
		2	1	(O) One (a) (b) (c)	Both go- The dise verified	noperative provided: -around buttons are operative, ngage switch on the non-affe to be operating normally, and opilot is not to be used below	cted side is
3.	Autopilot Disconnect Indicator Light	1	0	May be	e inoperati	ve provided autopilot is not u	sed.

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	LOCKHEED L-188 ELEC	JIRA		DATE: 25 OCTOBER 1996	23-1			
(1) Sys	stem & Sequence Numbers	(2) Number	(2) Number Installed					
	Item		(3) Number required for despatch					
				(4) Remarks or Exceptions				
23	COMMUNICATIONS							
1.	Flight Deck Speakers (If Installed)	-	0	May be inoperative for communications purpo each crew member has an operative headset.	ses provided			
2.	Passenger Address System (If Installed)	1	0	As required by Air Navigation Legislation. M inoperative provided: (a) Cabin interphone system is operative, (b) Chime system is operative, (c) Alternate normal and emergency procestablished and utilised, and (d) Aircraft may continue the flight or sebut shall not depart an airport where the made and shall not exceed 25 flight to completion of repairs.	cedures are ries of flights repairs can			
3.	Communication Systems (1) VHF System (2) HF System (3) UHF System		-	As required by Air Navigation Legislation. As required by Air Navigation Legislation. May be inoperative.				
4.	Cockpit Interphone System	1	1	Required for all crewmembers on flight deck of	luty.			
5.	Audio Selector Panels	-	3	One required for each crew member on flight of	deck duty.			
6.	Service Interphone System	1	0	As required by Air Navigation Legislation. M inoperative provided procedures do not dependent or alternate normal and emergency procedures established and utilised.	d on its use			

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	LOCKHEED L-188 ELEC	IKA		DATE: 25 OCTOBER 1996	23-2
(1) Sy	stem & Sequence Numbers	(2) Num	ber Installe		
	Item		(2) Ni	and an approximately also makely	
		1	(3) Nur	mber required for despatch	
				(4) Remarks or Exceptions	
<u>23</u>	COMMUNICATIONS				
	(Cont)				
7.	Cabin Attendants Interphone System	1	0	(O) May be inoperative provided:	
				(a) Procedures do not require its use.	
				OR	
				(b) Passenger address system is operati	ve, and
				(c) Alternate normal and emergency pr established and utilised.	ocedures are
8.	Selective Call System (SELCAL)	1	0	May be inoperative provided:	
	(If Installed)			(a) Procedures do not require its use,	
				OR	
				(b) Flight crew monitor appropriate HF frequency.	/VHF
9.	Radio Rack Cooling Fan	1	0	Fan may be inoperative provided:	
				(a) Cockpit ambient temperature does r 38°C (100°F),	ot exceed
				(b) Ground operating time does not exc minutes,	eed 30
				(c) The rack cover is removed and store the radio rack, and	ed away from
				(d) The aircraft is not operated unpress	urised.

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	LOCKHEED L-100 ELE	CIKA	DATE: 25 OCTOBER 1996				
(1) Sy	stem & Sequence Numbers Item	(2) Number		nber required for despatch (4) Remarks or Exceptions			
23	COMMUNICATIONS (Cont)						
10.	Crewmember Alerting System (Crew Call Light and Chime)						
	(1) Flight Crew Call Light	1	0	May be inoperative.			
	(2) Flight Attendant Call Light System	1	0	May be inoperative provided: (a) Passenger Address System is operative provided: (b) Flight Attendant Call Chime is operative provided: (c) Affected Light is not required for 1 Smoke Detector Alerting.	erative, and		
	(3) Flight Attendant Call Chime System						
	(1) Passenger Configuration	1	0	 (O) May be inoperative provided: (a) Passenger Address System is operation (b) Flight Attendant Call Light is operation (c) Affected Chime is not required for Smoke Detector Alerting, and (d) Alternative Normal and Emergence are established and used. 	rative, Lavatory		
	(2) Cargo Configuration	1	0	May be inoperative.			

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	LOCKHEED L-188 ELE	CTRA		DATE: 25 OCTOBER 1996	23-4		
(1) Sys	stem & Sequence Numbers	(2) Numb	er Install				
	Item	Т	(3) Nu	mber required for despatch			
		7	` '	(4) Remarks or Exceptions			
				(4) Normality of Exceptions			
<u>23</u>	COMMUNICATIONS						
<u> </u>	(Cont)						
11.	Cockpit Voice Recorder System (CVR)	1	0	As required by Air Navigation Legislation. I inoperative provided:	May be		
				(a) It is not reasonably practical to repa before commencement of the flight.	r or replace		
				(b) The aircraft shall not exceed six (6) flights with the CVR unserviceable with the first flight after the CVR w operating throughout the flight.	oeginning		
				(c) Not more than 48 hours have elapse CVR became unserviceable.	d since the		
				(d) The aircraft must not depart from its base with the CVR unserviceable.	maintenance		
				(e) The Flight Data Recorder must be o normally.	perating		
12.	Headsets	-	-	One headset (including boom microphone) moperative for each required crew member on			
				duty.			

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	LOCKHEED L-188 ELEC	TKA		DATE: 25 OCTOBER 1996 24-1					
(1) Sys	tem & Sequence Numbers	(2) Numb	er Installe	ed	_				
	Item	Г	(3) Number required for despatch						
			, ,	(4) Remarks or Exceptions					
				(,, , , , , , , , , , , , , , , , , , ,					
<u>24</u>	ELECTRICAL POWER								
1.	AC Generator Systems								
	(1) Aircraft fitted with four generators	4	3	(M) One generator may be inoperative provided:					
				(a) The cause of the malfunction is determined,					
				(b) Appropriate action is taken to ensure that no hazard exists.					
				(c) When generator 4 is inoperative do not operate engines with RPM Selector Switches in Low, an	nd				
				(d) Repairs or replacements are carried out within three calendar days.					
	(2) Aircraft fitted with three generators	3	2	(M) One may be inoperative provided:					
	Sanaranoro			(a) The cause of the malfunction is determined,					
				(b) Appropriate action is taken to ensure that no hazard exits, and					
				(c) Repairs or replacements are carried out after one flight.	e				
2.	Generator Warning Lights								
	(1) Trip Lights, Disengage Lights	-	-	(M) May be inoperative provided:					
				(a) Associated generator is inoperative, and					
				(b) Associated Mechanical Failure Light is operative	e.				
	(2) Mechanical Failure Lights	4	3	(M) One may be inoperative provided associated generate is removed or disengaged and secured by an acceptable procedure.	or				

