



NO: BM60 ISSUE: 5

TYPE: Mainair Blade 912S

(1) MANUFACTURER: P & M Aviation Ltd, Unit B, Crawford Street, Rochdale,

Lancashire, OL16 5NU

(2) UK IMPORTER: -

(3) CERTIFICATION: BCAR Section S Issue 1

BCAR Section S Issue 2 paras S23.525, S1557(c)(v) and S1583

(4) DEFINITION OF Gemini 5 Drawing Register Issue 4 29th March 2000 BASIC STANDARD: Beta Wing Drawing Register Issue 2 30th March 2000

(5) COMPLIANCE WITH THE MICROLIGHT DEFINITION

(a) MTOW 390 kg

(b) No. Seats 2

(c) Maximum Wing Loading 25 kg/m²

(d) Vso (not limiting) 28 kts IAS (32 mph IAS)

(e) Permitted range of pilot weights 55-90 kg per seat.

(f) Typical Empty Weight (ZFW) 195 kg

(g) ZFW + 172 kg crew + 1 hr fuel 380 kg

(18litres /13 kg)

(h) ZFW + 86 kg pilot + full fuel 313 kg

(litres / kg)

With 65 litre tank 328 kg

(i) Max ZFW at initial permit issue 205 kg



MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

(6) POWER PLANTS

(o) IOVERTE			
Designation	Blade 912S		
Engine Type	Rotax 912 ULS		
Reduction Gear	2.43/1		
Exhaust System	Mainair 912 Exhaust		
Intake System	Intake Filter		
Propeller Type	Mainair WD Ground ADJ		
Propeller Dia x Pitch	3 Blade 61" dia 122° at 12"		
Noise Type Cert No.	150M Iss 3		
AAN approving configuration	27342		

(7) MANDATORY LIMITATIONS:

(A) Max Take-Off Weight 390 kg

(B) CG Limits N/A
(C) CG datum N/A

(D) Cockpit Loadings Front Rear Total

Min 55 kg 15 kg 70 kg

Max 90 kg 90 kg 180 kg

(E) Never Exceed Speed 101 mph (88 kts)

(F) Manoeuvring Speed 60 mph (52 kts)

(G) Permitted Manoeuvres - maximum 30° Nose up / 30° nose down

bank angle 60° Maximum bank angle 60°

Non Aerobatic

Normal acceleration limits, +4 / 0g

(H) Fuel Contents (Max Useable) 44 litres (of which 22 litres optional)

65 litres (Mod 99)

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

(I) Power Plant		See Table		
Engine	Rotax 912 ULS			
Max RPM	5800 (5 mins)			
	5500 (Cont)			
MAX CHT	135°C (284°F)			
MAX EGT	850°C (1560°F)			
Fuel Spec	83 MON or 90 RON minimum unleaded to BS(EN)228 or 97+ octane 4-star /MOGAS leaded fuel to BS 4040, or AVGAS 100LL.			
Engine Oil Spec API Grade or SG Automotive				
Gearbox oil spec	N/A			
Fuel/Oil Mix	N/A			
Coolant Temperature	N/A			
Oil Pressure	Max 7 Bar			
	Min 0.8 Bar			
Oil Temperature	Min 50°C			
	Max 130°C			
Fuel Pressure	N/A			

(8) INSTRUMENTS REQUIRED:

ſ	ASI	Altimeter	RPM	CHT	Compass	Oil temp	Oil
							Pressure
	Required	Required	Required	Required		Required	Required

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

(9) CONTROL DEFLECTIONS: N/A

Elevator UP:	\pm	Tailplane trim UP:	±	
Elevator DOWN:	<u>±</u>	Tailplane trim DOWN	<u>±</u>	
Ailerons* UP:	<u>±</u>	Rudder LEFT:	<u>±</u>	
Ailerons* Down:	±	Rudder RIGHT:	<u>±</u>	

(10) PILOT'S NOTES, MAINTENANCE MANUALS REFERENCES:

- 10.1 Manuals approved for use with this aircraft.
- (a) BD912S Manual Iss 1 (as amended).
- 10.2 The following placards are to be fitted:-
- (a) Flight Limitations Placard (to be visible to pilot)
 See Annex D.
- (b) Engine Limitations Placard (to be located near to engine instruments)
 See Annex D.
- (c) Fuel Limitations Placard (to be located near to filler cap)

A placard is to be fitted showing fuel capacity (litres), fuel type(s), fuel:oil ratio (if relevant) and if MTOW can be exceeded with full fuel and maximum cockpit weight, the fuel loads at MTOW for fuel capacity in increments of 10 litres intervals down to the maximum fuel load. An example is shown at Annex D.

(d) Switches See Annex D.

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

(11) MANDATORY MODIFICATIONS / SERVICE BULLETINS / AIRWORTHINESS DIRECTIVES ETC:

See Annex A for required modifications.

Annual Bettsometer test is to be carried out to 1050 grammes with wing sails fitted and tensioned to flight. Test must be to both upper and lower surfaces.

(12) MINIMUM PERFORMANCE AT MAX TAKE-OFF WEIGHT

Rate of Climb: 1150 ft/min at 55 mph (48 kts) IAS

Stall or Minimum Flying Speed: 32 mph IAS (28 kts IAS) at MTOW / idle.