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LOCKHEED L-188 ELECT		ZIKA		24-2	
(1) System & Sequence Numbers		(2) Numl	ber Installe	DATE: 25 OCTOBER 1996 ed	
Item			(3) Number required for despatch		
		1	(3) Null		
				(4) Remarks or Exceptions	
24	ELECTRICAL POWER				
	(Cont)				
3.	Transformer Rectifier Unit	2	2	Both must be operative.	
4	TRU Cooling Fans	2	2	Doth must be energive	
4.	TRO Cooling Fails	2	2	Both must be operative.	
5.	Emergency Inverter	1	1	Must be operative.	
6.	AC Volt Meter	1	1	Must be operative.	
7.	AC Amp Meter		_	One may be inoperative for an inoperative ge	nerator
7.	AC Amp Meter	_	_	One may be moperative for an moperative ge	iciator.
8.	DC Amp Meter	1	1	Must be operative.	
9.	DC Volt Meter	1	1	Must be operative.	
10.	Battery System	1	1	Must be operative.	
				-	
11.	Generator System Annunciator	-	-	Must be operative.	
	Lights				
12.	External Power (Ground System)	1	0	(M)(O) May be inoperative.	
13.	Electrical Load Recirculating Fan	2	2	Both must be operative.	
		I	I	I	

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LOCKHEED L-188 ELECTRA				DATE: 25 OCTOBER 1996	24-3		
(1) System & Sequence Numbers (2) Number In				ed	- 		
	Item	Т	(3) Number required for despatch				
				(4) Remarks or Exceptions			
24	ELECTRICAL DOWER						
<u>24</u>	ELECTRICAL POWER (Cont)						
	<u>(1991)</u>						
14.	Generator Disengage Switch(s)	-	0	(M) May be inoperative provided:			
				(a) All Generator Warning Lights are	operative, and		
				(b) Repairs or replacements are carried three calendar days.	out within		
				OR			
				(c) Associated Generator is disengage and secured by approved means, as			
				(d) Repairs or replacements are carried three calendar days.	out within		
15.	Frequency Meter	1	0	May be inoperative provide repairs or repla carried out within three calendar days.	ements are		

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(1) System & Sequence Numbers			(2) Number Installed				
	Item	(3) Number required for despatch					
		(4) Remarks or Exceptions					
25	EQUIPMENT/FURNISHINGS						
1.	Megaphones	-	-	As required by Air Navigation Legislation. of those required by Air Navigation Legisla inoperative or missing provided the inoperamegaphone is removed from the cabin.	tion may be		
2.	Flight Crew Shoulder Harness Inertia Reels			As required by Air Navigation Legislation.			
	(1) Inertia Reels	-	-	(M) May be inoperative provided:			
				(a) The affected harness is adjusted an approved means to suit the requirer individual flight crew member, and	ments of		
				(b) Repairs or replacements are carried three calendar days.	l out within		
3.	Flight Deck Observer Seat and Harness	1	0	May be inoperative provided the seat is not is correctly stowed.	required and		
4.	Flight Attendants Seats and	_	-	(M)(O) As required by Air Navigation Legi	slation.		
	Harnesses (If Installed)			Any in excess of those required by legislation inoperative (see notes below).	on may be		
				NOTE 1 A folding seat that will not stow au remain stowed is considered to be shall be secured in the retracted por removed.	noperative and		
				NOTE 2 A seat with a defective harness is c inoperative and shall be placarded occupancy.			

All	RCRAFT:			REVISION NO: REVISION 2	PAGE:
	LOCKHEED L-188 ELEC	TRA		DATE: 25 OCTOBER 1996	25-2
(1) Sy	stem & Sequence Numbers	(2) Numl	per Installe	ed	1
	Item		(3) Nun	nber required for despatch	
				(4) Remarks or Exceptions	
<u> 25</u>	EQUIPMENT/FURNISHINGS				
	(Cont)				
5.	Flight Crew Smoke Protection Equipment (Basic and Portable)	-	-	As required by Air Navigation Legislation. In specified items may be missing or inoperative accordance with arrangements approved by the	in
6.	Cabin Attendants Portable Smoke Protection Equipment (If Installed)	-	-	As required by Air Navigation Legislation. In specified items may be missing or inoperative accordance with arrangements approved by the	in
7.	Torches	-	-	As required by Air Navigation Legislation.	
8.	Floatation Devices (Lifejackets and Liferafts)	-	-	As required by Air Navigation Legislation.	

AIRCRAFT: LOCKHEED L-188 ELEC		CTD A		REVISION NO: REVISION 2	PAGE:
	LUCKHEED L-188 ELE	CIKA		DATE: 25 OCTOBER 1996	26-1
(1) Sy	stem & Sequence Numbers Item	(2) Numl	ber Installe	ed	
	item		(3) Nur	mber required for despatch	
				(4) Remarks or Exceptions	
<u>26</u>	FIRE PROTECTION				
1.	Power Plant Fire Extinguisher System	2	2	All must be operative.	
2.	Engine Fire Detection System	12	12	All zones must be operative.	
3.	Portable Fire Extinguishers (Hand Held)	-	-	As required by Airworthiness Notice No 60.	
				Extinguishers in excess of the minimum requinoperative.	ired may be
4.	Fire Extinguisher Discharge Discs	4	0	Any or all may be inoperative or missing progauge pressure is verified by visual means to prior to each departure.	
5.	Toilet Compartment Smoke Detection System (If Installed)	-	-	(M) May be inoperative. The toilet compartre be electrically isolated, the waste-bin must be the toilet compartment(s) must be locked and placarded.	e emptied and
				OR	
		-	-	(O) May be inoperative provided:	
				(a) The toilet compartment is checked a minute intervals for evidence of fire	
				(b) Repairs or replacements are carried three calendar days.	out within
				OR	
				Cont	

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	LOCKHEED L-188 ELEC	CTRA		DATE: 25 OCTOBER 1996 26-2
(1) Sys	tem & Sequence Numbers	(2) Number	Installe	
	Item		(2) Num	mhor required for despetab
		1 '	(3) INUI	mber required for despatch
				(4) Remarks or Exceptions
<u> 26</u>	FIRE PROTECTION			
	(Cont)			
5.	Toilet Compartment Smoke Detection System (If Installed) (Cont)			
				(O) May be inoperative provided:
				(a) Toilet compartment Fire Extinguishers are fitted and operating normally.
				(b) The toilet compartment is checked at regular and frequent intervals for evidence of fire and smoke, and
				(c) Repairs or replacements are carried out within three calendar days
				OR
		-	-	(M) May be inoperative, missing or deactivated provided the total number of passengers the aircraft is certificated for does not exceed that specified in Airworthiness Notice No 60.
6.	Toilet Compartment Fire Protection Equipment (If Installed)	-	-	May be inoperative.
7.	APU Fire Detection System (If Installed)	1	0	May be inoperative provided APU is not used.
8.	APU Fire Extinguishing System (If Installed)	1	0	May be inoperative provided APU is not used.

AIR	CRAFT:	CEP 4		REVISION NO: REVISION 2	PAGE:
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(1) Sys	stem & Sequence Numbers	(2) Numb	per Installe		
	Item		(2) Nun	nber required for despatch	
		7	(5) Null		
				(4) Remarks or Exceptions	
<u>27</u>	FLIGHT CONTROLS				
1.	Wing Flap Position Indicator	1	1	Must be operative.	
2.	Take-off Warning Horn System (If Installed)	1	0	(O) May be inoperative provided flap selection before take-off.	n is verified

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(1) Sy	stem & Sequence Numbers	(2) Numb	er Instal	led
	Item	_	(3) Nu	umber required for despatch
				(4) Remarks or Exceptions
28	FUEL			
1.	Fuel Tank Boost Pumps	4	3	(O) One may be inoperative provided:
				(a) Aircraft is operated at or below 18,000 feet MSL, and
				(b) Repairs or replacements are carried out within three calendar days.
2.	Fuel Tank Sump Pumps			
	(1) Fuel/Hydraulic Fluid Heat	4	2	(O) The Outboard pumps may be inoperative provided:
	Exchanger Type Hydraulic System			(a) The associated fuel tank is placarded to indicate that unusable fuel is increased to 232 IMP gallons (1,863 pounds),
				(b) Crossfeed System is operative, and
				(c) Repairs or replacements are carried out within three calendar days.
	(2) Single Tube Fluid Cooler Type Hydraulic System (Snake)	4	0	May be inoperative provided fuel tanks are placarded to indicate that unusable fuel is increased to:
				(a) Outboards 232 IMP gallons (1,863 pounds) unusable,
				(b) Inboards 430 IMP gallons (3,458 pounds) unusable, and
				(c) Repairs or replacements are carried out within three calendar days.
3.	Fuel Temperature Indicator (If Installed)	1	0	(O) May be inoperative provided outside air temperature is monitored.

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	Item		(3) Nur	nber required for despatch	
				(4) Remarks or Exceptions	
<u> 28</u>	FUEL (Cont)				
4.	Fuel Dump System	1	0	(O) May be inoperative provided:	
				(a) Take off weight does not exce maximum landing weight,	ed 105% of
				(b) All Fuel Dump Valves are ver	ified closed,
				(c) Performance is not predicated for enroute engine(s) out process.	
				(d) AFM Limitations and Procedu	ares are followed,
				(e) Repairs or replacements are cathree calendar days.	arried out within
5.	Fuel Quantity Indicators (Flight Station)	4	3	(M) (O) One may be inoperative provide	led:
	(Fight Station)			(a) Fuel quantity is determined by means after each refuelling, ar	
				(b) All Fuel Flow Meters are oper	rative, and
				(c) Fuel consumed is recorded.	
6.	Fuel Quantity Indicators (Under Wing)	4	0	(M) May be inoperative provided the fu verified by an approved procedure usin and after refuelling.	

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Item		(3) Nur	mber require	ed for despatch	
			(4) Rema	rks or Exceptions	
28 FUEL (Cont)					
7. Fuelling Bay Fuel Cap	1	0	(M) Ma	ay be missing provided:	
			(a)	Refuelling receptacle is verified to contamination prior to each refue	to be free of elling,
			(b)	Receptacle is verified to have no afterwards,	leakage
			(c)	All Fuel Tank Quantity Indicator and	s are operative,
			(d)	Repairs or replacements are carri	ed out.
				•	

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	Item] [(3) Nu	mber required for despatch
				(4) Remarks or Exceptions
<u>29</u>	HYDRAULIC			
1.	Hydraulic Pumps (AC)			
	(1) Aircraft with Less Than Four Normal Pumps	-	-	All must be operative.
	(2) Aircraft with Four Normal Pumps	4	3	(M) One normal hydraulic pump may be inoperative provided:
				(a) Spare pump is connected to the same system as the inoperative pump using an approved procedure, and
				(b) Pump switch is placarded accordingly.
2.	Battery Powered DC Pump (Tow Pump)	1	0	May be inoperative provided ground operations are not dependant on its use.
3.	Overheat Warning Lights	2	2	Both must be operative.
4.	Hydraulic System Pressure Indicators	s 4	2	(M) Two may be inoperative provided:
				(a) Hydraulic Low Pressure Warning Lights are operative, and
				(b) Either No. 1 System or Normal Brake Pressure Indications are operative.
				OR
				(c) Emergency Brake Pressure or No. 2 System Pressure Indications are operative.
5.	Hydraulic Low Pressure Warning Lights	3	0	(M) May be inoperative provided respective system pressure gauge is operating normally.
5.		3	0	

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	LOCKHEED L-188 ELEC	CIKA		DATE:	25 OCTOBER 1996	30-1	
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	Item		(3) Nur	mber require	ed for despatch		
				(4) Remar	ks or Exceptions		
<u>30</u>	ICE AND RAIN PROTECTION						
1.	Engine Air Scoop and Inlet Vane Anti-ice Systems	4	3		One valve on one engine may be in provided:	noperative	
				(a)	Valve is verified to be in the close blanked off,	ed position or	
				(b)	Aircraft is not operated in known conditions,	or forecast icing	
				(c)	Remaining three Engine Anti-Ice operative,	Systems are	
				(c)	Associated Engine Air Scoop and Icing Light is operative, and	Inlet Vane	
				(d)	Repairs or replacements are carrie three calendar days.	ed out within	
2.	Engine Air Scoop and Inlet Vane Icing Lights	4	2	(M) Tw	o may be inoperative provided:		
	icing Lights			(a)	Associated Fuel Flow Indicators,		
				(b)	Associated Horsepower Indicators and	s are operative,	
				(c)	Repairs or replacements are carrie three calendar days.	ed out within	
3.	Propeller Anti-icing (Including Spinner and Islands)	4	0	(M)(O)	May be inoperative provided:		
	Spinner and Islands)			(a)	The AFM approved "ICEX" anti-is used.	icing procedure	
					OR		
				(b)	The aircraft is not operated in kno icing.	wn or forecast	

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(1) Sy	stem & Sequence Numbers	(2) Num	ber Install	ed
	Item		(3) Nu	mber required for despatch
				(4) Remarks or Exceptions
				(,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
30	ICE AND RAIN			
<u> </u>	PROTECTION (Cont)			
4.	Wing and Empennage De-icing	4	0	(M)(O) May be inoperative provided:
	and Anti-icing			(a) Valves are verified in the CLOSED position, and
				(b) Aircraft is not operated in known or forecast
				icing.
5.	Ice Detector (If Installed)	1	0	(O) May be inoperative.
6.	Engine Bleed Air Shut-off Valve	4	4	All must be operative.
	Ü			
7.	Engine Bleed Air Valve Lights	4	3	One may be inoperative provided manifold pressure gauge is operative.
8.	Airfoil Overheat Lights	4	4	All must be operative.
9.	Centre Annunciator Panel Overheat	1	1	Must be operative.
	Light			
10.	Leading Edge Temperature Indicator	1	0	(M)(O) May be inoperative provided:
				(a) All other components of the Wing and
				Empennage De-ice and Anti-ice System are
				operative, and
				(b) Alternate procedures are established and used.
				OR
				(c) Aircraft is not operated in known or forecast icing
				conditions.