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

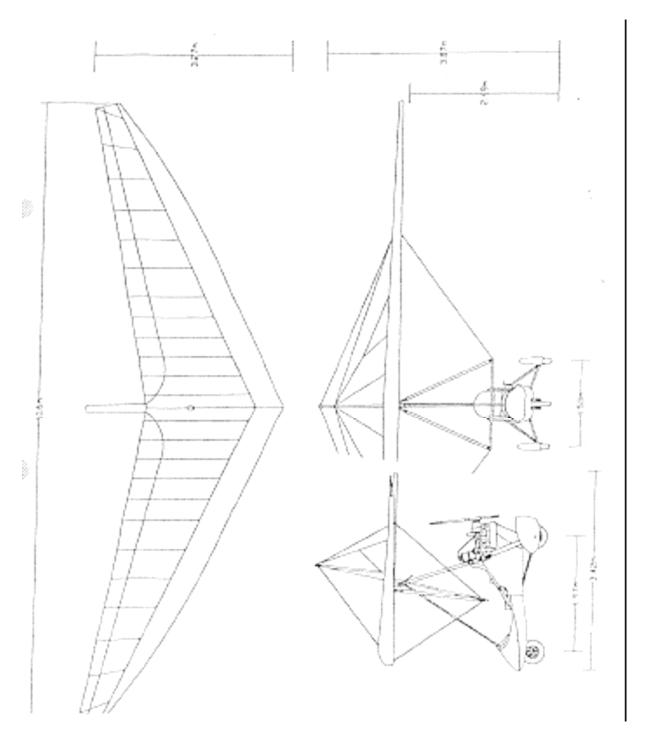
Issue History

	Issue No.	Reason and signatory
1	18/05/00	Initial issue C J WHITTAKER
2	07/01/02	Issue 2 raised to record Mandatory Modification Number 98, which introduces passenger upper torso restraints. J BARRATT
.3	08/07/02	Editorial corrections J BARRATT
4	06/07/05	New company name J BARRATT
5	18/11/09	To record Modification Number 187, which increases seat load limit to 110kg but still with a total cockpit load of 180g.
		DS CORTIZO

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

Illustration of Aircraft - 3 View



MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

ANNEX A MANDATORY MODIFICATIONS

Description Modification Number

Shoulder Straps 98

ANNEX B

APPROVED OPTIONAL MODIFICATIONS

The installation of all optional modifications is to be inspected by a BMAA inspector and an entry made in the appropriate logbook(s). Note that other approved modifications may exist which are not listed here.

Description Modification Number

65 litre Fuel Tank 99 Increase Seat Load Limit to 110kg 187

ANNEX C

WEIGHING INFORMATION

CG Datum: N/A
 Weighing attitude: N/A
 Mainwheel moment arm: N/A

Nosewheel moment arm: N/A (units) (direction) of datum
 Fuel moment arm: N/A (units) (direction) of datum
 Crew moment arm: N/A (units) (direction) of datum

7. Crew weights: Minimum 55 kg / maximum 90 kg

(maximum reducible, not below 86 kg, if required).

8. Aft CG Limit: N/A (units) (direction) of datum9. Fwd CG Limit: N/A (units) (direction) of datum

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

ANNEX D

EXAMPLE PLACARDS

(a) Flight Limitations and Engine Limitations Placard (to be visible to pilot)

MAINAIR BLAI	DE 912S				
STALL SPEED	30MPH				
MANOUVRE SPD	60MPH				
NEVER EXCEED Vne	101MPH				
MIN LANDING APP	50MPH				
MAX CROSS WIND	10MPH				
MAX OIL PRESSURE	7 BAR				
MIN OIL PRESSURE	1.5 BAR				
MAX OIL TEMP	130 DegC				
MIN OIL TEMP	50 DegC				
MAX CHT 135 DegC					
MAX CONT RPM	5500				
MAX RPM	5800				
MAX ALL UP WEIGHT	390KG				
MAX COCKPIT LOAD	180KG				
MIN COCKPIT LOAD 70KG					
EMPTY WEIGHT 195KG					
WARNING					
THIS AIRCRAF	THIS AIRCRAFT IS				
NON-AEROBA	TIC				
MAXIMUM NOSE	UP 30°				
MAXIMUM NOSE DO	OWN 30°				
MAXIMUM BANK ANGLE 60°					
SPINNING ATTEMPS PROHIBITED					
NO WHIPSTALLS LOOPS TAILSLIDES					
OR AEROBATIC MANOEUVRES					
DO NOT EXCEED MAX. LOAD					
MAINAIR SPO	MAINAIR SPORTS				
ROCHDALE ENGLAND					
www.mainairsports.co.uk					

(b) Fuel Limitations Vs Max Cockpit Load Placard (to be visible to the pilot)

Empty Wt	Kg
Date	
Fuel Load	Max Cockpit
(litres)	Wt (kg)
65	
55	
45	
35	
25	
15	
5	_

MICROLIGHT TYPE APPROVAL DATA SHEET (TADS)

NO: BM60 ISSUE: 5

(c) Fuel Capacity and Type (to be located near to filler cap)

FUEL - 22 LITRES

EN 228 UNLEADED or AVGAS 100LL INSPECT INTERNAL AND EXTERNAL FILTERS FREQUENTLY **FUEL - 65 LITRES**

EN 228 UNLEADED or AVGAS 100LL INSPECT INTERNAL AND EXTERNAL FILTERS FREQUENTLY

(d) Switches

All switches are to be marked with function and sense (up=on, down=off).