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	ltem		(3) Nu	mber required for despatch	
				(4) Remarks or Exceptions	
<u>30</u>	ICE AND RAIN PROTECTION (Cont)				
11.	Pitot Heaters	2	1	One may be inoperative provided:	
				(a) Flights are restricted to da	y VMC only,
				(b) The aircraft is not operate icing conditions, and	d into known or forecast
				(c) Repairs or replacements a three calendar days.	re carried out within
12. Electric Windshield Heat (Not Applicable to Aft Stationary	(Not Applicable to Aft Stationary	1	0	(M)(O) May be inoperative provid	ed:
	Panels)			(a) The aircraft is placarded i Lockheed Service letter F January 1961,	
				(b) Windshield heat is selected breaker pulled and tagged	
				(c) The aircraft is not operate icing conditions,	d into known or forecast
				(d) Maximum speed is limited altitudes below 10,000 fee	
				(e) Windshield defog fan ope	erates normally, and
				(f) Repairs or replacements a three calendar days.	re carried out within
				NOTE Turn on windshield defog descent. In extremely hur defogging is accomplishe station temperature. Simuthe flight station freon system dehumidifier provides addefogging protection.	mid conditions, further d by raising the flight altaneous operation of stem (if installed) as a

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	Item		(3) Nu	mber required for despatch
				(4) Remarks or Exceptions
<u>30</u>	ICE AND RAIN PROTECTION (Cont)			
13.	Windshield Wipers	2	0	One or both may be inoperative provided aircraft is not operated in precipitation within arrival and departure areas
14.	Anti-ice Manifold Leakage Test System (Accept Leak Light)	1	0	 (M) May be in operative provided: (a) System is verified by checking of the manifold gauge that the leakage decay time is within acceptable limits (from 24.5 PSI to 14.5 PSI in 8 or more seconds), and (b) Repairs or replacements are carried out within three calendar days.
15.	Pitot Heat Annunciator Lights	2	1	 (M) One may be inoperative provided: (a) Before each departure both heaters are verified operative or the flight is restricted to day VMC, (b) Aircraft is not operated in known or forecast icing, and (c) Repairs or replacements are carried out within three calendar days.

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		_	(3) Nur	mber required	for despatch	
				(4) Remark	s or Exceptions	
<u>31</u>	INDICATING AND RECORDING SYSTEMS					
1.	Clock	-	0	inoperati	equired by Air Navigation Legisla ve provided an accurate timepiece deck indicating the time in hours	is available on
2.	Flight Recorder	1	0	inoperati (a)	red by Air Navigation Legislation ve provided: It is not reasonably practical to rebefore commencement of flight,	
					The aircraft shall not exceed six (flights with the FDR unserviceabl with the first flight after the FDR operating throughout the flight,	e beginning
					Not more than 48 hours have elap FDR became unserviceable,	sed since the
					The aircraft may not depart from base with the FDR unserviceable,	
					The Cockpit Voice Recorder mus normally.	t be operating

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	Item		(3) Nun	nber required for despatch	
			(5) 14411		
				(4) Remarks or Exceptions	
<u>32</u>	LANDING GEAR				
1.	Landing Gear Warning and Indicating System	1	1	Must be operative.	
2.	Parking Brake	1	1	Must be operative.	
3.	Emergency Air Brakes	1	1	Must be operative.	
4.	Air Brake Pressure Gauges	2	1	(M) Air pressure gauge in hydraulic service inoperative provided the brake pressure gaucentre instrument panel operates normally.	centre may be ge on the

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(1) Sy	stem & Sequence Numbers	(2) Numl	ber Install		
	Item		(3) Nur	umber required for despatch	
				(4) Remarks or Exceptions	
<u>33</u>	<u>LIGHTS</u>				
1.	Cockpit and Instrument Lighting Systems	-	-	As required by Air Navigation Legislation. M inoperative for daylight operations only. OR As required by Air Navigation Legislation. In lights may be inoperative provided: (a) Sufficient lighting is operative to mak required instrument, control, and othe which it is provided easily readable, (b) Sufficient flight deck emergency ligh operative, and (c) Lighting configuration at despatch is the flight crew.	dividual se each er device for ting is
2.	Cabin Interior Lighting (If Installed)		-	As required by Air Navigation Legislation. In lights may be inoperative provided: (a) Lighting is adequate for the cabin cre their required duties. (b) Cabin emergency lighting is verified of OR (c) Passengers are not carried. NOTE Cabin emergency lighting does not in proximity lights.	w to perform operative.

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(1) Sys	stem & Sequence Numbers	(2) Numb	per Installe				
	Item		(3) Nur	mber required	d for despa	atch	
			(-,	(4) Remark			
				(4) Keman	NS OF LACE	plions	
33	LIGHTS (Cont)						
<u> </u>	LIGITIS (COIIL)						
3.	Passenger Notice System ("NO SMOKING/FASTEN SEAT BELT/RETURN TO CABIN") Signs	-	-	passeng occupie Return t	er seat, cand from what of Cabin" must be	ed by Air Navigation Leginabin attendant seat or lavate hich a "No Smoking/Faster sign is not readily legible of blocked and placarded - "I	ory may be a Seat Belt/ or that seat or
		-	-	may be	inoperativ tendant se	Fasten Seat Belt/Return to we and the affected passeng eat(s) or lavatories may be	er seat(s),
				(a)		system operates normally a leard throughout the cabin	
				(b)	passenge smoking	ptable procedure is used to ers when seat belts must be is prohibited and (if applied ers should return to cabin fi ments.	fastened, cable) when
					OR		
				(c)	Passenge	ers are not carried.	
4.	Cargo Compartment Light System	2	0			underfloor baggage compa noperative.	artment
5.	Anti-Collision Lights						
	(1) Daylight Operations	-	0	be inope	erative pro	ir Navigation Legislation. ovided the light(s) is repair le opportunity.	

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` / - ʃ ·	Item	Г,			
			(3) Nu	Imber required for despatch	
				(4) Remarks or Exceptions	
33	LIGHTS (Cont)				
5.	Anti-Collision Lights (Cont)				
	(2) Night Operations	-	1	As required by Air Navigation Legislation operative, and a high intensity strobe light installed and operative.	
				NOTE Operations with inoperative anti-oracle are limited to flights within the U.	
6.	Wing Illumination Lights (If Installed)	-	0	Any may be inoperative for daylight opera	tions.
	(II Ilistanea)	-	-	Light(s) may be inoperative for night oper sufficient lighting is available to allow inspleading edge and engine inlet scoop.	
		-	0	(O) All may be inoperative for night opera an alternate means is available and utilised illuminate ice accretion on the wing leadin engine inlet scoop.	to adequately
7.	Landing Lights	2	0	One or both may be inoperative for daylight OR	nt operations.
		2	1	One may be inoperative for night operation	ns provided:
				(a) A taxy light is installed and opera	tes normally,
				(b) It is not reasonably practical to rebefore departure, and	pair or replace
				(c) Repairs or replacements are carried three calendar days.	ed out within
8.	Taxy Lights	2	0	Both may be inoperative.	

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	Item] [(3) Nu	mber required for despatch				
				(4) Remarks or Exceptions				
33	LIGHTS (Cont)							
<u> </u>	LIGITIO (COIII)							
9.	Position Lights System	3	-	Any or all may be inoperative for daylight	operations.			
10.	Interior Emergency Exit Lighting System	1	0	As required by Air Navigation Legislation inoperative provided passengers are not ca				
11.	Exterior Emergency Illumination System (If Installed)	-	-	As required by Air Navigation Legislation inoperative for daylight operations. OR	. May be			
				May be inoperative provided passengers a	re not carried.			
12.	Floor Proximity Emergency Escape Path Marking System Lights (If Installed)	-	-	As required by Air Navigation Legislation lights may be inoperative in accordance w arrangements approved by the Authority folighting configuration.	ith			
				If the equipment becomes unserviceable the continue to fly in accordance with arrange by the Authority.				
				OR				
				May be inoperative provided passengers a	re not carried.			
13.	Wing Tip Clearance Lights (If Installed)	2	0	Both may be inoperative.				

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				(4) Remarks of Exceptions	
0.4	NAVIOATION				
<u>34</u>	NAVIGATION				
1.	Airspeed Indicator	2	2	Both must be operative.	
	•				
2.	Mach Airspeed Warning	1	0	(O) May be inoperative provided the aircraft	is operated
2.	warming		Ü	more than 30 knots below Flight Manual lim	
				AND	
				If aural warning is operative and overspeed v	varnings occur
				earlier than scheduled during flight, speed m below the point at which the warning occurs.	
				below the point at which the warning occurs.	
2	41.5				
3.	Altimeter				
	(1) Basic Altimeters	-	2	Any in excess of two may be inoperative pro	vided:
				(a) One altimeter is operative for each p	oilot,
				(b) At least one of the above is a pneum	natic, or servo
				pneumatic altimeter, and	
				(c) Repairs or replacements are carried	out within
				three calendar days.	
	(2) Servo Pneumatic Altimeter Mode (If Installed)	-	0	(M) May be inoperative provided Altimeter in pneumatic mode.	emains in the
				NOTE Transponder mode "C" will be inop	erative in
				barometric mode.	Stative III
4.	Rate of Climb Indicator	2	1	One may be inoperative for day VMC flight	conditions.
5.	Attitude Display Indication				
		2	1	One may be inongrative for day VMC provide	lad:
	(1) Attitude Display Indicators	2	1	One may be inoperative for day VMC provide	
				(a) The Standby Attitude Indicator oper and	ates normally,
				Cont	

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			(5) 14411		
				(4) Remarks or Exceptions	
<u>34</u>	NAVIGATION (Cont)				
5.	Attitude Display Indication (Cont)				
	(1) Attitude Display Indicators (Cont)				
				(b) Repairs or replacements are carried of three calendar days.	out within
	(2) ADI Command Bars	2	0	One or both command bars may be inoperative approach minima are not dependant on their under the command of the command bars may be inoperative.	
6.	Vertical Gyro Systems	2	1	One may be inoperative for day VMC provide	ed:
				(a) The Standby Attitude Indicator opera	ites normally,
				(b) Repairs or replacements are carried of three calendar days.	out within
7.	Standby Attitude Indicator	1	0	May be inoperative for day VMC provided:	
	(If Installed)			(a) Both ADI's are operative,	
				(b) Both Vertical Gyros are operative, an	nd
				(c) Repairs or replacements are carried of three calendar days.	ut within
8.	Turn and Bank Indicators (If Installed)	2	1	One may be inoperative for Day VMC operat	ions only.
		2	0	Both may inoperative provided three independent indicators are operative.	lent attitude

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	Item		(3) Nur	mber required for despatch	
		7	(6)		
				(4) Remarks or Exceptions	
<u>34</u>	NAVIGATION (Cont)				
9.	Horizontal Situation Indication				
	(1) Horizontal Situation Indicators	2	1	(M) One Indicator may be inoperative provid	ed:
				(a) Both Directional Gyros are operative	e, and
				(b) An independent stabilised heading in available on each pilots panel.	ndication is
	(2) Directional Gyros	2	1	One may be inoperative for day VMC provide	ed:
				(a) A stabilised heading indication is av each pilots panel,	ailable on
				(b) The Standby Compass is operative, a	and
				(c) Repairs or replacements are carried of three calendar days.	out within
	(3) Automatic slaving	2	1	May be inoperative for one Directional Gyro	provided:
				(a) A stabilised heading indication is av each pilots panel, and	ailable on
				(b) The Standby Compass is operative.	
	(4) Radio Magnetic Indicators	2	1	(M) One Indicator may be inoperative provid	ed:
				(a) Both Directional Gyros are operative	e, and
				(b) An independent stabilised heading in available on each pilots panel.	ndication is
10.	Magnetic Compass	1	0	(O) May be inoperative provided:	
				(a) Both directional gyro compass system operating normally, and	ms are
				(b) Repairs or replacements are carried of three calendar days.	out within

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stem & Sequence Numbers	(2) Numl	ber Installe		
Item		(2) Nur	mhor required for depostoh	
	1	(3) Nul		
			(4) Remarks or Exceptions	
NAVIGATION (Cont)				
Flight Director System (If Installed)	-	0	May be inoperative provided weather minima require their use.	do not
Distance Measuring Equipment (DME)	-	-	As required by Air Navigation Legislation.	
Marker Beacon	-	-	As required by Air Navigation Legislation.	
Doppler Navigation System (If Installed)	-	-	As required by Air Navigation Legislation.	
Weather Radar	1	0	when flying for the purpose of public transpo	rt except that
			only one pilot, so long as the aircraft only to the place at which it first bec	is flying omes
			OR	
			the commander of the aircraft indical cumulo-nimbus clouds or other pote hazardous weather conditions, which detected by the system when in work unlikely to be encountered on the into or any planned diversion therefrom a commander has satisfied himself that weather conditions will be encounted daylight and can be seen and avoided aircraft is in either case operated through the commander has a seen and a second through the commander has set the commander has set through the commander has set	te that intially in can be coming order, are iended route or the it any such red in ind, and the bughout the
	NAVIGATION (Cont) Flight Director System (If Installed) Distance Measuring Equipment (DME) Marker Beacon Doppler Navigation System (If Installed)	NAVIGATION (Cont) Flight Director System (If Installed) Distance Measuring Equipment (DME) Marker Beacon Doppler Navigation System (If Installed) -	Postem & Sequence Numbers Item (2) Number Install (3) Number Install (4) Number Install (5) Number Install (6) Number Install (7) Number Install (8) Number Install (9) Number Install (1) Number Install (1) Number Install (2) Number Install (3) Number Install (A) Number Install (B) Number Install (C) Number Install (A) Number Install (B) Number Install (C) Number Ins	DATE: 25 OCTOBER 1996

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	(2) Num	ber Install	ed			
nem	_	(3) Nu	mber required for despatch			
			(4) Remarks or Exceptions			
NAVIGATION (Cont)						
Radio Compass (ADF) System	2	-	As required by Air Navigation Legislation.			
VOR Systems	2	-	As required by Air Navigation Legislation.			
ILS	2	-	As required by Air Navigation Legislation.			
ATC Transponder	-	-	As required by Air Navigation Legislation.			
LORAN (If Installed)	1	-	As required by Air Navigation Legislation.			
Instrument Comparator or Warning Systems (If Installed)	1	0	May be inoperative provided weather minima do not require their use.			
Outside Air Temperature Gauge	1	1	Must be operative.			
Ground Proximity Warning System (GPWS)	1	0	As required by Air Navigation Legislation. May be inoperative. The aircraft may continue the flight or series of flights but shall not depart an airport where it is reasonably practicable for repairs or replacements to be made. NOTE Particular circumstances may require the use of additional or alternate procedures. The alternate procedures would require the operator to consider the routes over which he is flying and ensure that the pilot adopted a flight path which would give him the protection which would otherwise be afforded.			
	ICCKHEED L-188 ELECTION (Septem & Sequence Numbers Item) NAVIGATION (Cont) Radio Compass (ADF) System VOR Systems ILS ATC Transponder LORAN (If Installed) Instrument Comparator or Warning Systems (If Installed) Outside Air Temperature Gauge Ground Proximity Warning System	ILOCKHEED L-188 ELECTRA Stem & Sequence Numbers Item	ILS 2 - ATC Transponder - LORAN (If Installed) 1 - Instrument Comparator or Warning Systems (If Installed) 1 0 Ground Proximity Warning System 1 0 Ground Proximity Warning System 1 0			

AIRCRAFT:			REVISION NO: REVISION 2	PAGE:	
	LOCKHEED L-188 ELF	ECTRA		DATE : 25 OCTOBER 1996	34-6
(1) Sys	stem & Sequence Numbers	(2) Numl	ber Install	ed	
	Item		(3) Nu	mber required for despatch	
			(6) 114		
				(4) Remarks or Exceptions	
<u>34</u>	NAVIGATION (Cont)				
24.	Traffic Alert and Collision Avoidance System (TCAS) (If Installed)				
	(1) TCAS System	-	0	(M) May be inoperative provided the system is and secured.	s deactivated
				If the aircraft is intended to be flown in airspace TCAS operation is required, it may fly for not 10 calendar days with the equipment complete unserviceable provided that this is permitted by appropriate Air Traffic Control Authorities, but depart from an aerodrome where it is reasonab for the equipment to be repaired or replaced.	more than ely y the it shall not
	(2) Combined TA and RA Dual Displays	2	1	(O) May be inoperative on the non-flying pilot provided:	side
				(a) TA and RA elements and audio funct operative on flying pilot side, and	ions are
				(b) TA and RA display indications are v non-flying pilot.	isible to the
	(3) Resolution Advisory (RA) Display System(s)	2	1	(O) One may be inoperative on non-flying pilo	ot side.
		-	0	(O) May be inoperative provided:	
				(a) All Traffic Alert (TA) display element command audio functions are operation	
				(b) TA only mode is selected by the crew	⁷ .
	(4) TA Display System(s)	-	0	(O) May be inoperative provided all installed I and audio functions are operative.	RA display

AIRCRAFT:				REVISIO	N NO: REVISION 2	PAGE:
	LOCKHEED L-188 ELF	ECTRA		DATE:	25 OCTOBER 1996	34-7
(1) Sys	tem & Sequence Numbers	(2) Number	r Installe			
	Item		(2) Nun	mbor roquirod	for despatch	
		┤	(3) INUI			
				(4) Remark	s or Exceptions	
<u>34</u>	NAVIGATION (Cont)					
25.	Altitude Alerting System	-	0	inoperati of flights	red by Air Navigation Legislatio ve. The aircraft may continue the but shall not depart an airport welly ly practicable for repairs or repla	ne flight or series where it is
26.	Radio Altimeter System	-	0	May be i	noperative provided:	
				(a)	The GPWS is considered inoper	rative,
					Approach minima or operating prequire its use, and	procedures do not
					Repairs or replacements are carr calendar days.	ried within three
27.	Microwave Landing System	-	0	May be i	noperative.	
28.	Transfer Switching Systems (If Installed)	-	0	(M)/(O)	May be inoperative provided:	
	((a)	The required indicators are oper	ative, and
					Operation from independent sou with the Transfer switch(es) in t position.	
				NOTE	See item 34-5 or 34-9 as approp	riate.
		1 1				

MASTER MINIMUM EQUIPMENT LIST

(1) System & Sequence Numbers (3) Number installed 35 OXYGEN (4) Remarks or Exceptions (5) Number required for despatch (6) Remarks or Exceptions (7) As required by Air Navigation Legislation. The automatic presentation system may be inoperative provided: (8) The flight is limited to FL 300 or below. (9) The flight is limited to FL 300 or below. (1) Affected seats are blocked and placarded to prevent occupancy, and (1) Units operate normally for all usable passenger seats, tolet compartments and flight attendant locations. (1) OR (2) Number required for despatch (4) Remarks or Exceptions (1) O (1) As required by Air Navigation Legislation. The automatic presentation system may be inoperative provided: (1) The flight is limited to FL 300 or below. (2) (4) Or (7) One or more passenger service units (PSUs) may be inoperative without flight altitude restriction provided: (1) Affected seats are blocked and placarded to prevent occupancy, and (2) (3) Number required for despatch (4) Remarks or Exceptions (4) Affected seats are blocked and placarded to prevent occupancy, and (b) Units operate normally for all usable passenger seats, tolet compartments and flight attendant locations, (6) OR (7) OR (8) OR (9) Both air conditioning packs operate normally, (1) Conditioning packs operate normally, (2) All other components of the pressurisation system operate normally, (3) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (1) Passengers are appropriately briefed, and (2) Repairs or replacements are carried out within three calendar days.	AIRCRAFT:			REVISION NO: REVISION 2	PAGE:				
Comparison of the pressurisation of the pressurisation of the prevalued for despatch (3) Number required for despatch	LOCKHEED L-188 EI	LECTKA		DATE: 25 OCTOBER 1996	35-1				
(3) Number required for despatch (4) Remarks or Exceptions (5) Remarks or Exceptions (6) Remarks or Exceptions (7) As required by Air Navigation Legislation. The automatic presentation system may be inoperative provided: (a) The manual deployment system operates normally, and (b) The flight is limited to FL 300 or below. (a) Affected seats are blocked and placarded to prevent occupancy, and (b) Units operate normally for all usable passenger seats, toilet compartments and flight attendant locations, OR (a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within	(1) System & Sequence Numbers	(2) Numb	per Install	r Installed					
(4) Remarks or Exceptions (3) Remarks or Exceptions (4) Remarks or Exceptions (5) Air variation (6) Air variation (6) Air variation (6) Air variation (7) Air variation (7) Air variation (7) Air variation (7) Air variation (8) Remarks or Exceptions (8) Air variation (8) Remarks or Exceptions (8) Air variation (8) Air variation (8) Remarks or Exceptions (8) Air variation (8) Air variation (8) Air variation (8) Repairs or replacements are carried out within (ltem		(3) Nu	mber required for despatch					
1. Passenger Oxygen System 1. Passenger Oxygen System 1. O (M) or (O) As required by Air Navigation Legislation. The automatic presentation system may be inoperative provided: (a) The manual deployment system operates normally, and (b) The flight is limited to FL 300 or below. (M) or (O) One or more passenger service units (PSUs) may be inoperative without flight altitude restriction provided: (a) Affected seats are blocked and placarded to prevent occupancy, and (b) Units operate normally for all usable passenger seats, toilet compartments and flight attendant locations, OR 1. O (O) May be inoperative provided: (a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within									
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automatic presentation system may be inoperative provided: (a) The manual deployment system operates normally, and (b) The flight is limited to FL 300 or below. - (M) or (O) One or more passenger service units (PSUs) may be inoperative without flight altitude restriction provided: (a) Affected seats are blocked and placarded to prevent occupancy, and (b) Units operate normally for all usable passenger seats, toilet compartments and flight attendant locations, OR 1 0 (O) May be inoperative provided: (a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within	O OXTOLIN								
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(M) or (O) One or more passenger service units (PSUs) may be inoperative without flight altitude restriction provided: (a) Affected seats are blocked and placarded to prevent occupancy, and (b) Units operate normally for all usable passenger seats, toilet compartments and flight attendant locations, OR 1 0 (O) May be inoperative provided: (a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within					perates				
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seats, toilet compartments and flight attendant locations, OR 1					acarded to				
1 (O) May be inoperative provided: (a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within				seats, toilet compartments and flig					
(a) Flight is not conducted where the minimum enroute altitude is above 12,000 feet MSL, (b) Both air conditioning packs operate normally, (c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within				OR					
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(c) All other components of the pressurisation system operate normally, (d) Maximum flight altitude does not exceed FL 250, (e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within									
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(e) Portable oxygen units containing sufficient oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within					urisation system				
oxygen for 30 minutes endurance are provided for 10% of the passengers, (f) Passengers are appropriately briefed, and (g) Repairs or replacements are carried out within				(d) Maximum flight altitude does not	exceed FL 250,				
(g) Repairs or replacements are carried out within				oxygen for 30 minutes endurance					
				(f) Passengers are appropriately brief	ed, and				
					d out within				

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LOCKHEED L-188 ELEC	CTRA	DATE	27 OCTODED	1007	25.2
(1) System & Sequence Numbers	(2) Number I	DATE:	25 OCTOBER	1996	35-2
Item	(_,				
	(3) Number require	d for despatch		
		(4) Remai	ks or Exceptions		
35 OXYGEN (Cont)					
Passenger Oxygen System (Cont)		NOTE:	Schedule 4 Sca depends upon of airworthiness, may have exams scales of requir The amount of considerably be operations above operator supplied despatch is considerably be operator to allow the operator to allow the operator supplied to all supplied to all supplied to allow the operator supplied to allow the	oxygen required varieties and L2, payer FL 250/300. Provides the required amous sidered acceptable. a large number of perfer to Air Navigation erator to adapt the Min the constraints applies.	effectively certificate of pe of aircraft of the two es rticularly for ided the nt of oxygen, rmutations, it in Legislation EL as
		(a)		t issue of a certificate for individual aircraft	
		(b)	The aircraft altiflown, and	itude and cabin altitude	de on routes
		(c)	The numbers of	f passengers and crew	/ carried.

AIRCRAFT: LOCKHEED L-188 ELECTRA				REVISION NO: REVISION 2	PAGE:
				DATE: 25 OCTOBER 1996	49-1
(1) Sys	stem & Sequence Numbers Item	(2) Numl	ber Installe	ed nber required for despatch	
			(0) 1401	(4) Remarks or Exceptions	
<u>49</u>	AIRBORNE AUXILIARY POWER				
1.	Auxiliary Power Unit (If Installed)	1	0	(M)(O) May be inoperative.	
2.	APU Generator	1	0	May be inoperative.	
3.	APU Bleed Air	1	0	May be inoperative.	
4.	APU Caution and Warning Lights	6	0	May be inoperative provided the APU is con inoperative.	nsidered

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AIRCRAFT: LOCKHEED L-188 ELECTRA			REVISION NO: REVISION 2 DATE: 25 OCTOBER 1996	PAGE: 52-1	
(1) Sy	stem & Sequence Numbers	(2) Numb	er Install		32-1
	Item		(3) Nui	mber required for despatch	
				(4) Remarks or Exceptions	
<u>52</u>	DOORS				
1.	Forward Airstair (Electric Activation)	-	-	 (M) (O) May be inoperative provided: (a) Airstair can be deployed manually, a (b) The Forward Main Cabin Door is on modes. 	
2.	Door Warning Light System	2	0	(O)(M) May be inoperative provided doors a failed annunciator are determined by inspectic closed and locked prior to each departure. NOTE: Belly Door is closed and locked if a pressurised before take-off.	on to be
3.	Main Cabin Cargo Door Warning Light System (If Installed)				
	(1) Lockheed System	2	0	(O) Door Warning Lights may be inoperative determined by visual inspection that the door and Locked before each departure.	
4.	Lockable Flight Deck Door	1	1	As required by Air Navigation legislation.	
	(1) Door Lock	1	0	As required by Air Navigation Legislation. may be inoperative provided (a) The latch is operative and the door of in the appropriate position, either cland (b) Repairs or replacements are carried three calendar days. Cont	can be secured osed or open,

AIRCRAFT:			REVISION NO: RE	EVISION 2	PAGE:	
	LOCKHEED L-188 ELE	CIKA		DATE: 25 OCTOB	ER 1996	52-2
(1) Sys	stem & Sequence Numbers	(2) Numb	per Installe	d		•
	Item		(3) Nur	ber required for despatch		
				(4) Remarks or Exception	ıs	
<u>52</u>	DOORS (Cont)					
4.	Lockable Flight Deck Door (Cont)					
	(2) Door and Door Latch	1	0	or door latch may be can not be secured in or open, it shall be re	Air Navigation Legislation inoperative provided that the appropriate position, emoved.	if the door either closed
5.	Main Cabin Cargo Door (Electric Activation)		0	May be inoperative p manually.	provided door can be open	ned

AIRCRAFT:			REVISION NO: REVISION 2 PAGE						
	LOCKHEED L-188 ELEC	JIKA		DATE: 25 OCTOBER 1996 61-1					
(1) Sy	stem & Sequence Numbers	(2) Numl	ber Install	ed					
	Item		(3) Nur	Number required for despatch					
				(4) Remarks or Exceptions					
				(1) Nomanie di Exceptione					
61	PROPELLERS PROPERTY								
01	FROFELLERS								
1.	Beta Lights	4	4	Must be operative.					
2	NTC Lighto	4	,	Must be exerctive					
2.	NTS Lights	4	4	Must be operative.					
3.	NTS System	4	4	Must be operative.					
4.	Autofeather - TSS System	4	0	(M) (O) One or all may be inoperative provided:					
	(Thrust Sensitive Switch)			(a) Operations are conducted in accordance with Flight Manual limitations,					
				(b) Autofeather arming switch is selected to OFF,					
				(c) Autofeather is considered inoperative on all engines,					
				(d) Manual feathering of all engines is verified to be operating normally,					
				(e) The cause of the malfunction must be determined to ensure no hazards exist, and					
				(f) Repairs or replacements are carried out within three calendar days.					
				NOTE Autofeather system must NOT be de-activated by circuit breakers as this will de-activate manual feathering by emergency handle or button.					
5.	Propeller Feather System	4	4	Must be operative.					
6.	Propeller Spinners	4	4	Must be fitted.					
7.	Propeller Synchronization System								
	(1) Phase Mode	1	0	(O) May be inoperative.					

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		LOCKHEED L-188 ELEC	JIRA		DATE:	25 OCTOBER 1996	61-2	
(1) Sys	stem & Sec	quence Numbers	(2) Numb	per Install	ed			
		Item		(3) Nui	mher require	ed for despatch		
			1	(3) 1401		·		
					(4) Remar	ks or Exceptions		
<u>61</u>	PRO	PELLERS (Cont)						
7.	Propel (Cont	ler Synchronization System .)						
	(2)	Synch Mode	1	0	(O)(M)	May be inoperative provided:		
					(a)	Pedestal synchronization switch is and	s selected off,	
					(b)	Propeller synchronization circuit and tagged.	breaker is pulled	
					NOTE	 Pulling the synchronization circuirenders synchronization inoperation propellers. 		
					NOTE:	<u>2</u> Defect analysis may require main procedure detailed in MMEL iten Actuator.		
	(3)	Rotary Actuator	4	0	(M)(O)	One or more may be inoperative p	rovided:	
					(a)	Scribe marks on speed ring and replate are aligned and locked in the approved means.		
					(b)	Electrical connection is removed inoperative actuator and secured I means, and		
					(c)	Propeller synchronization circuit and tagged.	breaker is pulled	
					NOTE	Pulling the synchronization circuirenders the synchronization inoperopellers.		
8.		on Standard Low Oil Level ng Light	4	0	May be	inoperative provided:		
	vv urmin	15 1.15m			(a)	Propeller synchronization operate	s normally,	
					(b)	Associated propeller(s) shows no leakage,	evidence of	
					(c)	Associated propeller(s) oil level is recommended capacity prior to ea and		
					(d)	Repairs or replacements are carrie three calendar days.	ed out within	

AIRCRAFT:			REVISION NO: REVISION 2	PAGE:	
	LOCKHEED L-188 EL	ECTRA		DATE: 25 OCTOBER 1996	73-1
(1) Sys	tem & Sequence Numbers	(2) Numb	er Installe		<u> </u>
	Item		(3) Nur	nber required for despatch	
			(3) 1401		
				(4) Remarks or Exceptions	
<u>73</u>	ENGINE FUEL &				
	CONTROL				
1.	Temperature Datum System	4	3	(M) One may be inoperative provided:	
				(a) Associated Fuel Flow Indicator is ope	erative, and
				(b) Associated TIT Indicator is operative	; .
2.	Fuel Flow Indicators	4	3	One may be inoperative provided:	- arativa
				(a) Associated Engine Instruments are op(b) All Fuel quantity Indicators are opera	
				(c) Repairs or replacements are carried o	
				three calendar days.	

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LOCKHEED L-188 EI	LECTRA	DATE : 25 OCTOBER 1996 74-1
(1) System & Sequence Numbers Item	(2) Numb	(3) Number required for despatch
74 IGNITION		(4) Remarks or Exceptions
1. Speed Sensitive Control (Switch)		(a) Associated Switch is secured by an accepted procedure, and (b) An alternate operating procedure is developed and used.

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:				REVISIO	N NO:	REVISION 2	PAGE:		
	LOCKHEED L-188 ELEC	TKA		DATE:	25 OCT	OBER 1996	77-1		
(1) Sys	tem & Sequence Numbers	(2) Numb	per Installe	ed			-		
	Item] [(3) Nur	Number required for despatch					
				(4) Remark	ks or Excep	otions			
<u>77</u>	ENGINE INDICATING								
1.	TIT Indicators (Digital Function)	4	0			ve provided repairs on three calendar days			
	(Digital Function)				out within	ir timoo caronaar aays			
2.	Torque Indicating System			Deleted	in Revisi	on 2.			
	(Horsepower Gauge System)								
3.	Temperature Datum (TD) System			Moved t	to ATA C	hapter 73.			
4.	Fuel Flow System			Moved t	to ATA C	hapter 73.			
	·					1			
5.	Vibration Indicator System	1	0	May ba	inoperativ				
5.	Vibration Indicator System (If Installed)		U	Wiay be	пореган	v C.			
6.	Tachometer	4	4	Must be	operative).			
					•				

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:				REVISION NO: REVISION 2 PAGE:					
	LOCKHEED L-188 ELEC	TRA		DATE:	25 OCTOBER 1996	79-1			
(1) Sys	tem & Sequence Numbers	(2) Numb	er Installe	ed		<u> </u>			
	Item	ļ	(3) Number required for despatch						
				(4) Remar	ks or Exceptions				
<u>79</u>	ENGINE OIL								
1.	Engine Oil Pressure Indicators (Power Section)			Deleted	in Revision 2.				
2.	Engine Oil Pressure Indicators (Gear Box)			Deleted	in Revision 2.				
3.	Engine Oil Pressure Warning Lights	4	3	One ma	y be inoperative provided:				
				(a)	Associated Power Section and Go Pressure Indicators are operative,				
				(b)	Associated Oil Temperature Indicoperative,	cators are			
				(c)	Associated Oil Quantity Indicator	rs are operative,			
				(d)	Associated engine is not started f	irst, and			
				(e)	Repairs or replacements are carrie three calendar days.	ed out within			
4.	Engine Oil Temperature Indicator			Deleted	in Revision 2.				
5.	Oil Cooler Flap Position Indicator	4	3		y be inoperative provided the associature indicator is operative.	ciated oil			
6.	Oil Cooler Flap Control System								
	(1) Manual Function			Deleted	in Revision 2.				
	(2) Automatic Function (If Installed)	4	0	All may	be inoperative.				
		1							

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	LOCKHEED L-188 EL	ECTRA		DATE: 25 OCTOBER 1996	79-2
(1) System & Sequence Numbers (2		(2) Number	Number Installed		
Item					
		\dashv \mid	(3) Nur	mber required for despatch	
				(4) Remarks or Exceptions	
79	ENGINE OIL (Cont)				
7.	Oil Quantity Indicators	4	3	(M) One may be inoperative provided:	
				(a) The associated Power Section and Ge Pressure Indicators are operative,	ear Box Oil
				(b) The associated Oil Temperature Indicoperative,	cators are
				(c) The associated Low Pressure Warnin operative,	g Lights are
				(d) There is no evidence of above norma consumption or leakage,	l oil
				(e) The associated Oil Quantity is verified before each departure, and	d adequate
				(f) Repairs or replacements are carried o three calendar days.	ut within
8.	5. Oil Cooler Inducers 4		0	(O) May be inoperative provided:	
				(a) The associated Oil Temperature Indicoperative,	cator is
				(b) The Oil Temperature is closely monit ground operations, and	cored during
				(c) Repairs or replacements are carried o three calendar days.	ut within